



# POLARIS EXPERIMENT: DATA COLLECTED DURING THE STRATOSPHERIC FLIGHT ON THE BALLOON BEXUS18

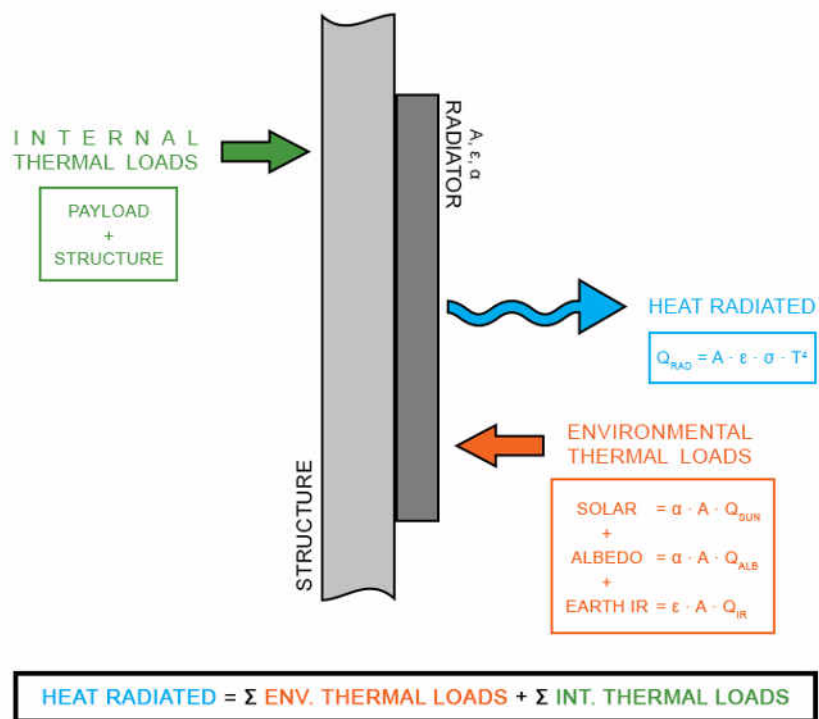
D. Paganini, C. Cacco, F. Cipriani, F. Cocco, T. Cortese, R. Dalla Vecchia, M. La Grassa, M. Lora, M. Zorzan  
F. Branz, L. Olivieri, F. Sansone, A. Francesconi

# PRESENTATION OVERVIEW

1. RADIATOR CONCEPT
2. POLARIS EXPERIMENT
3. SETUP EVOLUTION DURING RX/BX PROGRAMME
4. FLIGHT DATA
5. LESSONS LEARNED & CONCLUSIONS

# RADIATOR CONCEPT

Heat radiators are designed to manage the heat of P/L Internal Generation



Variable P/L activity and Environment heat fluxes require **FLEXIBILITY & CONTROL** on **RADIATORS PERFORMANCE**

# RADIATOR CONCEPT

The radiator is composed by three metallic plates linked together so that an actuation system can put them in good thermal contact or separate them



TWO GEOMETRY  
CONFIGURATIONS

# RADIATOR CONCEPT

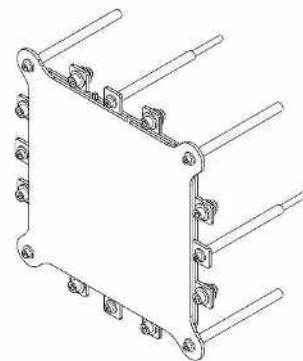
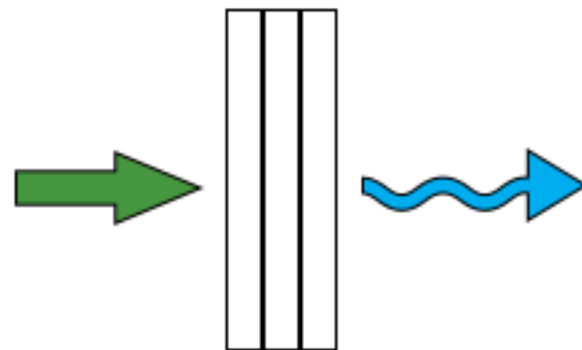
The radiator is composed by three metallic plates linked together so that an actuation system can put them in good thermal contact or separate them

## TIGHENED CONFIGURATION

CONDUCTIVE LINK  
BETWEEN THE PLATES

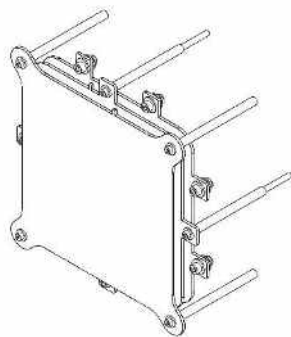
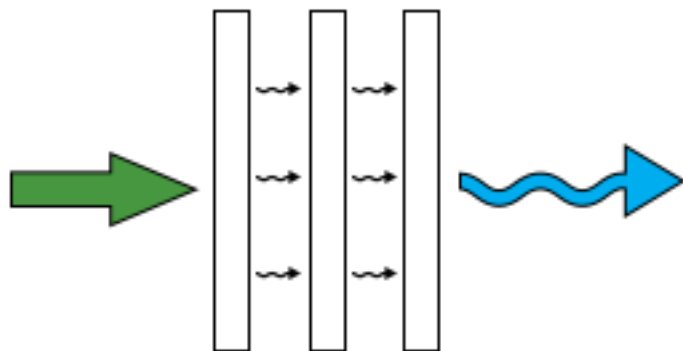
LOWER EQUIVALENT  
THERMAL RESISTANCE

MAXIMUM HEAT EXCHANGE



# RADIATOR CONCEPT

The radiator is composed by three metallic plates linked together so that an actuation system can put them in good thermal contact or separate them



