



CONVECTIVE HEAT TRANSFER MEASUREMENTS AT THE MARTIAN SURFACE

1st Symposium on Space Educational Activities
Student Research

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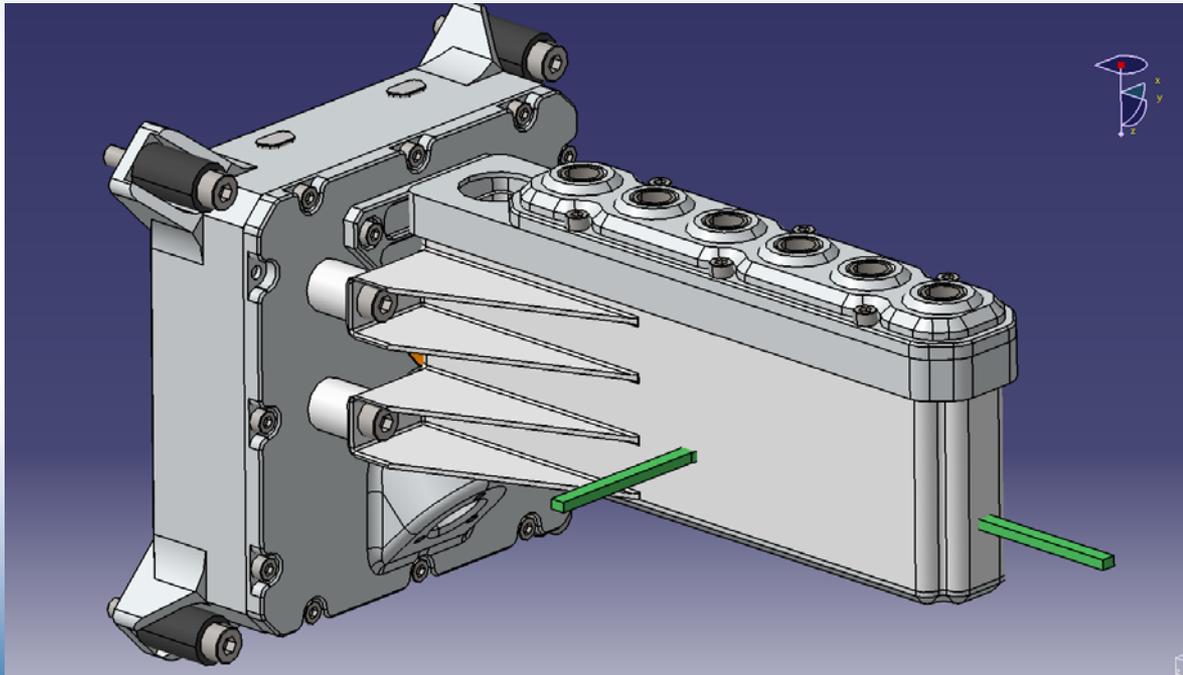
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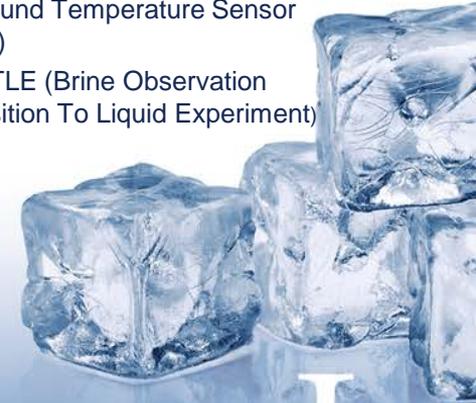
1. Introduction

- Context: HABIT (HabitAbility: Brines, Irradiation and Temperature), instrument included in ExoMars 2018 mission (http://www.esa.int/Our_Activities/Space_Science/European_payload_selected_for_ExoMars_2018_surface_platform). First Swedish instrument for the surface of Mars.
- REMS (Rover Environmental Monitoring Station) heritage.



