

UNIVERSITÀ
DEGLI STUDI
DI PADOVA

Development of On Orbit Assembly technologies to enable spacecraft servicing

Federico Basana - 37th Cycle

Supervisor: Dr. Francesco Branz

Meeting - 17/09/2024

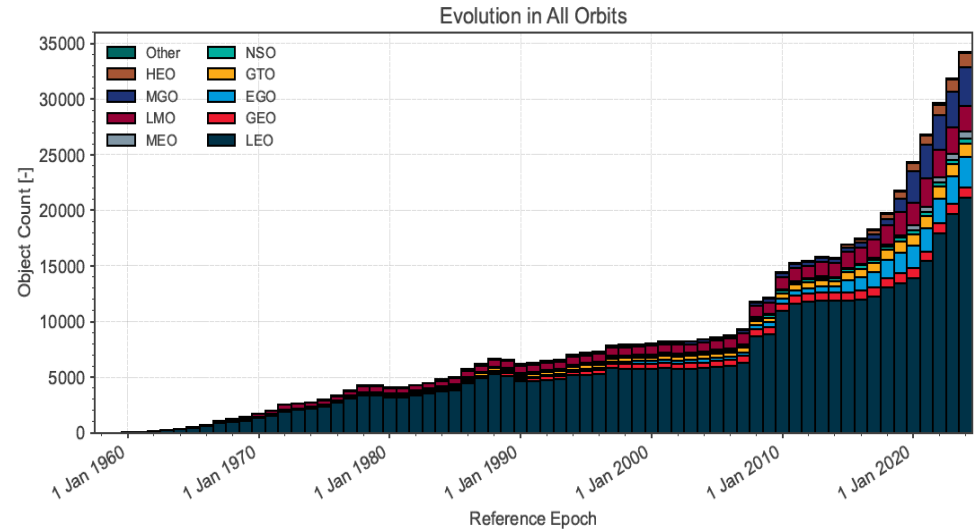
Space debris pose significant risks to
functioning satellites



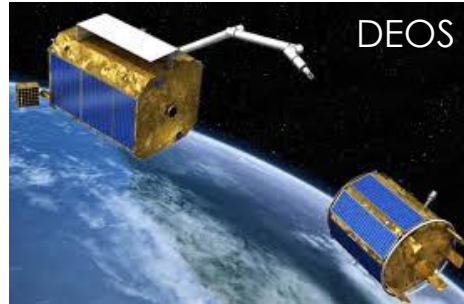
Their number is expected to
grow in the next future



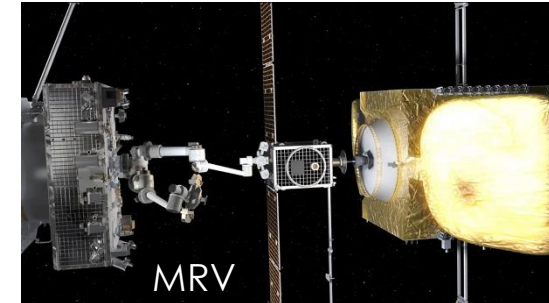
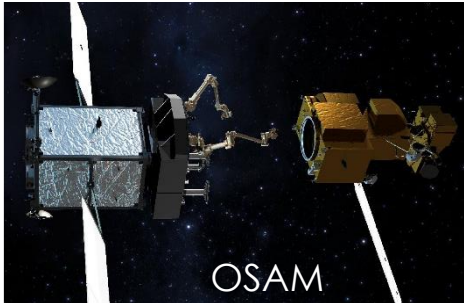
Space Surveillance and Tracking
Space Situational Awareness
Collision Avoidance Manoeuvres

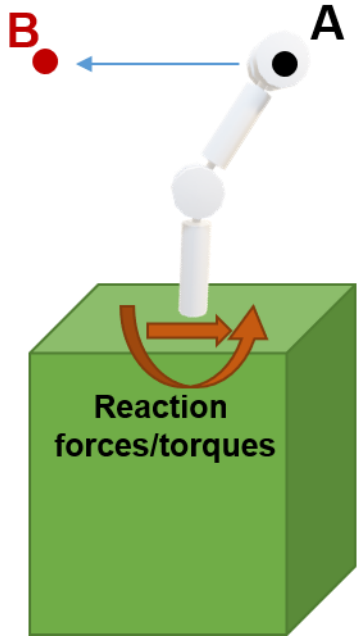


Why don't we remove them?

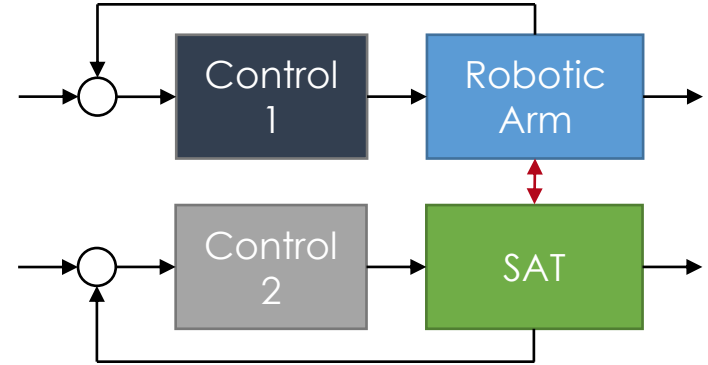


Why don't we repair them?