**OPEN CONFIGURATION**

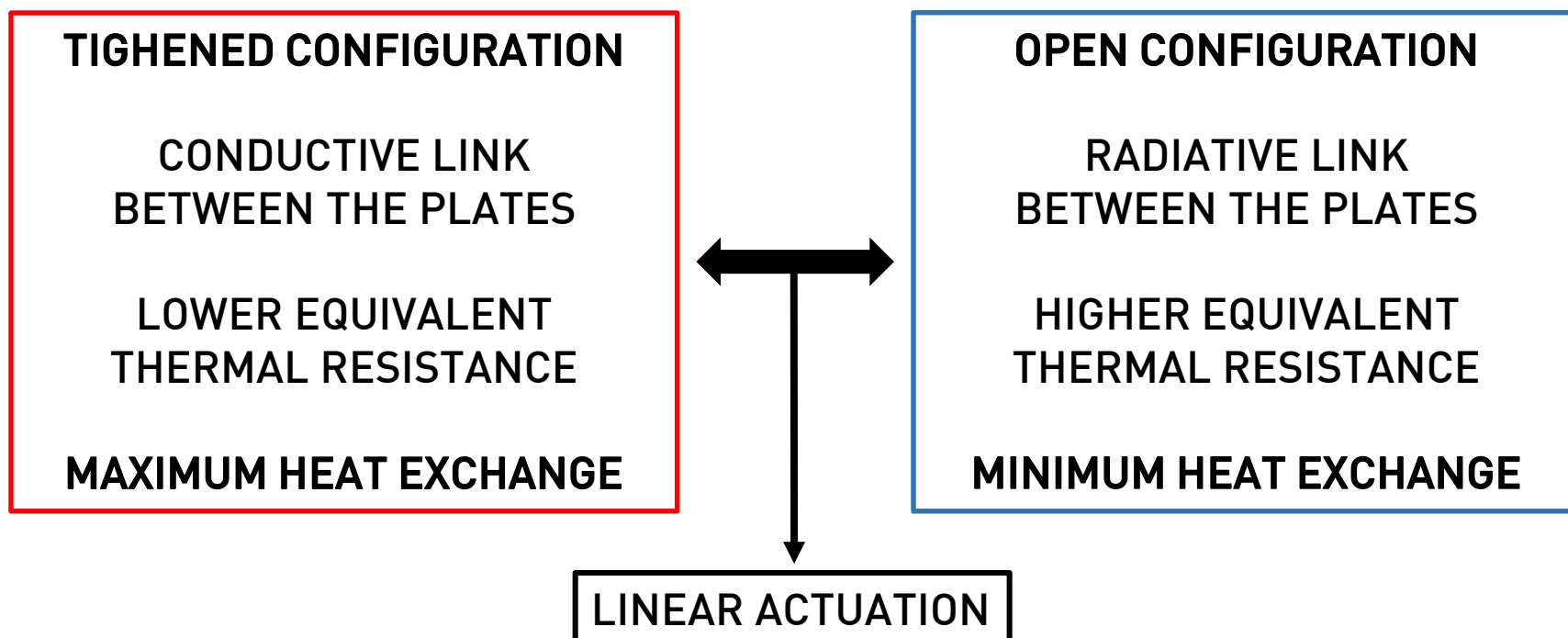
**RADIATIVE LINK  
BETWEEN THE PLATES**

**HIGHER EQUIVALENT  
THERMAL RESISTANCE**

**MINIMUM HEAT EXCHANGE**

# RADIATOR CONCEPT

The radiator is composed by three metallic plates linked together so that an actuation system can put them in good thermal contact or separate them



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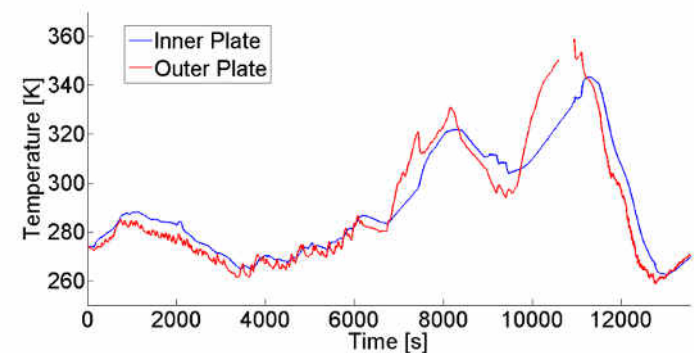
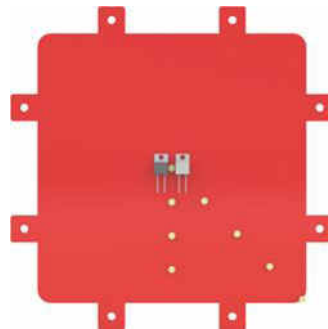
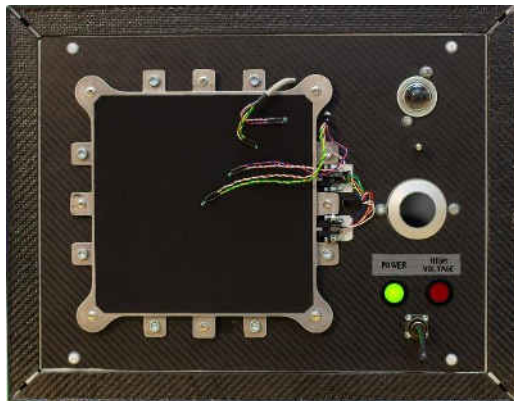
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# POLARIS EXPERIMENT

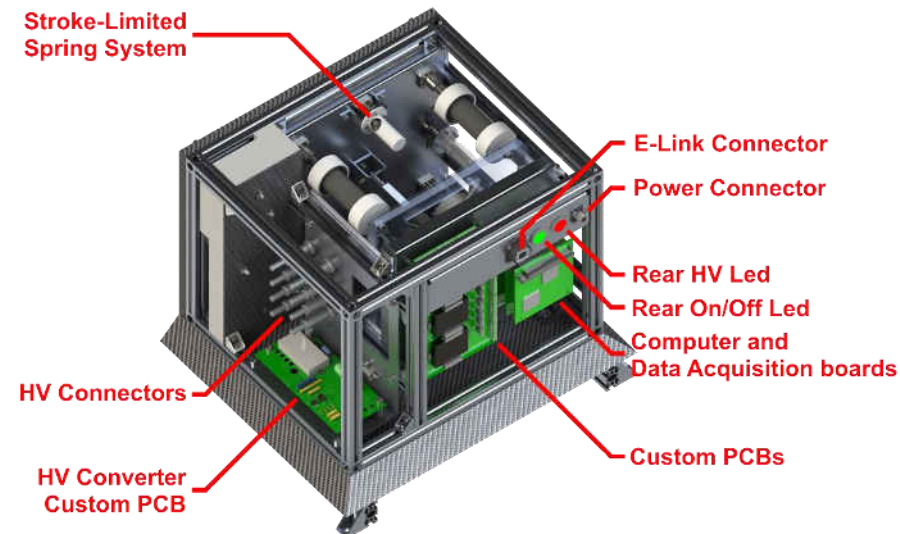
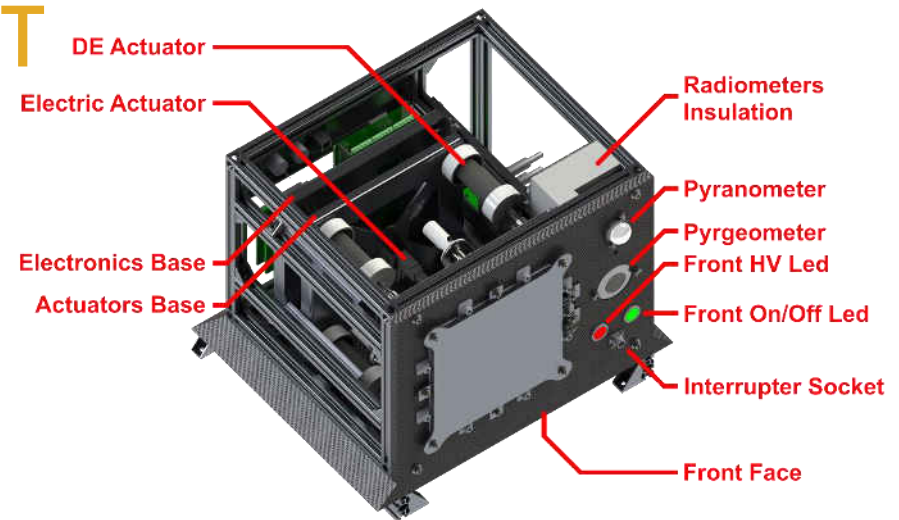
## POLymer Actuated RAdiator with Independent SURfaces

1. Test a new concept of heat radiator in variable environmental conditions
2. Control the radiator to guarantee the thermal steadiness of a dummy payload
3. Compare the numerical thermal model with the measured data