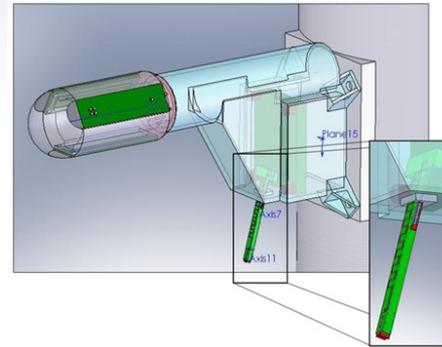
Components:

- 6 UV sensors
- 3 Air Temperature Sensors (ATS)
- 1 Ground Temperature Sensor (GTS)
- BOTTLE (Brine Observation Transition To Liquid Experiment)



1. Introduction

- Objective: Use the HABIT ATS sensors to characterize the wind on Mars surface through the convective heat transfer coefficient.
 - M-Model
- Preliminary study:
 - Physical model for the heat transfer problem.
 - Computational fluid dynamics study.
 - Comparison between theoretical and numerical results.



2. M-model

- Energy Balance:

$$\frac{d}{dx} \left(\kappa A_c \frac{dT}{dx} \right) = h_c \frac{dA_s}{dx} (T - T_\infty) + \varepsilon \sigma \frac{dA_s}{dx} (T^4 - T_\infty^4)$$

- Heat transfer coefficient:

$$h = h_c + h_r = h_c + \varepsilon \sigma (T^2 + T_\infty^2) (T + T_\infty)$$

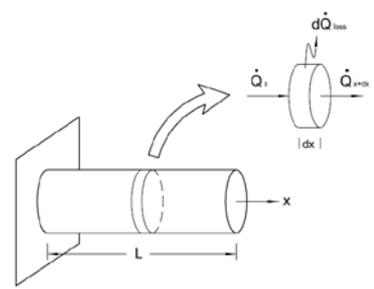
- Organizing:

$$\frac{d^2 \theta}{d\chi^2} - m_\chi^2 (\chi, \theta, T_f) \theta = 0$$

$$\theta = T_x - T_f, \chi = x/L$$

- M-Parameter:

$$m_\chi = L \sqrt{\frac{[3(1 + \cos(\beta)) - \sqrt{(3 + \cos(\beta))(1 + 3\cos(\beta))}] h}{D\kappa}}$$

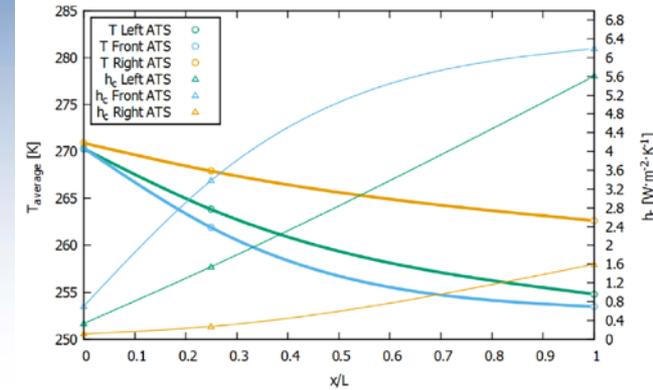


2. M-model

- Non-Linear Boundary-Value Problem along ATS sensors:

$$\left\{ \begin{array}{l} \theta|_{x=0} = \theta_b \\ \frac{d\theta}{dx}|_{x=1} = 0 \end{array} \right. \rightarrow \theta(x) = \theta_b \frac{\cosh(m(1-x))}{\cosh(m)} \rightarrow \left\{ \begin{array}{l} (T_a - T_f) = (T_b - T_f) \frac{1}{\cosh(m)} \\ (T_{Ln} - T_f) = (T_b - T_f) \frac{\cosh(m(1-\frac{1}{n}))}{\cosh(m)} \end{array} \right.$$

- Analytical solution: m averaged over length of the rod.
- Retrieval Method:
 - Three simultaneous measurements
 - T_b : Temperature at the base (Platform).
 - T_{Ln} : Temperature at $x = L/n$.
 - T_a : Temperature at the tip.
 - Unknowns: m, T_f
 - T_f : Fluid temperature



- In this preliminary study, the h_c coefficient is estimated theoretically and compared with a numerical model in order to confirm the usefulness of the ATS as wind sensors.