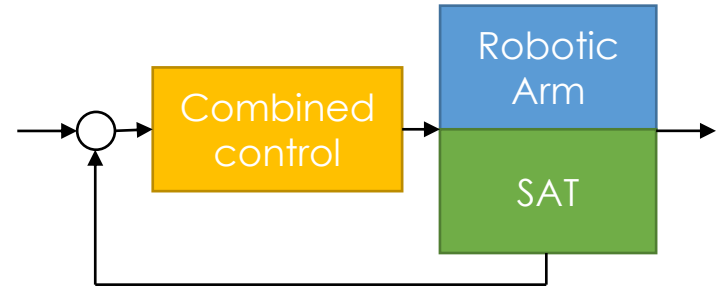




Decoupled control



Combined control



Strategies to reduce the base torque
IF redundant manipulators are used



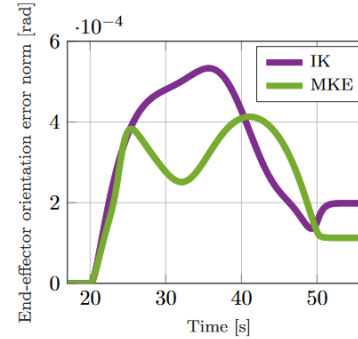
The MKE locally minimize the kinetic energy
of the manipulator



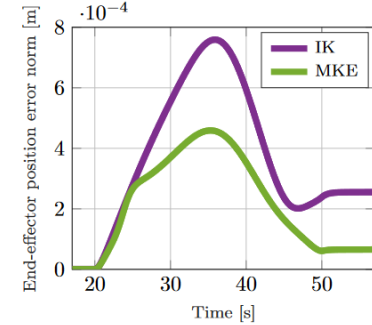
Comparison with classic Inverse Kinematics
(IK) algorithm



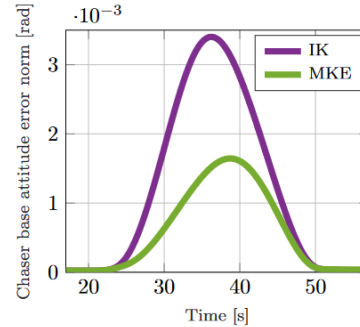
Case study: 7-DoF free-flying robotic arm
performing different trajectories



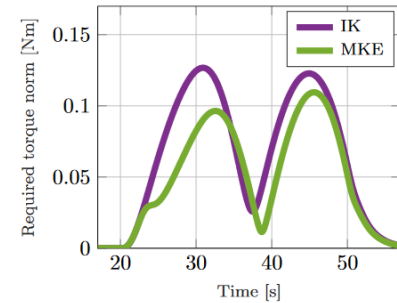
(a) End-Effector orientation error norm



(b) End-Effector position error norm



(c) Satellite attitude error norm



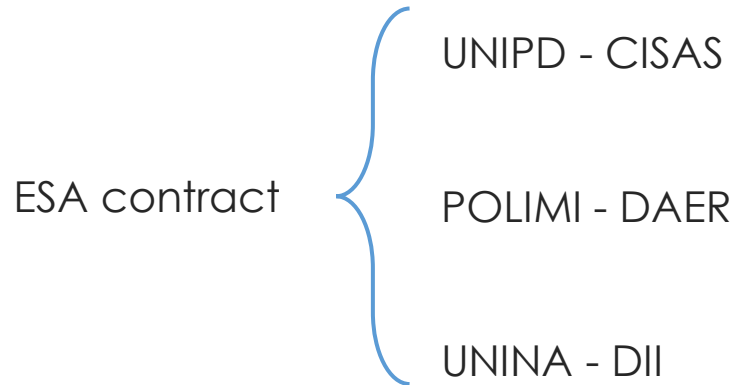
(d) Required torque norm

1. F. Basana, F. Branz, Simulation of robotic space operations with minimum base reaction manipulator, Journal of Space Safety Engineering (JSSE), 6/07/2022, <https://doi.org/10.1016/j.jsse.2022.06.005>
2. F. Basana, F. Branz, A simulation tool for robotic active debris removal with minimum reaction space manipulator, 72nd International Astronautical Congress (IAC), Dubai, United Arab Emirates, 25-29 October 2021

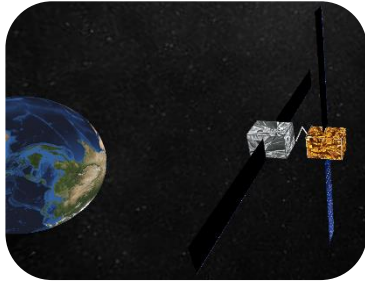
Study for developing innovative GNC solutions for IOS/ADR missions

Adoption of:

- a multi-variable combined control approach;
- a fully autonomous navigation architecture based on active or passive EO sensors.