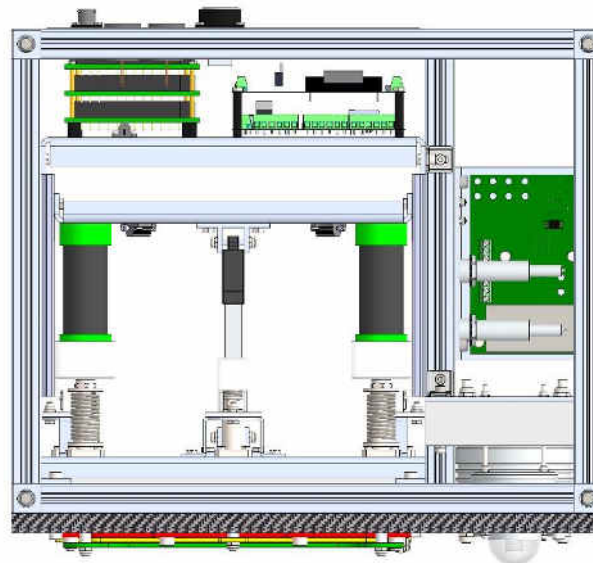
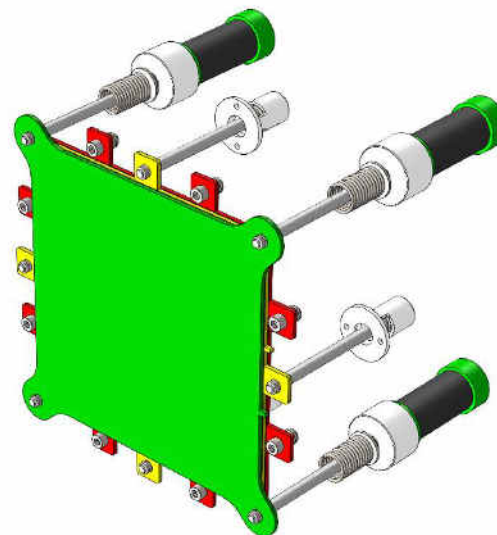
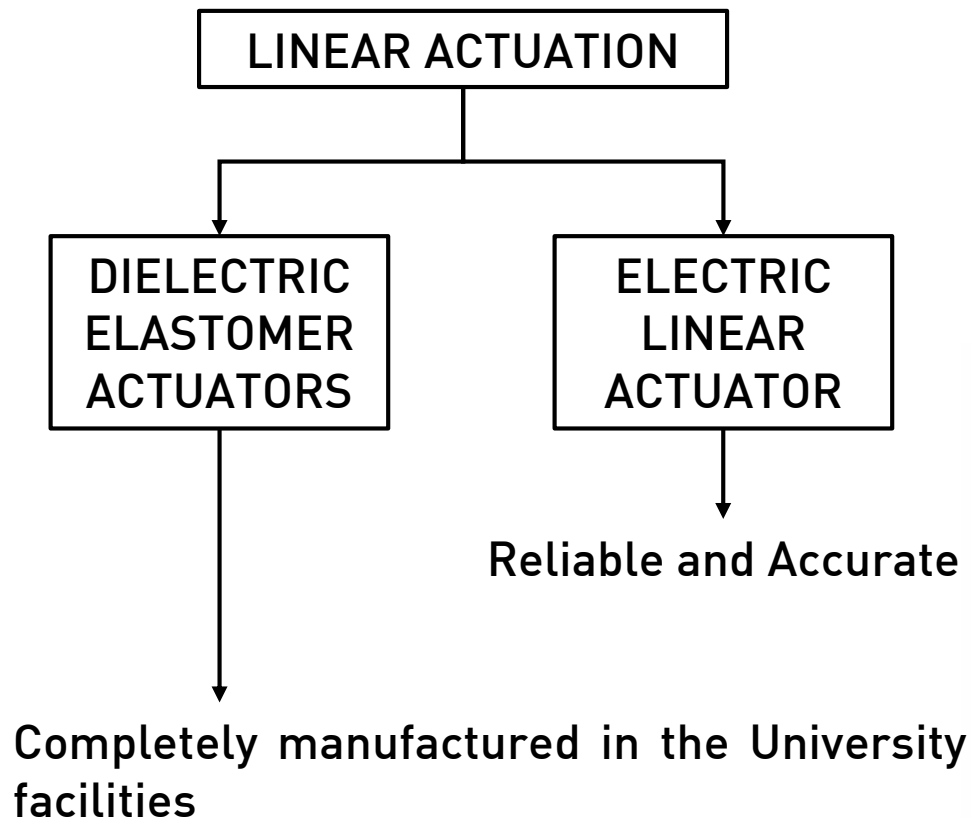


# POLARIS EXPERIMENT

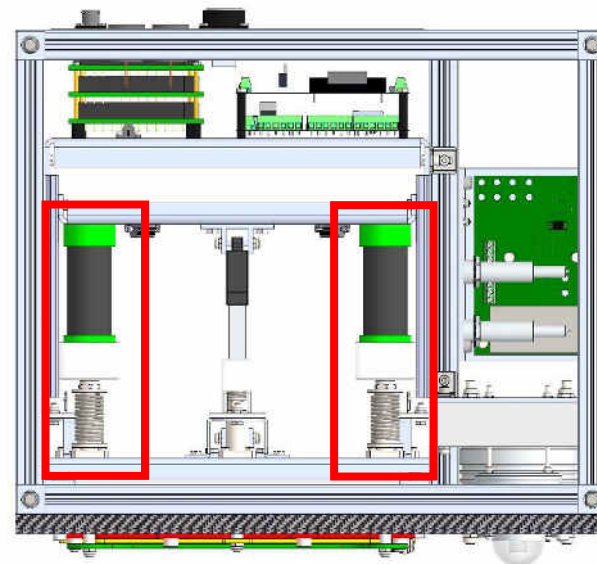
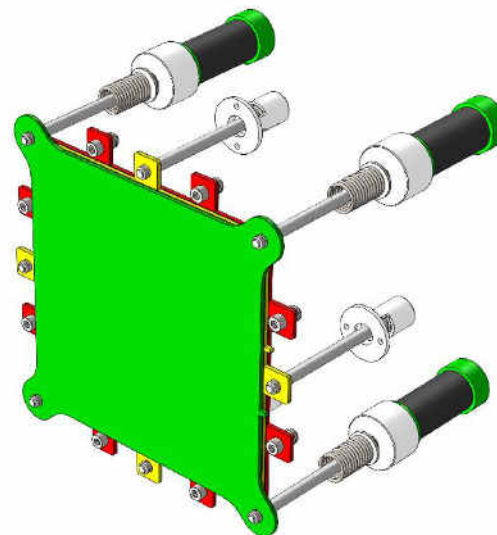
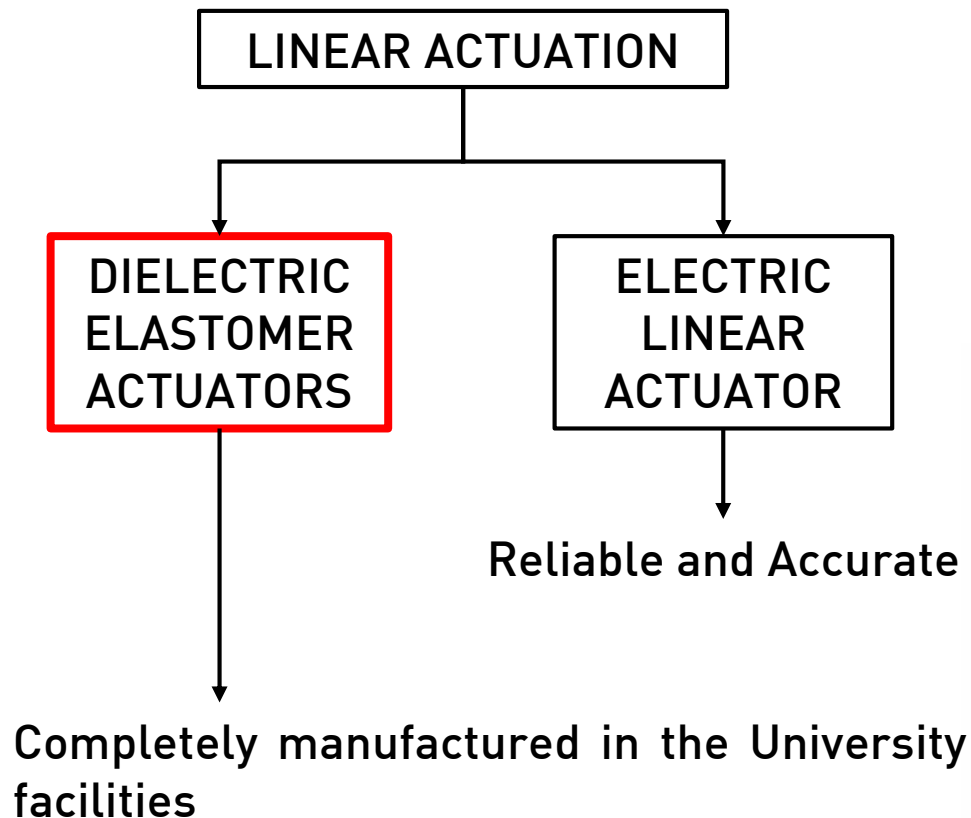
- POLARIS Setup flew into stratosphere on October 2014
- Aluminium Radiator Prototype
- Two Independent Actuation Systems for the radiator movement
- Set of sensors for full radiator performance evaluation