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3. Physical model

- Fluid Model:

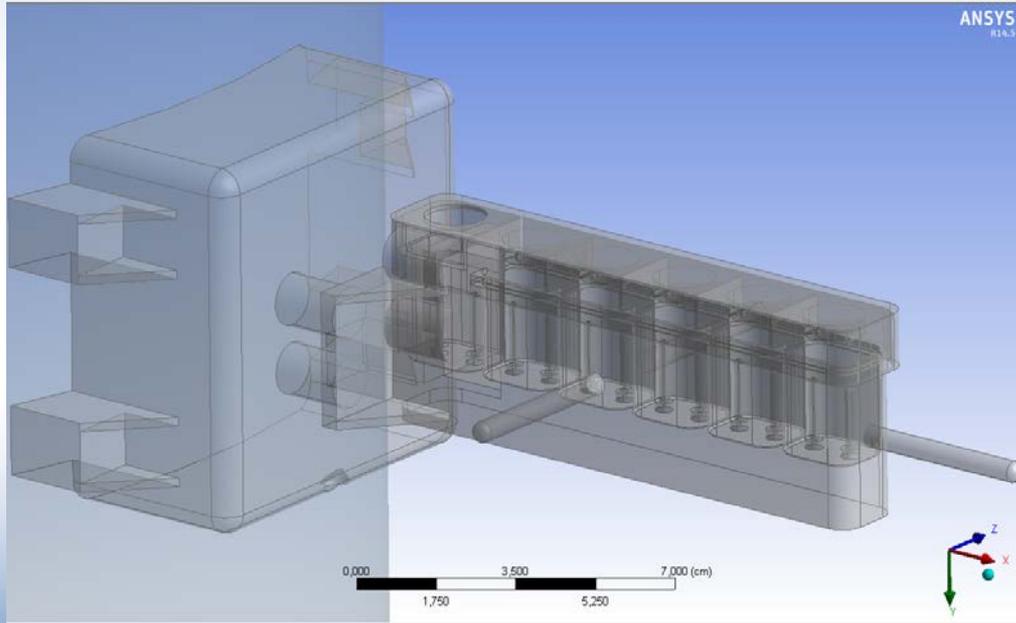
$$\text{Pr} = \frac{C_p(T)\mu}{\kappa_f(T)}, \text{Re} = \frac{\rho v_{\text{wind}} L_c}{\mu}, L_c = D/\cos(\beta)$$
$$\text{Nu} = \frac{h_c L_c}{\kappa_f(T)}; \text{Nu} = 0.3 + \frac{0.62 \text{Re}^{1/2} \text{Pr}^{1/3}}{\left[1 + \left(\frac{0.4}{\text{Pr}}\right)^{2/3}\right]^{1/4}} \left[1 + \left(\frac{\text{Re}}{28200}\right)^{5/8}\right]^{4/5}$$

- Temperature dependent polynomial models for $C_p(T)$ and $\kappa_f(T)$ in low-pressure $[1, 10^4] \text{mbar}$ and low-temperature $[200, 1000] \text{K}$ domains.
- Chapman-Enskog's Kinetic Theory for μ (constant).
- Incompressible ideal gas CO_2 laminar flow ($M < 10^{-3}$).



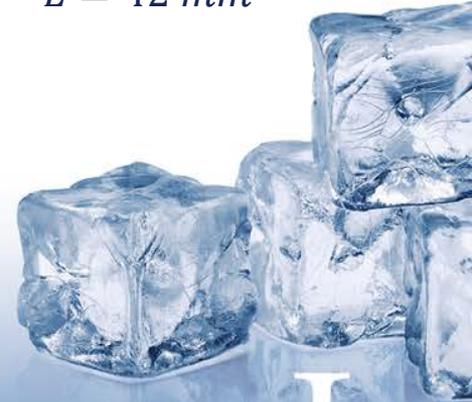
4. Numerical model

- Geometrical Model



ATS sensors:

- Cylindrical rods
- $D = 5 \text{ mm}$
- $L = 42 \text{ mm}$



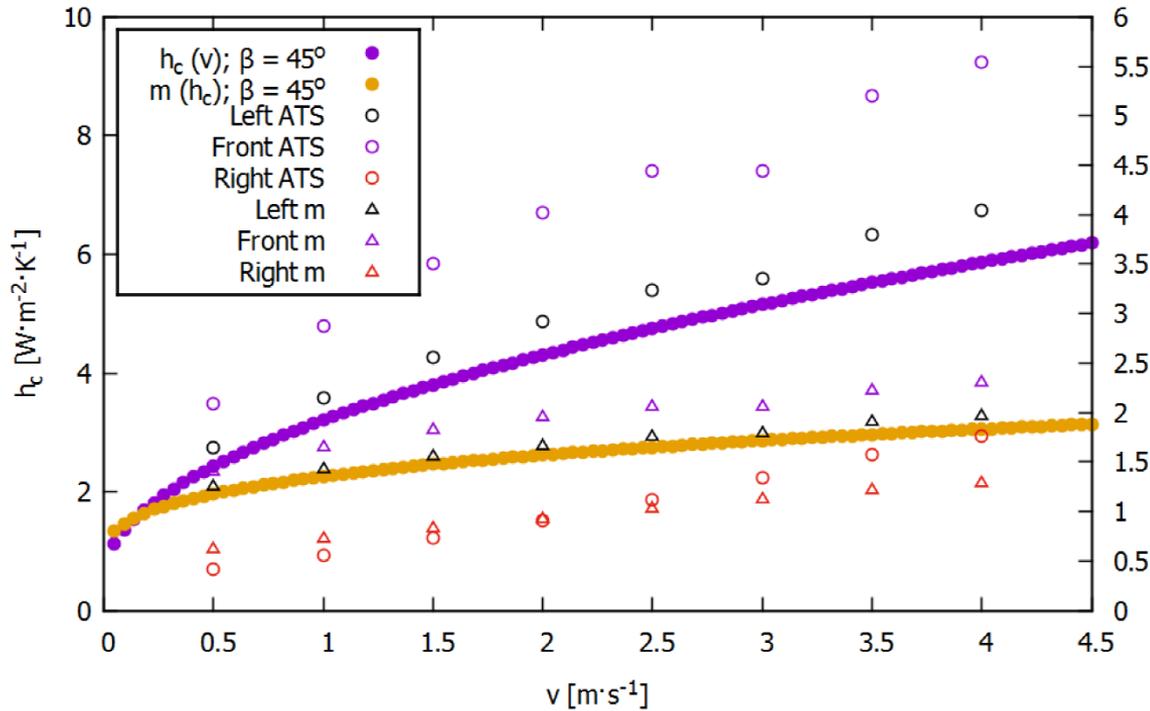
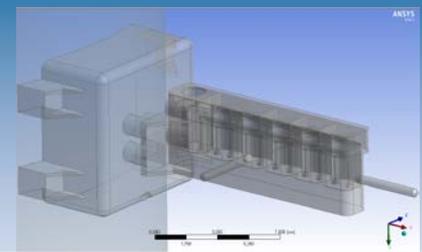
4. Numerical model

- Boundary conditions
 - Vertical pole.
 - 2 Structures: HABIT (Al) + ATS sensors (FR4).
 - Carbonate properties for Mars surface fixed at 280 K.
 - Surface Platform temperature fixed at 272.177K.
 - Radiative heat transfer considered, tacking the IR band $[0.4,0.7] \mu m$.
 - Solar load not included.
 - Emissivity and absorptivity (table).
- Study cases:
 - Deviation angle $\beta = 45^\circ$.
 - Fluid temperature $T_\infty = 250 K$ (sol 45).
 - Velocity inlet: Range $[0.5, 4] m s^{-1}$ each $0.5 m s^{-1}$.
 - Each case with and without radiative heat transfer

Surface	Emissivity ε	Absortivity α_r
Space Paint	0.88	0.2
FR4	0.9	0.3
Dolomite	0.9	0.5