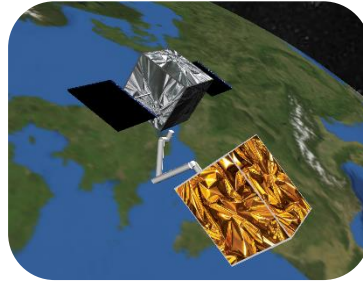
Scenario 1



The target (large telecom satellite) is:

- operative,
- controllable
- capable to receive a refuel/update/repair

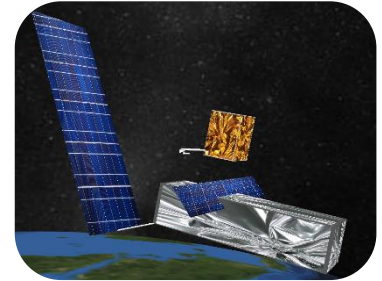
Scenario 2



The target (One web) is tumbling and is equipped with:

- a grapple fixture
- fiducial markers across the spacecraft body

Scenario 3

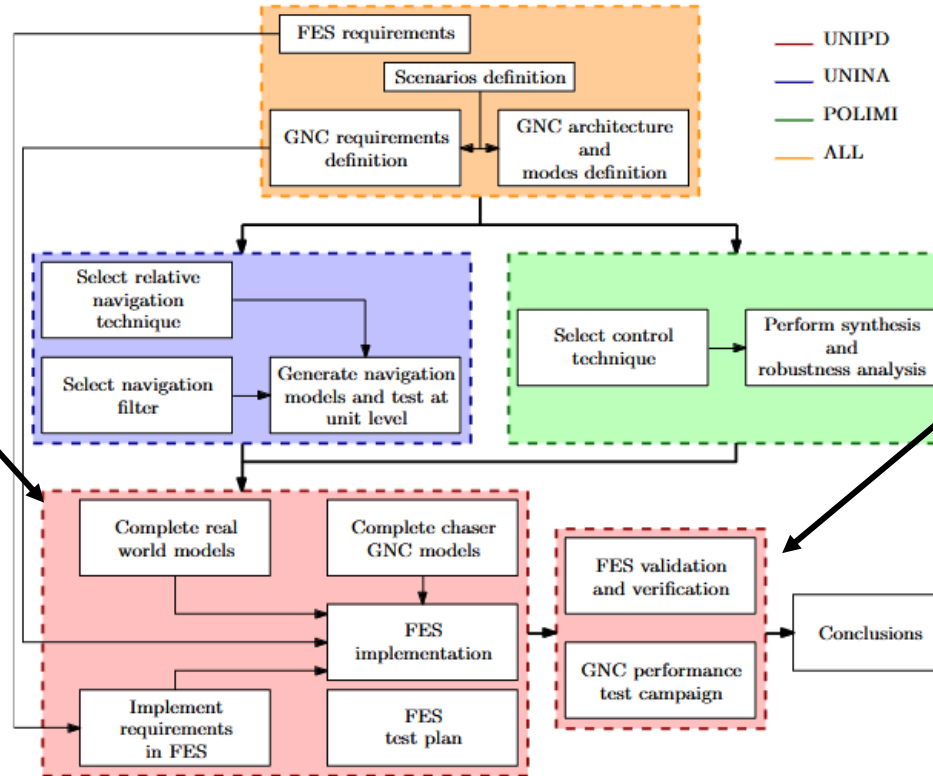


ENVISAT is captured. The chaser is synchronized with the motion of the non-cooperative target which is spinning at a rate of 5 deg/s.

In all the scenarios, the capture and the post-capture stiffening of the 7-DoF robotic arm phases are considered

FES implementation

- simplified models of the two satellites
- environmental disturbances
- propellant sloshing
- real actuators and sensors behaviour

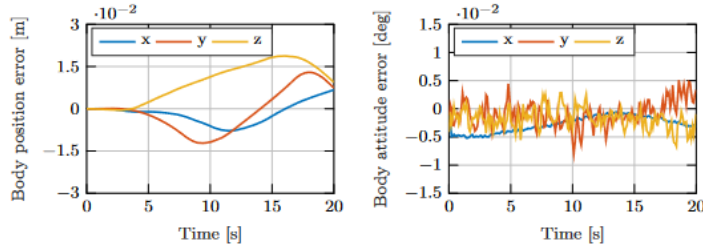


Test campaign

- FES validation and verification
- Nominal simulations
- Error budget analysis
- Monte-Carlo analysis

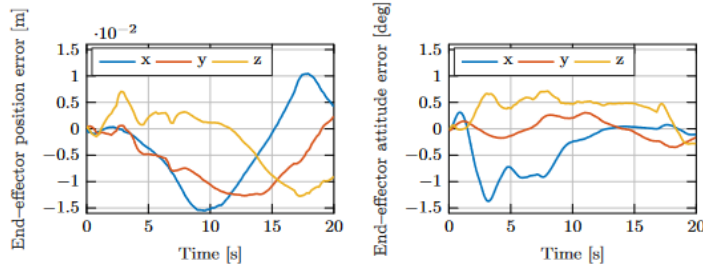
Examples of results – SC2

Phase A



(a) Chaser body position error.