# POLARIS EXPERIMENT

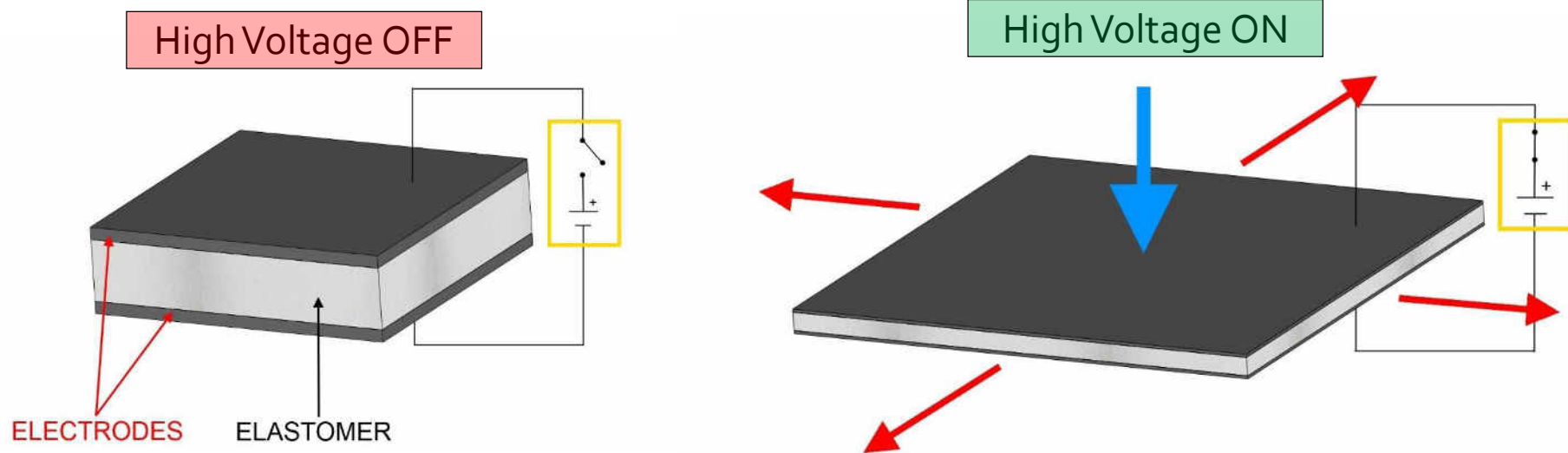


# POLARIS EXPERIMENT

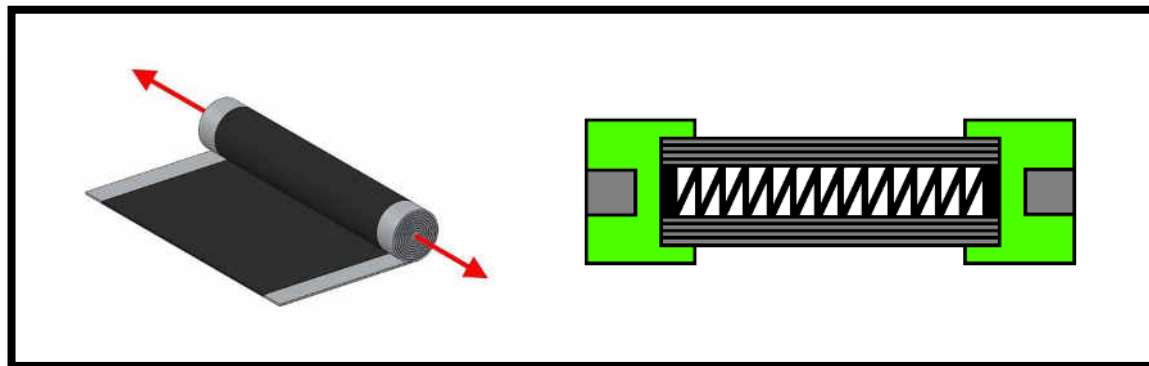


# POLARIS EXPERIMENT

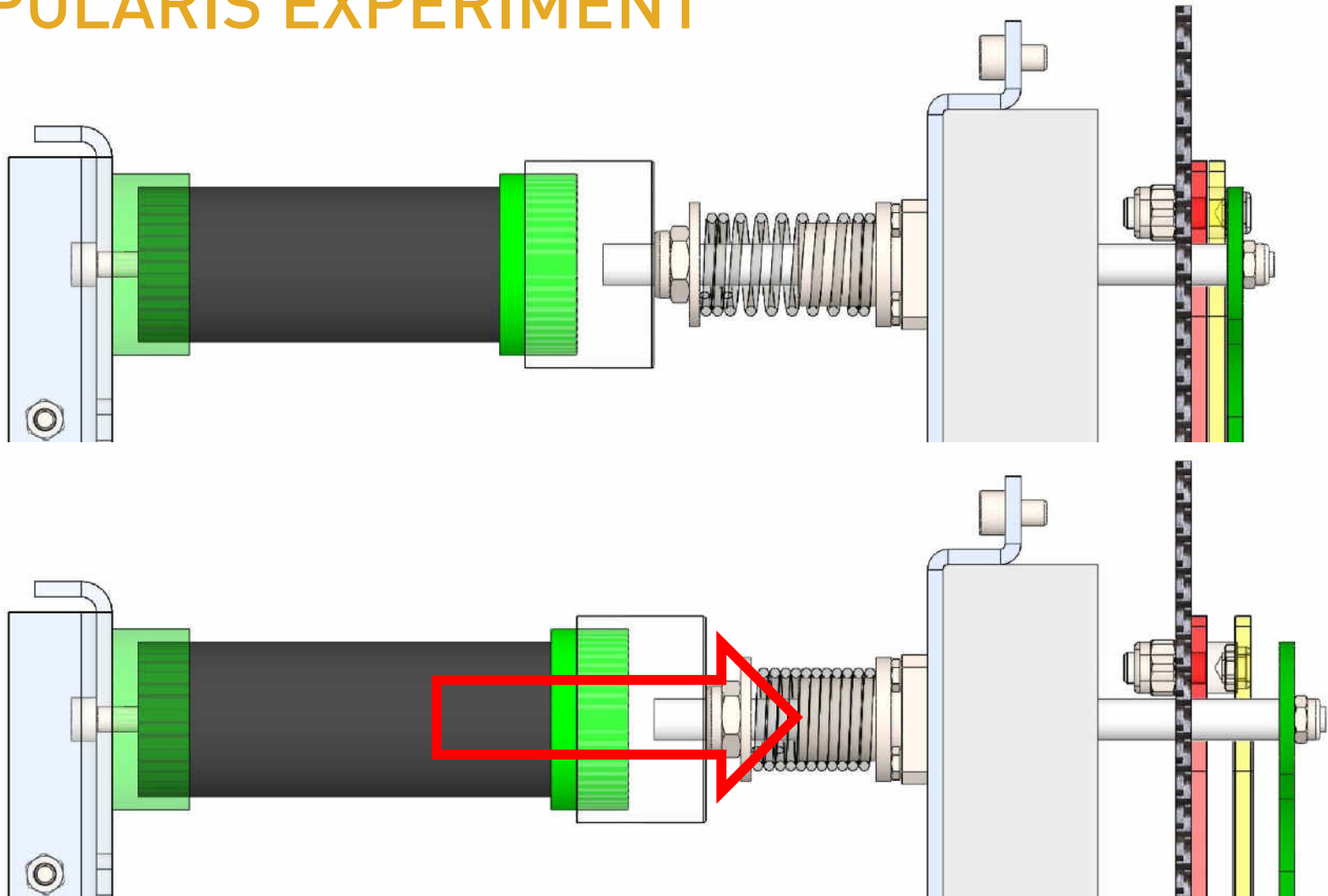
Basically they are COMPLIANT CAPACITORS



