

Optical and Opto-mechanical Analysis and Design of the Telescope for the Ariel Mission

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Mission logo and artist's rendering of the Ariel spacecraft from Phase A. Credit: ARIEL/Science Office

1. Introduction – The Ariel ESA Mission





The Ariel Mission

 4th medium-class mission of ESA Cosmic Vision program to survey exoplanet atmospheres through infrared transit spectroscopy

1. Introduction

 Payload. An all-aluminum telescope feeding two instrument modules: a spectrometer (1.95–7.8 µm), and a combined fine guidance system/visible photometer/NIR spectrometer



Frame from the concept video of Ariel detecting light from a distant star <u>https://arielmission.space/</u> Credit: ESA / STFC RAL Space / UCL / UK Space Agency / ATG Medialab



1. Introduction









Mission Schedule



2. Summary of Main Personal Contributions





PhD Dissertation in paper collection form

- 1. Chioetto, P. et al. "Preliminary Analysis of Ground-to-Flight Mechanical Tolerances of the Ariel Mission Telescope" Proc. SPIE (2022) doi:10.1117/12.2628900
- 2. Chioetto, P. et al. "Qualification of the Thermal Stabilization, Polishing and Coating Procedures for the Aluminum Telescope Mirrors of the ARIEL Mission" Exp. Astr. (2022) doi:10.1007/s10686-022-09852-x
- 3. Chioetto, P. *et al. "The Primary Mirror of the Ariel Mission: Cryotesting of AluminumMirror Samples with Protected Silver Coating"* Proc. SPIE (2020) doi:10.1117/12.2562548
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- 5. Chioetto, P. *et al. "Long Term Durability of Protected Silver Coating for the Mirrors of Ariel Mission Telescope"* submitted to ICSO 2022
- 6. Chioetto, P. *et al. "Initial Estimation of the Effects of Coating Dishomogeneities, Surface Roughness and Contamination on the Mirrors of Ariel Mission Telescope"* Proc. SPIE (2021) doi:10.1117/12.2603768

Telescope Tolerance Analysis

> Mirrors Technological Development

Mirrors Coating Qualification

Telescope Throughput Analysis 3. Telescope Opto-mechanical Design and Tolerancing

3. Telescope Opto-mechanical Design





Optical Design

- Off-axis, unobscured, afocal Cassegrain
- Collecting area: 0.6 m²
- FoV: 30" (diff. limited), 50" (unvignetted)
- Wavefront Error: diff. limited at 3 μm
- Avg. throughput: >0.82
- Angular magn.: 55

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Telescope Assembly Structure (interim design)



Highlights: all-aluminum telescope, large primary mirror (1.1 m x .7 m optical aperture, first time flying in a space mission), refocusing M2 mechanism.



3. Telescope Opto-mechanical Design



Analysis of the Effects of Mirrors Surface Errors

- Each mirror is represented as sum of Zernike polynomials
- Inverse sensitivity analysis to find initial tolerances
- Monte Carlo analysis with Zernike coefficients sampled randomly from initial tolerances
- Telescope performance (PSF Encircled Energy) computed for each random realization
- Statistical analysis of results

Zernike polynomials on elliptical aperture (left) and coefficients from inverse sensitivy analysis (right)



Surface error RMS (nm)





Analysis of the Effects of Mirrors Displacements

- **Statistical analysis** of the **effects of rigid body motions** of the four telescope mirrors (caused eg. by dynamic loads during launch)
- Optical performance evaluated in terms of **Enclosed Energy in the PSF**, and position and orientation of the **exit beam**.



3. Telescope Opto-mechanical Design



Analysis Procedure

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- 1. Inverse sensitivity: determine the range of admissible misalignments for each element individually, based on the acceptable ranges (acceptance criteria).
- 2. Monte Carlo simulation on the ranges identified in Step 1 to evaluate the combined effect.

Results

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Reducing the perturbation range to $\pm 10 \ \mu m$ for linear shifts and ± 10 " for rotations brings most of the Monte Carlo cases (>95 %) within requirements. These ranges are deemed as realistic by the telescope manufacturing Prime Contractor.



	beam shift (mm)		beam tilt (arcsec)		EE lost	
	x	y	x	y		
min	0	0	0.1	0.4	4.15%	
50%	0.021	0.023	45.0	45.0	4.18%	
75%	0.034	0.039	45.0	45.0	4.20%	
90%	0.048	0.051	45.0	45.0	4.23%	
95%	0.055	0.057	45.0	45.0	4.66%	
\max	0.061	0.065	45.3	46.4	6.54%	

4. Telescope Mirrors Technological Development and Prototyping





Objective

Develop aluminum Al6061-T651 mirrors manufacturing technologies to **TRL 6**: "System/subsystem model or prototype demonstration in a relevant environment (ground or space)" (ISO 16290:2013)

Materials

- 1. 25 mm sample disks
- 2. 150 mm sample disks
- 3. Full-size primary mirror demonstrator (PTM)

Processes

- 1. Thermal Stabilization
- 2. Machining and Polishing
- 3. Optical Coating



PTM and 150 mm disks in climatic chamber. Credit: TAG S.r.l. / Media Lario S.r.l.