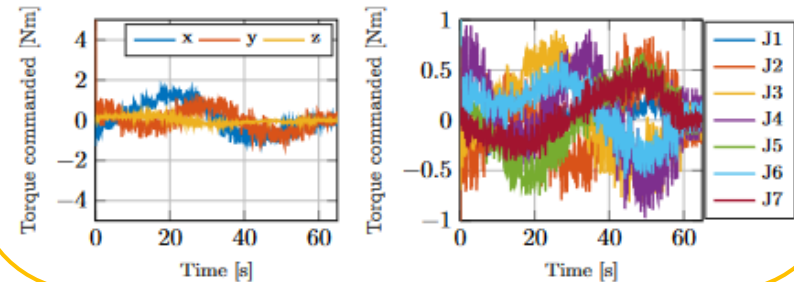
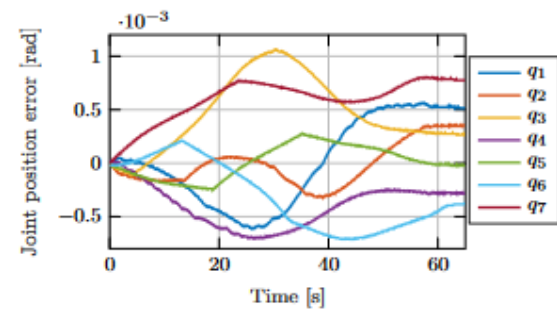
(b) Chaser body attitude error.



(c) End-effector position error.

(d) End-effector attitude error.

Phase B



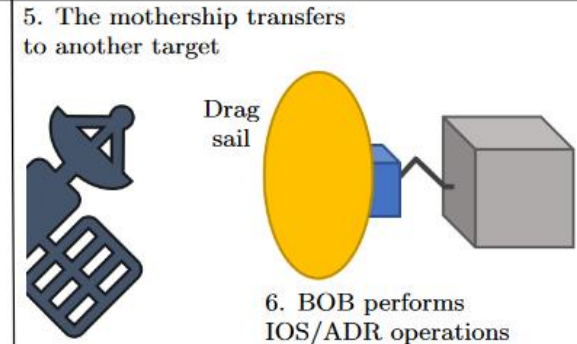
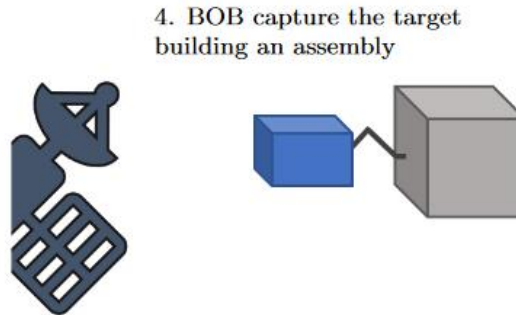
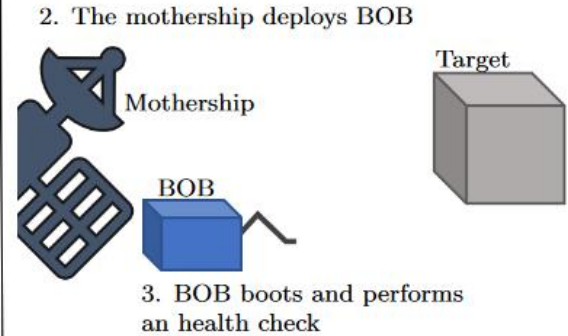
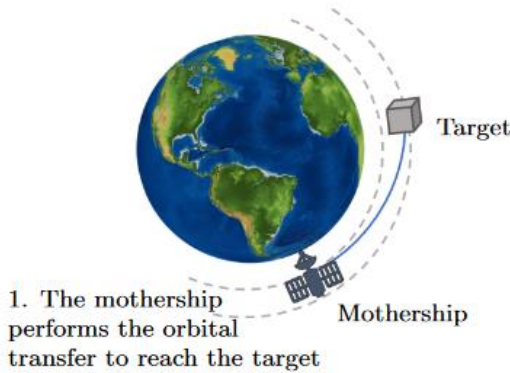
Main problems of IOS/ADR missions:

- Complexity
- Cost
- Collision risk

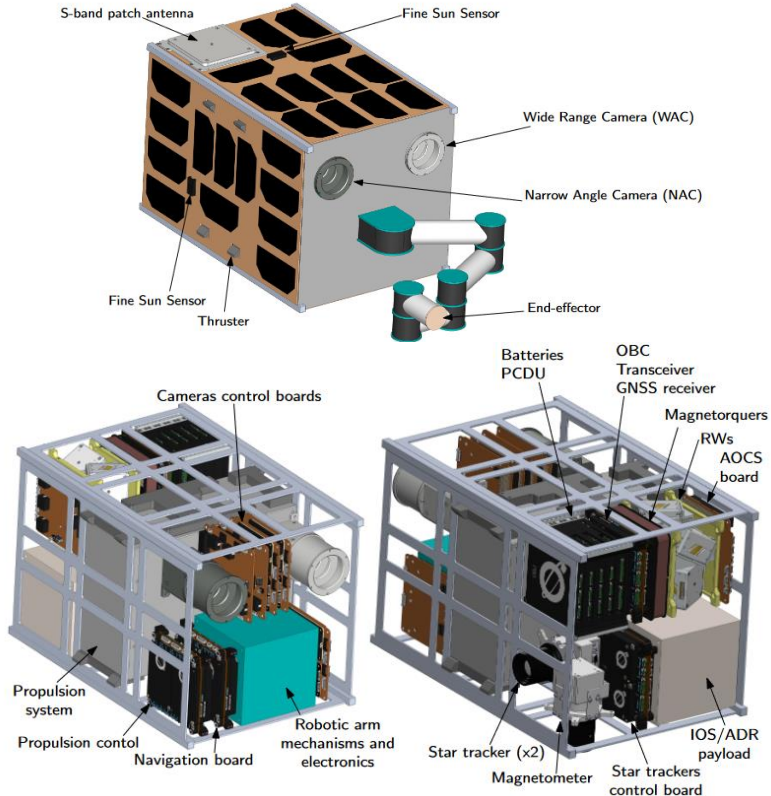
Employ a CubeSat-sized servicer equipped with a robotic arm

Advantages:

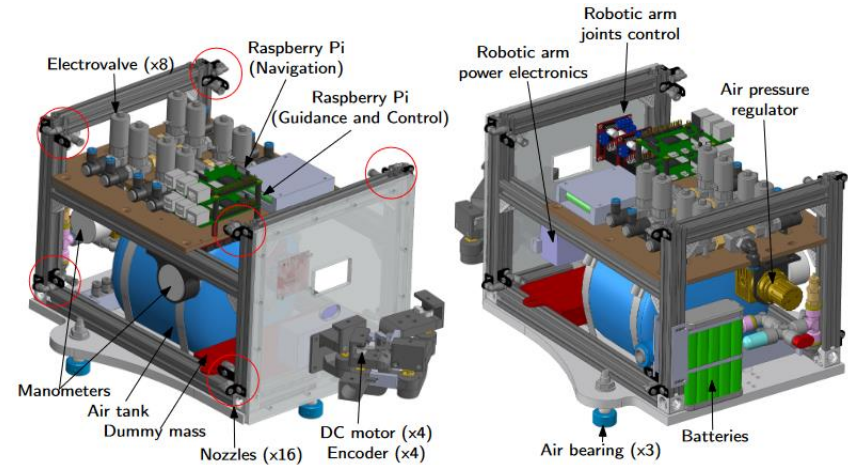
- Fast development
- Cheap
- Less severe effects in case of collision



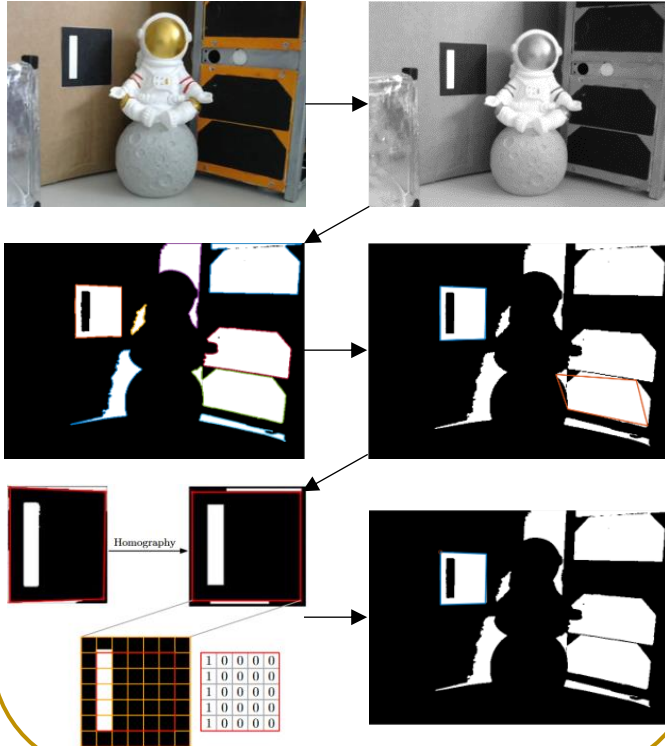
Space – Basic Orbital Butler (BOB)