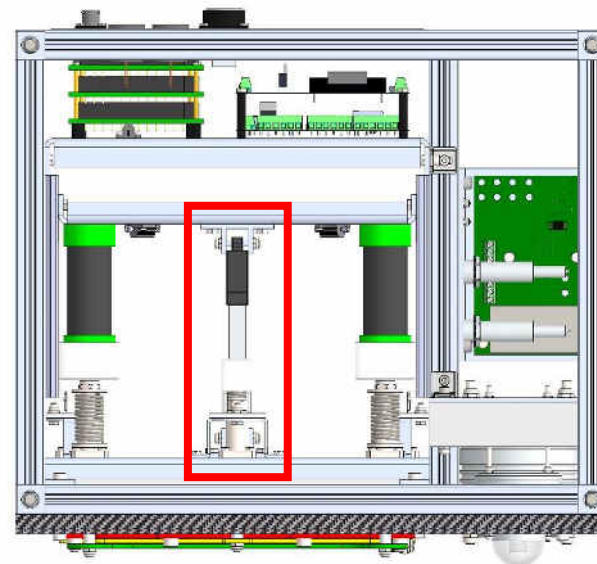
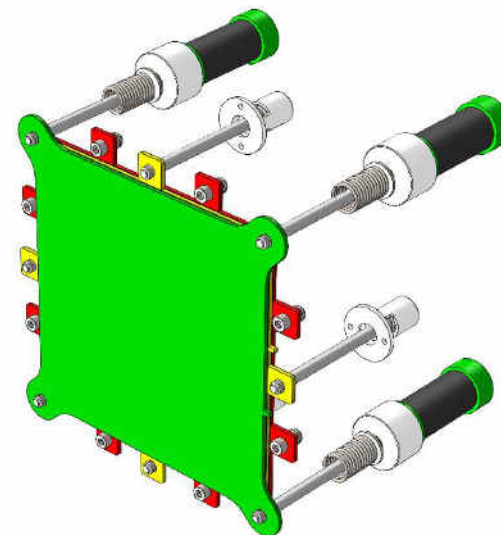
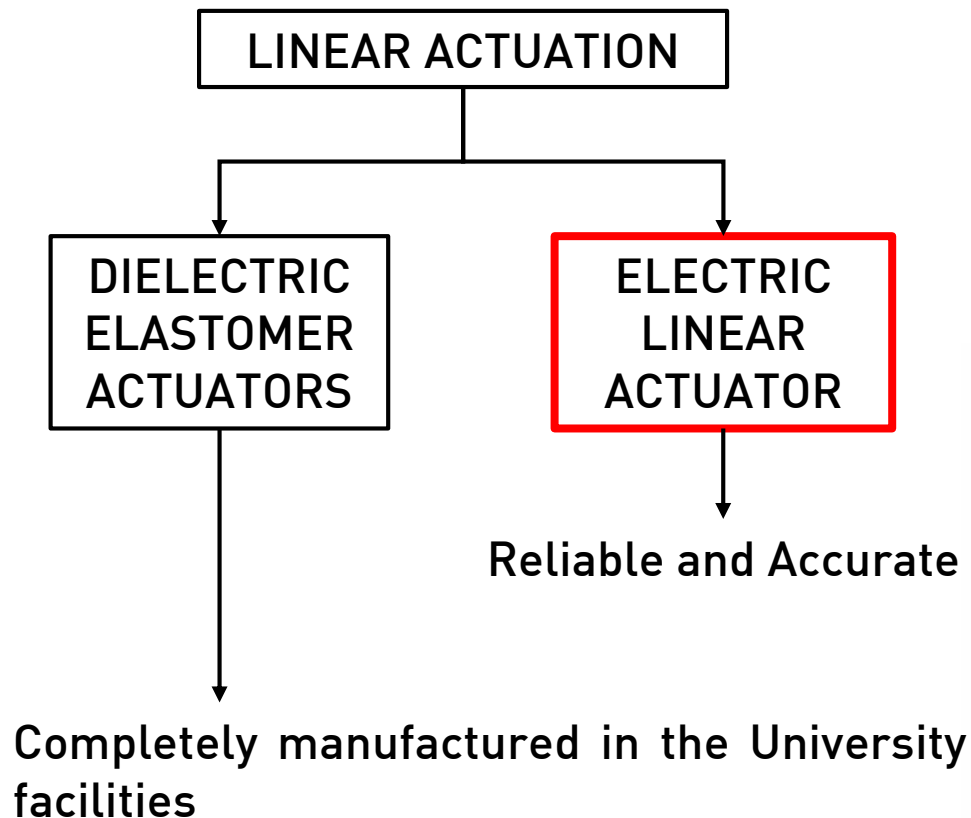
Wrapping it around a compressed coil spring is possible to obtain a **LINEAR ACTUATOR**