5. Results

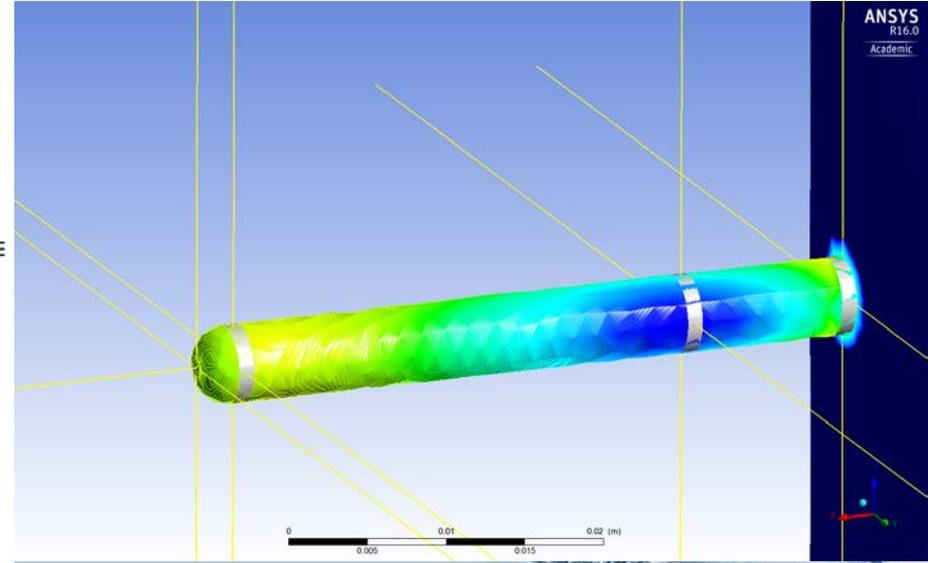
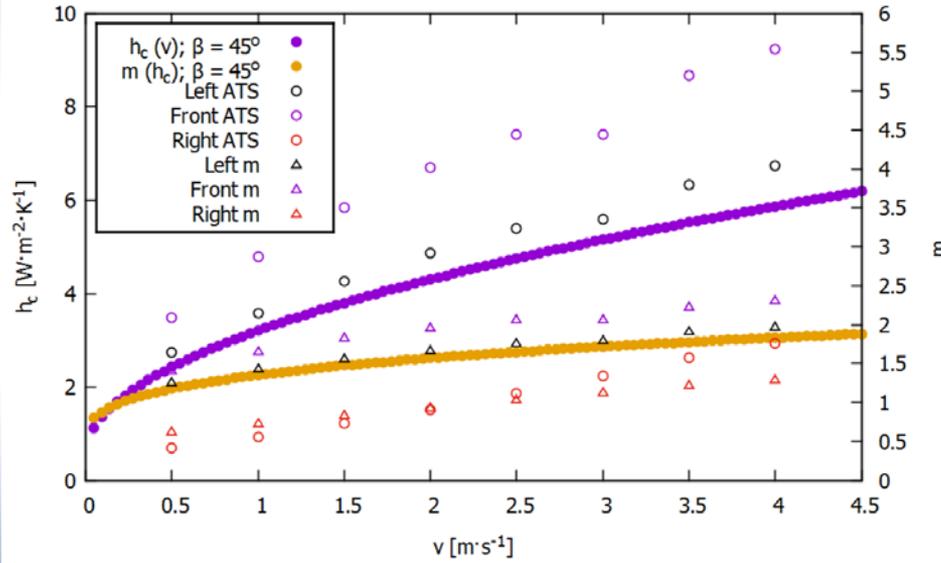


Case	v [$m s^{-1}$]	ATS	h_c [$W m^{-2} K^{-1}$]	m
1	0.5	Left	2.7375	1.2559
		Frontal	3.4820	1.4164
		Right	0.6860	0.6287
2	1.0	Left	3.5745	1.4351
		Frontal	4.7835	1.6602
3	1.5	Right	0.9335	0.7334
		Left	3.5745	1.4351
4	2.0	Frontal	4.7835	1.6602
		Right	0.9335	0.7334
5	2.5	Left	5.3980	1.7636
		Frontal	7.4055	2.0656
6	3.0	Right	1.8675	1.0370
		Left	5.5980	1.7959
7	3.5	Frontal	7.4105	2.0663
		Right	2.2410	1.1363
8	4.0	Left	6.3345	1.9104
		Frontal	8.6765	2.2359
		Right	2.6205	1.2288
		Left	6.7390	1.9705
		Frontal	9.2360	2.3068
		Right	2.9330	1.2999