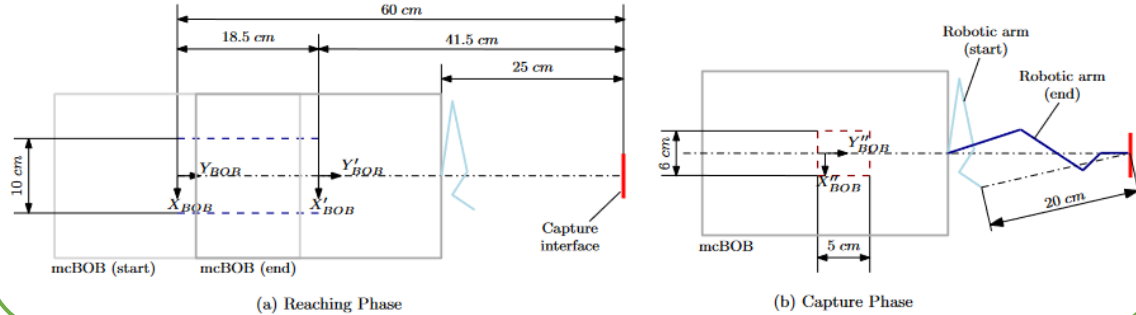
Space – BOB mock-up (mcBOB)