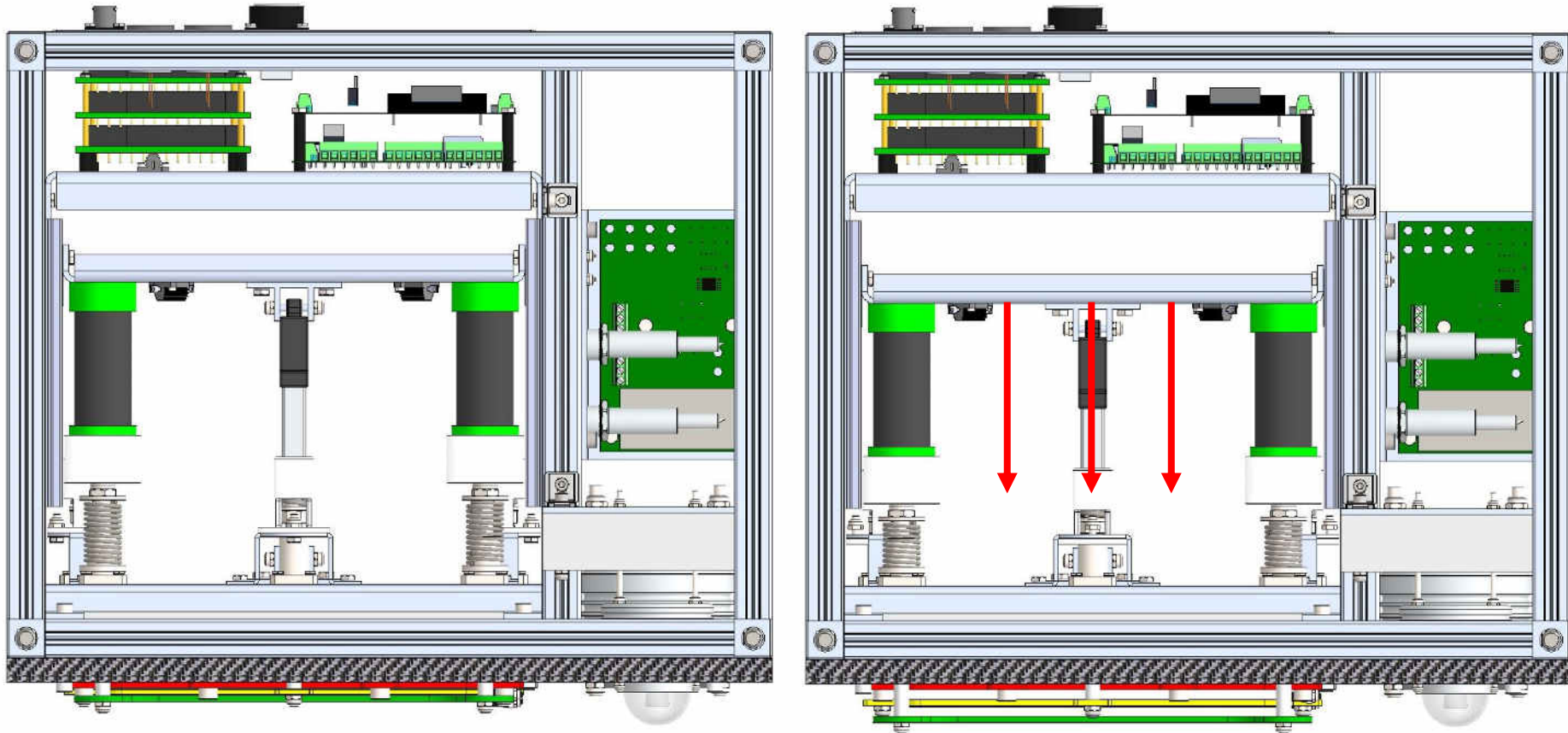
## POLARIS EXPERIMENT



# POLARIS EXPERIMENT



# POLARIS EXPERIMENT

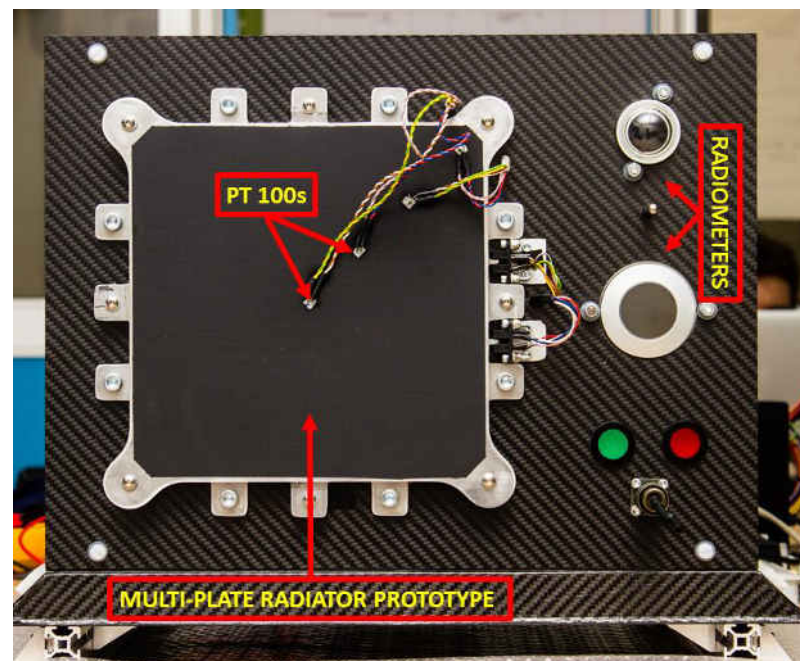




# POLARIS EXPERIMENT

A complete set of sensors was included in the setup, monitoring both radiator temperatures and environment boundary conditions

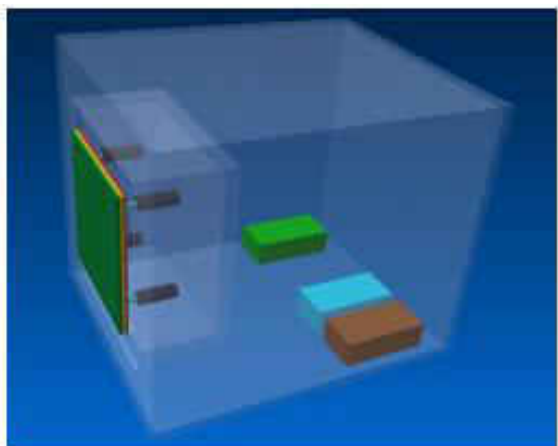
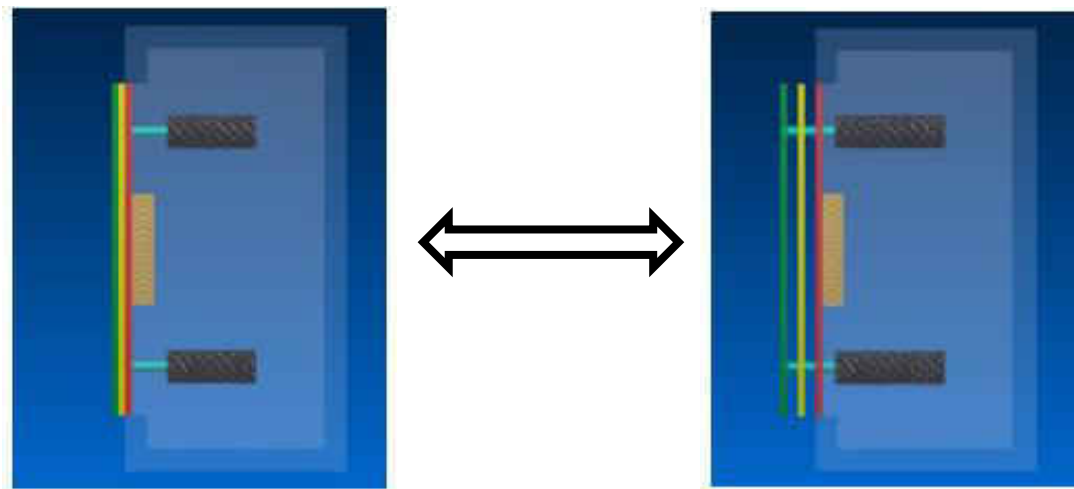
SPOT	MEASURE
Dummy Payload	Temperature Voltage Amperage
Inner Radiator Plate	Temperature and its spatial distribution
Outer Radiator Plate	Temperature and its spatial distribution
Plates Supports	Plates position
Environment Boundary Condition	Pressure Incoming Visible Radiation Incoming IR Radiation Temperature



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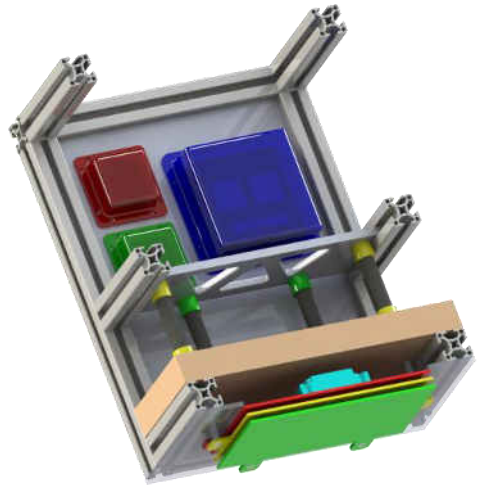
## DURING RX/BX (PROPOSAL)



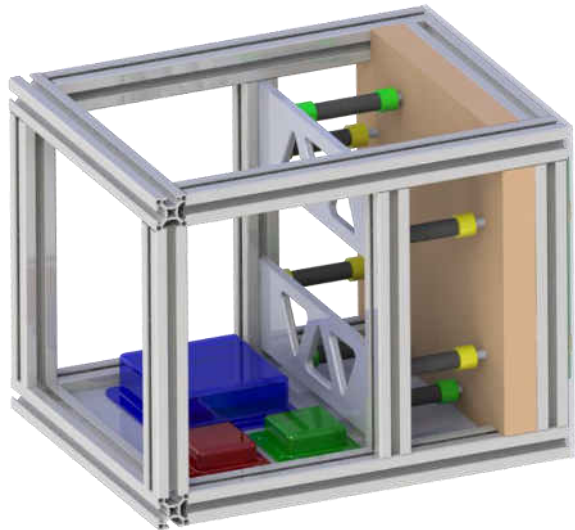
- ✓ Mission Concept
- ✓ Mission Objectives