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4. Telescope Mirrors Technology Dev.



Substrate Thermal Stabilization

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- Improves cryogenic opto-mechanical stability reducing residual stress
- Customized recipe based on the work of R. G. Ohl et al. from NASA/Goddard, consisting of quenching and temperature cycling steps
- Developed on the 150 mm disks





Evolution of the interferometric measurements on the first 150 mm disk during thermal stabilization (credit: Media Lario S.r.l)

150 mm samples being loaded in a cryochamber (credit: G. Morgante/INAF OASBO)

4. Telescope Mirrors Technology Dev.



Machining and Polishing - 1

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- Final shape approximated to within 1 µm RMS with fly-cutting and diamond turning (SPDT) technologies
- Achieving final specifications of shape (80 nm RMS) and surface roughness (10 nm RMS) with several runs of deterministic polishing technology

Zeeko IRP 1200X robotic polishing machine (left) and detail of the polishing bonnet (right). Credit: MediaLario S.r.l.



Fly-cutting (left) and SPDT (right) machines. Credit: LT Ultra-Precision Technology GmbH







Machining and Polishing – 2

- Challenges due to material specificities (Al 6061-T651 rolled plate)
- Mg and Si agglomerates form during forging and treating, and scratch the surface during polishing
- Process development ongoing alternating figuring and superpolishing steps with carbon-based slurries



	ISO 25178	
	Height Parameters	
	Sq	21.0 nm
A State of the second	Sp	48.2 nm
a start	Sv	99.3 nm
A second second	Sz	148.0 nm
	Sa	15.7 nm
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Profilometry image of sample after a step of aggressive figuring polishing highlight agglomerates. Credit: Media Lario S.r.l.



Spectrum	0	Mg	Al	Si	Fe	Total			
1	15.15	19.27	43.36	22.22		100.00			
2			76.87		23.13	100.00			
3	14.10	18.92	48.32	18.66		100.00			
4	18.45		66.68	14.87		100.00			

Processing option: all elements analyzed (normalized) All results in weight%

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40 um

5. Mirrors Optical Coating Qualification and Performance Assessment





Telescope Mirrors Optical Coating Qualification Campaign

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- **Protected Ag coating** with space heritage from CILAS-ArianeGroup. Deposition through magnetron sputtering
- Additional **qualification** required because of **substrate size** and cryogenic operating environment
- Qualification campaing on 25 mm disks on holder shaped as primary mirror
- Verification tests on additional samples and PTM
- Ageing monitoring



Photo and schematic drawing of the disks on the samples holder ready for coating. Credit: CILAS-ArianeGroup



Photo of the PTM and samples ready for coating. Credit: CILAS-ArianeGroup



5. Optical Coating Qualification



Qualification Test Program

Environmental

- Humidity: 24 hours, 90% humidity, 55 °C
- Temperature: 30 cycles, -40°C / 70°C, <2°C/min (accelerated aging)
- Cryogenic: 10 cycles, 54 K / 293 K, <5 K/min

Mechanical/chemical

- Abrasion test with cheescloth
- Adhesion test through tape stripping
- Cleaning test: CILAS proprietary cleaning procedure, ethanol and acetone solutions

Verification methods

- Visual inspection, to check coating integrity
- Spectral reflectance, to verify unaltered performance
- Additional inspection/imaging with Optical and Atomic Force Microscopes (AFM)

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Schematic representation of the coating layers (not to scale)

5. Optical Coating Qualification



Tests Results

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Measurements performed after each test, and periodically after qualification

- no visible signs of degradation
- no significant reduction in performance (reflectance)



Comparison of visual inspection pictures of a sample right after coating (left) and after testing and storage (right)



Darkfield photograph of a sample after storage

5. Optical Coating Qualification



Atomic Force Microscopy

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- Imaging of one sample immediately after coating (top) and after tests and storage (bottom)
- Morphology is qualitatively unaltered and roughness is unchanged



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Update of the Telescope Throughput Model

- Evaluation of the expected end-oflife throughput performance of the telescope
 - using measured reflectivity data from coating qualification
 - scattering losses from surface roughness and particulate contamination
 - absorption losses from molecular contamination

Single mirror reflectivity, Baseline and EOL







Sources of funding

- Implementation Agreement n. 2018-22-HH.0 of the ASI-INAF Framework Agreement "Participation to the B1 study phase of the Ariel mission")
- ESA CSL/INAF contract "Cryotesting of Ariel M1 mirror and coating process qualification"
- Implementation Agreement n. 2021-5-HH.0 of the ASI-INAF Framework Agreement "Italian Participation to Ariel mission phase B2/C".

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Mission Credits

- **PI**: Prof. G. Tinetti, University College London, UK
- **Mission Scientist**: Prof. E. Pascale, Sapienza University of Rome
- Italian PM: Prof. E. Pace, University of Florence

Artist's impression of Ariel on its way to L2. Credit: ESA / STFC RAL Space / UCL / Europlanet-Science Office.

Italian Co-PIs: Dr. G. Micela, INAF-OAP; Dr. G. Malaguti, INAF-OASBO

6. Credits



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Thank you for the attention



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