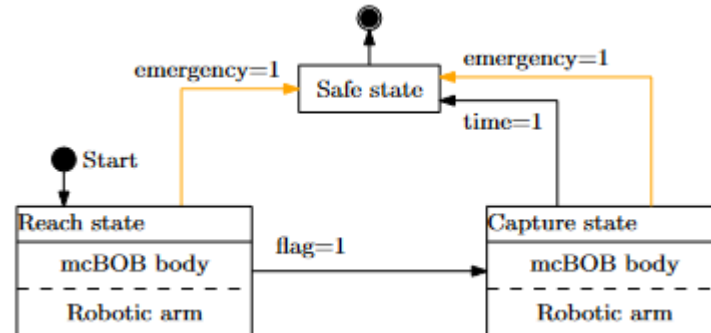
Navigation



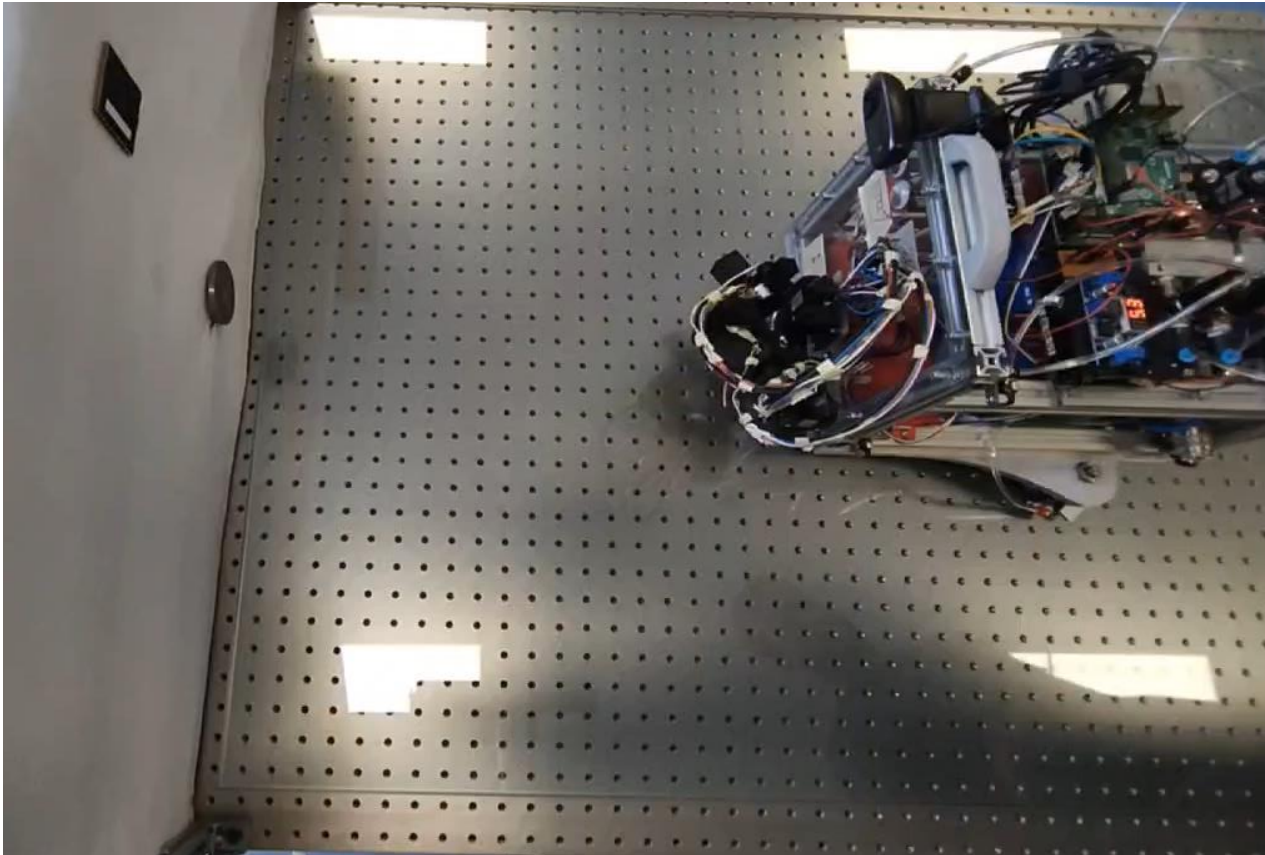
Guidance



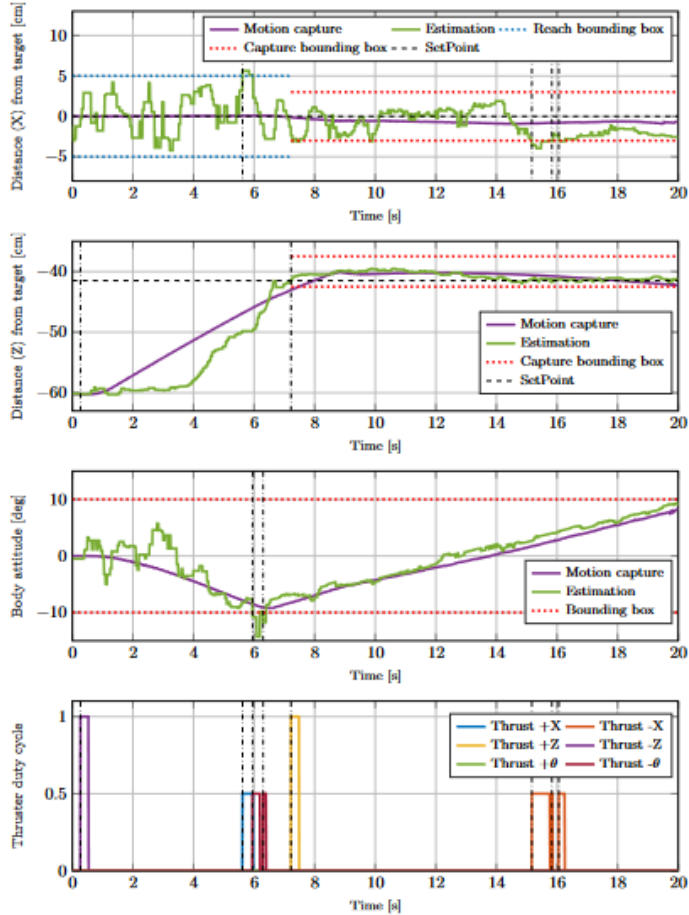
Control



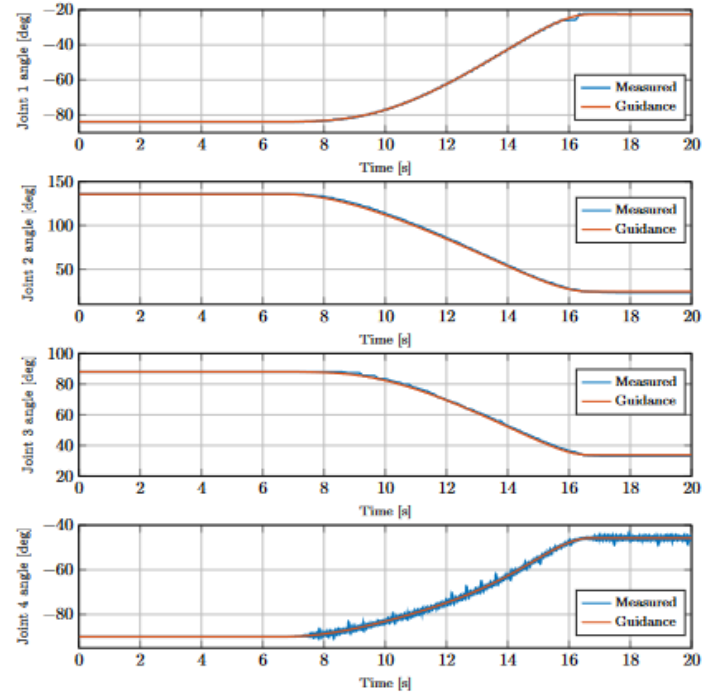
A new approach for servicing satellites



mcBOB body



mcBOB robotic arm





AlbaSat is a **2U CubeSat** developed by a **student team** under the ESA's Fly Your Satellite! – Design Booster programme



FLY YOUR SATELLITE!