Experiment MASS: 4 Kg  
Experiment DIMENSIONS: 30 X 30 X 30 cm

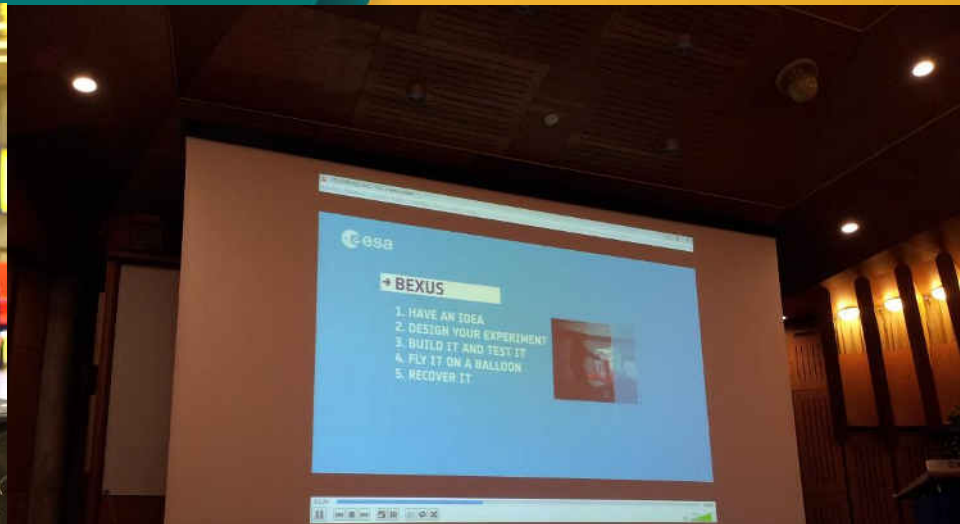
## DURING RX/BX (SW)



- ✓ Preliminary WBS
- ✓ 8 DE Actuators configuration
- ✓ Experiment divided in 3 main sections



Experiment MASS: 5 Kg  
Experiment DIMENSIONS: 40 X 30 X 30 cm



## DURING RX/BX (PDR)

- ✓ First version of the S.E.D.
- ✓ Definition of the WBS
- ✓ Definition of the Experiment REQUIREMENTS

**2.3 Design Requirements**

The experiment shall

Req.D.1

2.1 Functional Requirements

Req.F.1

Req.F.2

Req.P.3.5 measure the pressure outside the gondola with accuracy of 5%

Req.P.3.6 sample the pressure outside the gondola with a frequency of 1 Hz

Req.P.3.7 measure the incoming power radiation within the range 0 to 2000 W/m<sup>2</sup>

Req.P.3.8 measure the incoming power radiation with accuracy of 1 W/m<sup>2</sup>

Req.P.3.9 sample the incoming power radiation with a frequency of 1 Hz

Req.P.3.10 measure the incoming power radiation within the wavelength ranges from 285 to 2000 nm and from 4 to 40 μm

Response to F.4:

Req.P.4.1 measure the inner plate temperature from 8 different points on its surface

Req.P.4.2 measure the temperature of the inner plate with an accuracy of ± 0.1 K

Req.P.4.3 measure the temperature of the inner plate within the range 173 to 293 K

Req.P.4.4 sample the temperature of the inner plate with a frequency of 1 Hz

Req.P.4.5 measure the temperature of the external plate from 6 different points on its surface (3 spotted along a half diagonal, 3 along a half side)

Req.P.4.6 measure the temperature of the external plate with an accuracy of ± 0.1 K

Req.P.4.7 measure the temperature of the external plate within the range 173 to 293 K

Req.P.4.8 sample the temperature of the external plate with a frequency of 1 Hz

Response to F.5:

Req.P.5.1 store a total data volume larger than 100 MByte

Response to F.6:

Req.P.6.1 transmit data with a data rate up to 50 kbit/s

**2.4 Operational Requirements**

The experiment shall

Req.O.1 be easily be able to a

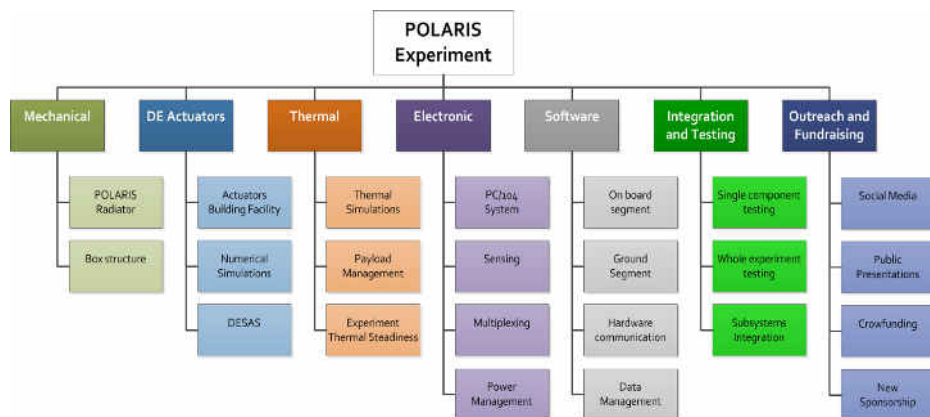
Req.O.2 compute sens.

Req.O.3 be able to compu

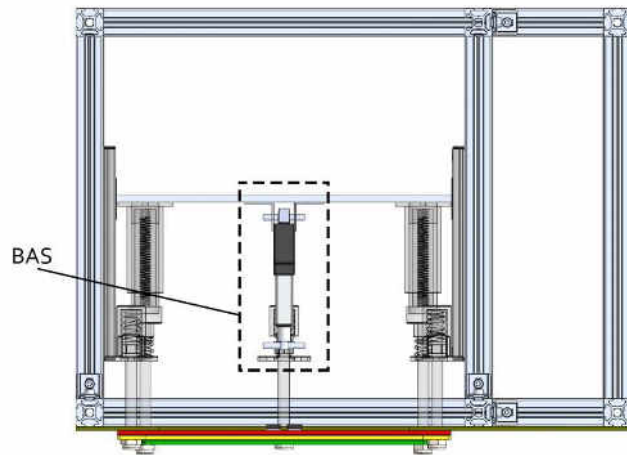
Req.O.4 temperature for Tel

Req.O.5 be able to follow th

autonomously

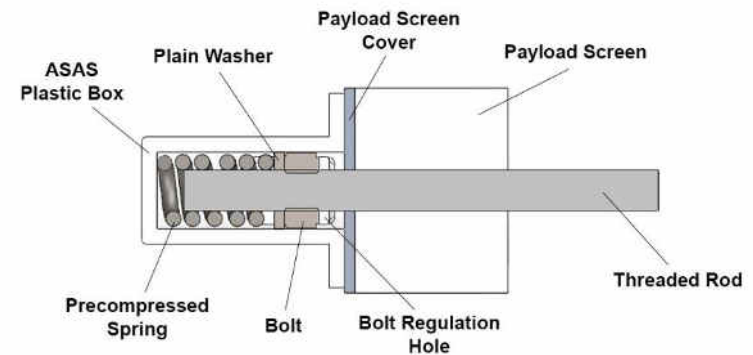
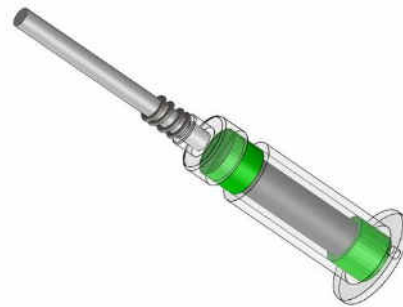
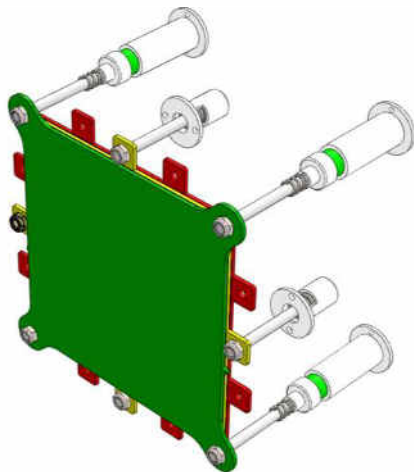


## DURING RX/BX (PDR)



- ✓ Design of the Radiator Assembly
- ✓ Preliminary Design of the Actuation System
- ✓ Preliminary Design of the Backup A.S.