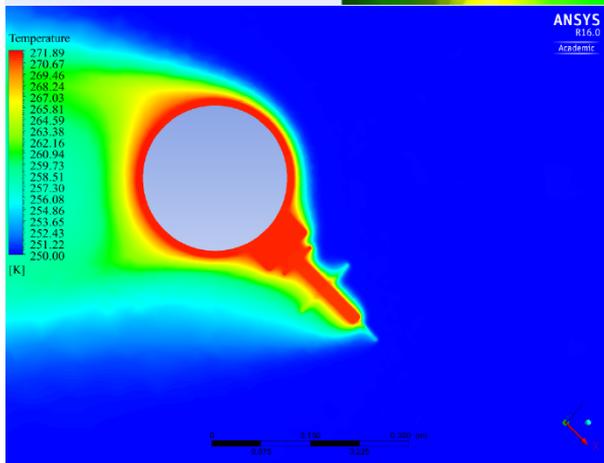
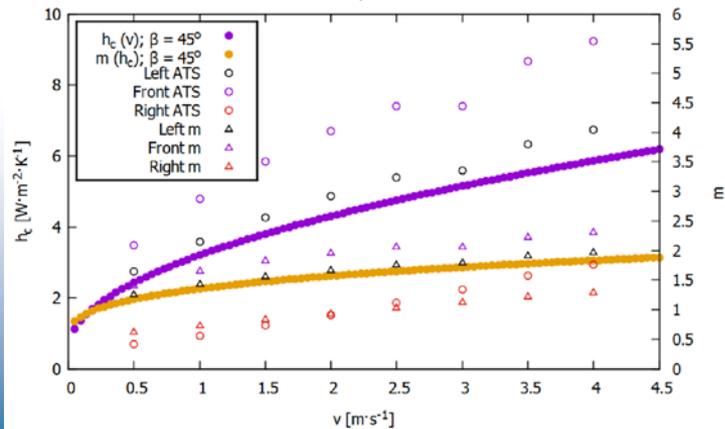
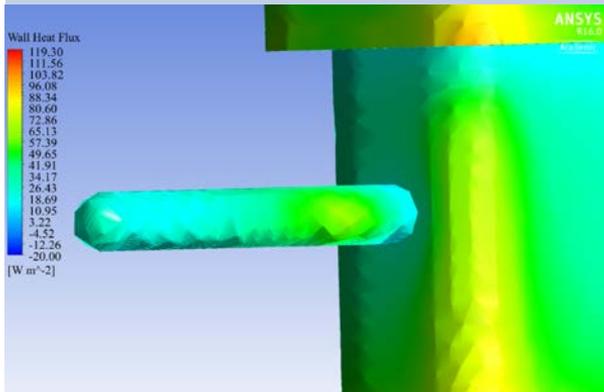
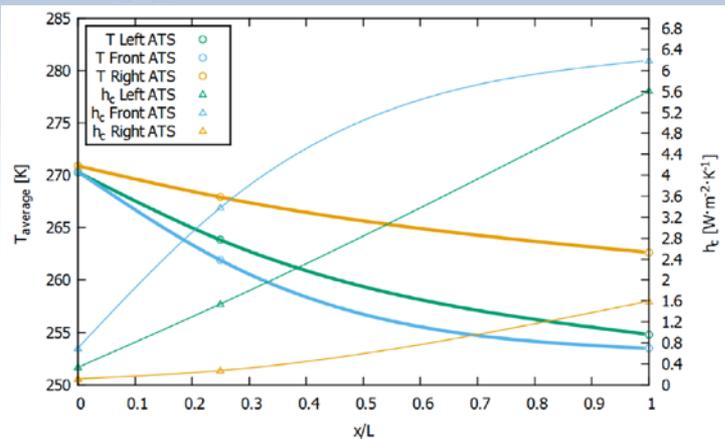


5. Results

$$h_c = \frac{\dot{q}_c}{T_s - T_\infty}$$



5. Results



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6. Conclusions

- ATS exposed to free flow show a h_c in agreement with the expected from the model.
- ATS not exposed to free flow, in the HABIT-Platform wake, has a lower h_c given that it is not exposed to forced convection completely.
- For the same set of atmospheric conditions, the convective term, and m value, increase with wind speed.
- The different values of the m parameter can be used as a proxy for orientation of the wind speed.
- This method is sensitive to the wind orientation and speed and can thus be used to retrieve information, in real time, about the wind activity and heat fluxes.
- Assumptions/plausible sources of errors :
 - Nu number for a cylinder normal to the fluid flow.
 - Temperature and h_c distributions along the ATS sensors no one-dimensional.
 - Local force convection along the sensors.
 - h_c calculated from two averaged values over control surfaces for a non-linear distribution.
- Future work: validation in the wind tunnel facility at LTU (Kiruna).



6. Acknowledgements





Thank you for your attention

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