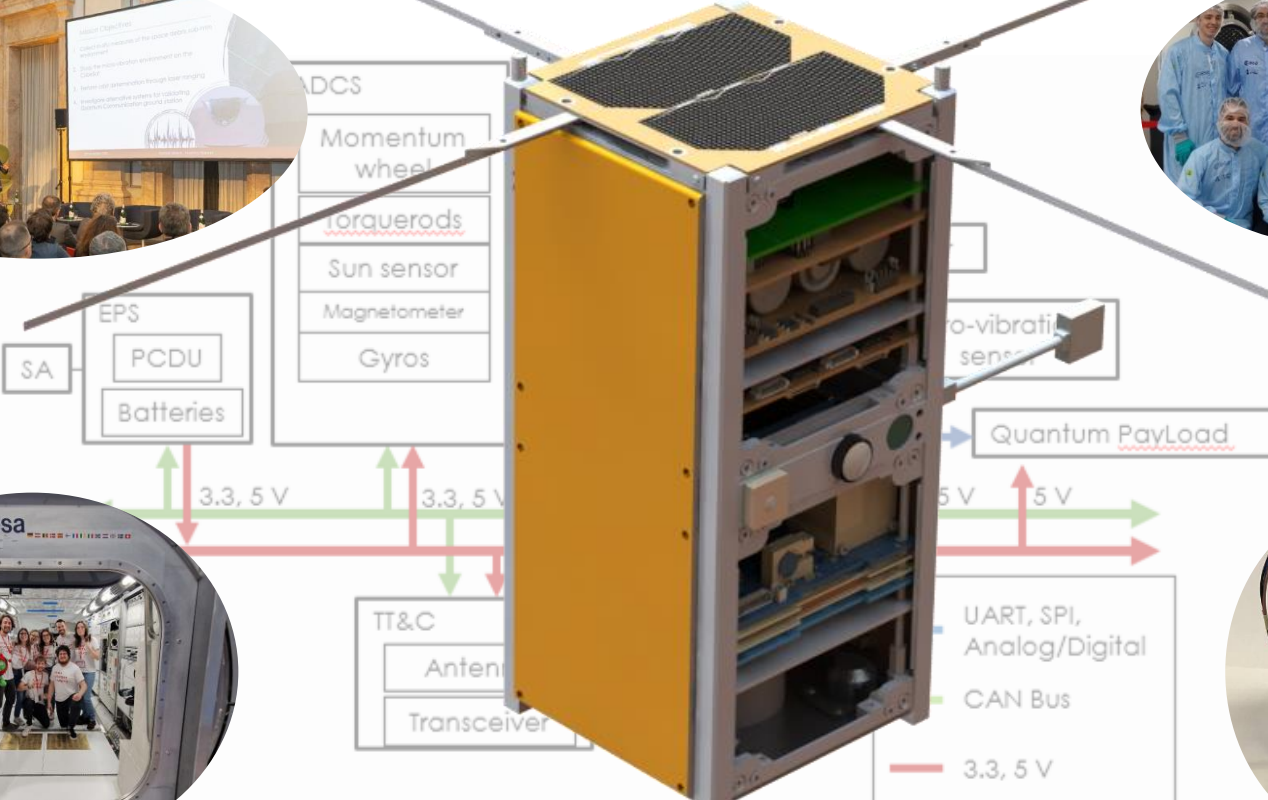
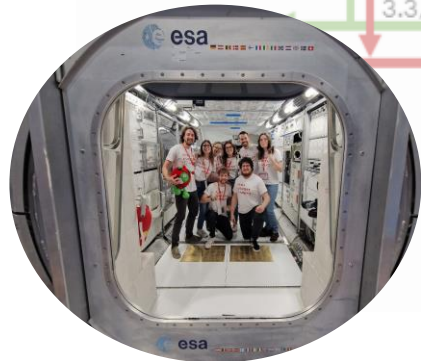
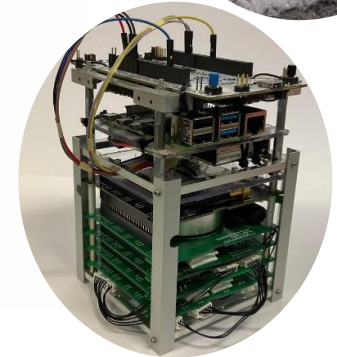
Mission objectives:

1. Collect in-situ measurements of the space debris environment
2. Study the micro-vibration environment on the satellite
3. Perform orbit determination through laser ranging
4. Investigate alternative systems for possible Satellite Quantum Communication applications on nanosatellites


AlbaSat successfully concluded the PDR and the FYSI-DB programme in May 2024

Four payloads:

1. Impact Sensor (IS)
2. Micro Vibration Sensor (MVS)
3. Corner Cube Retroreflectors (CCR)
4. QuantumFuture PayLoad (QPL)



The work investigated new solutions for IOS/ADR missions.

- Investigations of methods for decoupling robotic arm motion from satellite body
 - Development of a fully representative simulator for CPOs
 - V&V and test of GNC combined control strategy with EO sensors
 - Proposal and validation of a new approach for servicing satellites
- 
- Improvement of existing technology
 - Enhance performance and robustness of IOS/ADR missions
 - Increase of mission safety

Thank for the attention



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