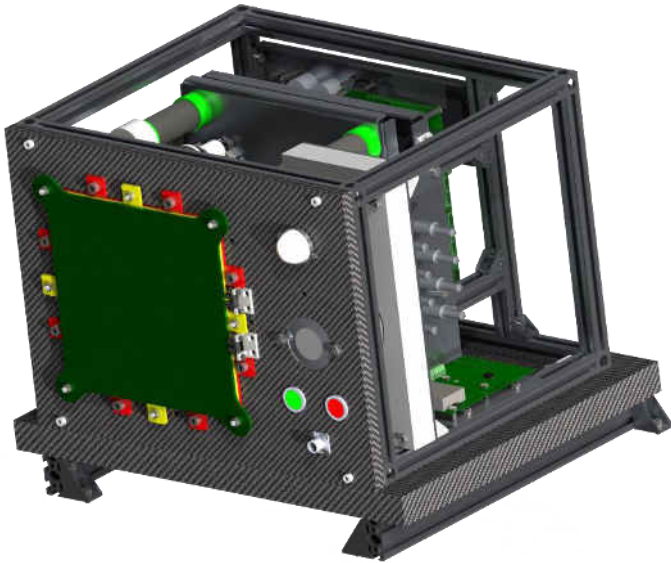
Experiment MASS: 12 Kg  
 Experiment DIMENSIONS: 46 X 34 X 36 cm







## DURING RX/BX (CDR)



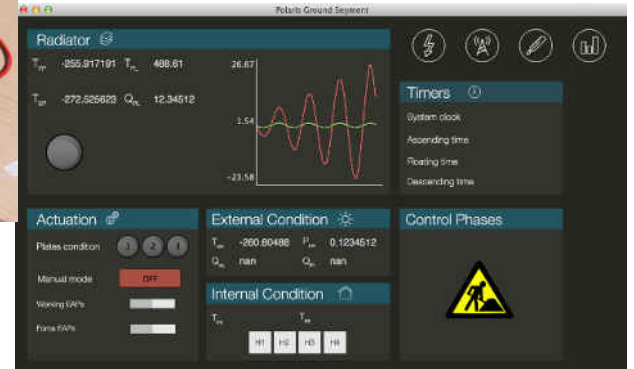
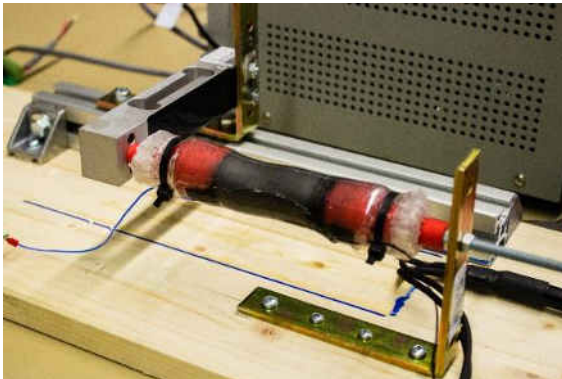
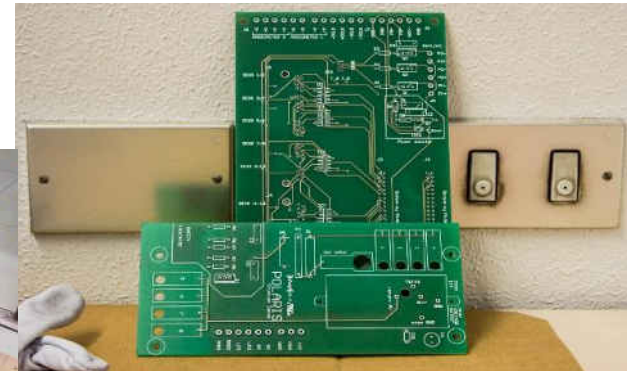
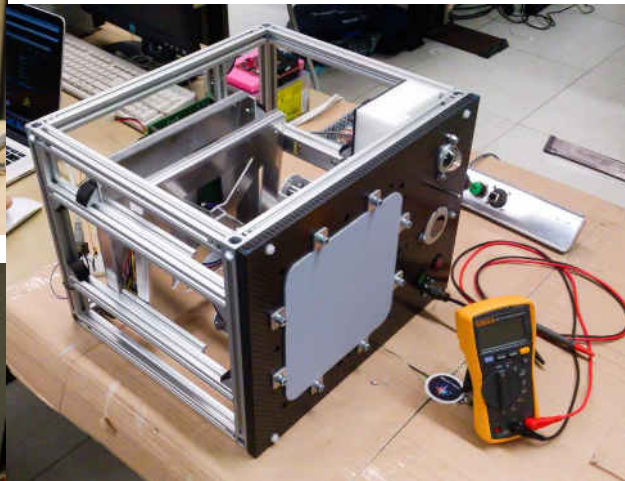
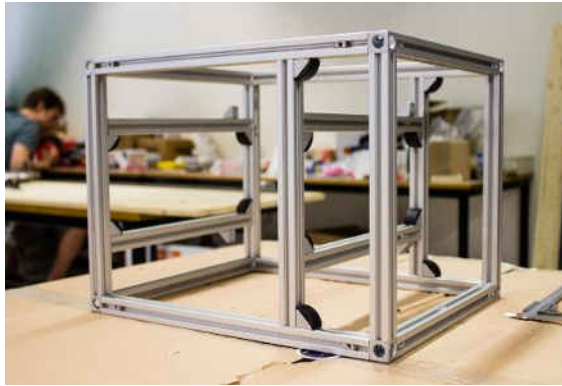
POLARIS Experiment DESIGN completed

Experiment MASS: 15 Kg

Experiment DIMENSIONS: 46 X 38 X 40 cm

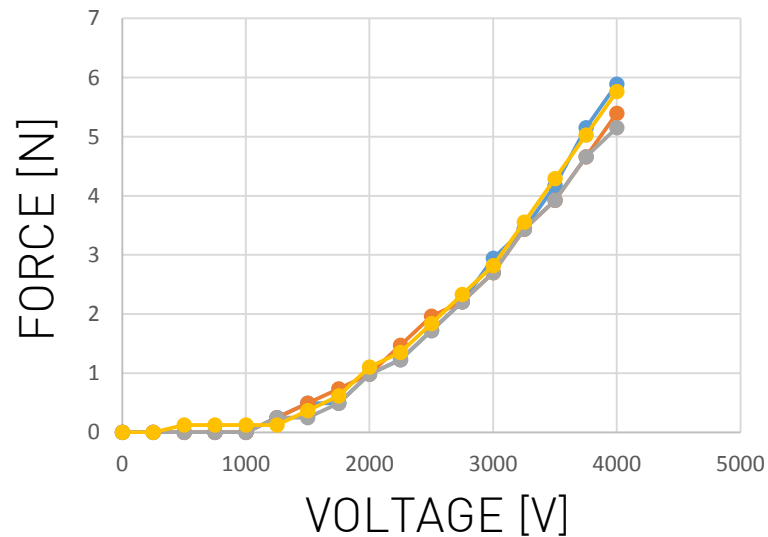
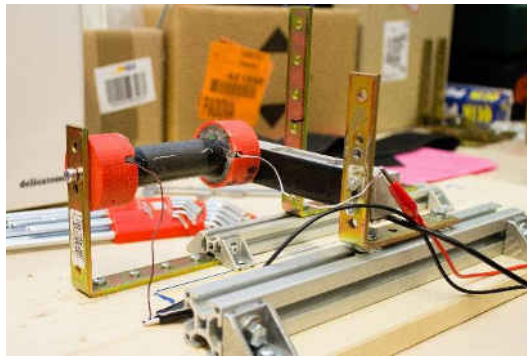


# DURING RX/BX (IPR)

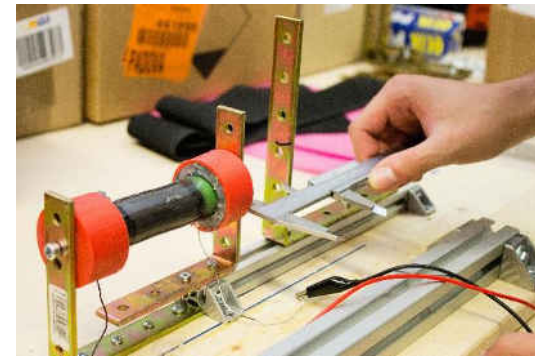


# DURING RX/BX (IPR)

## FORCE

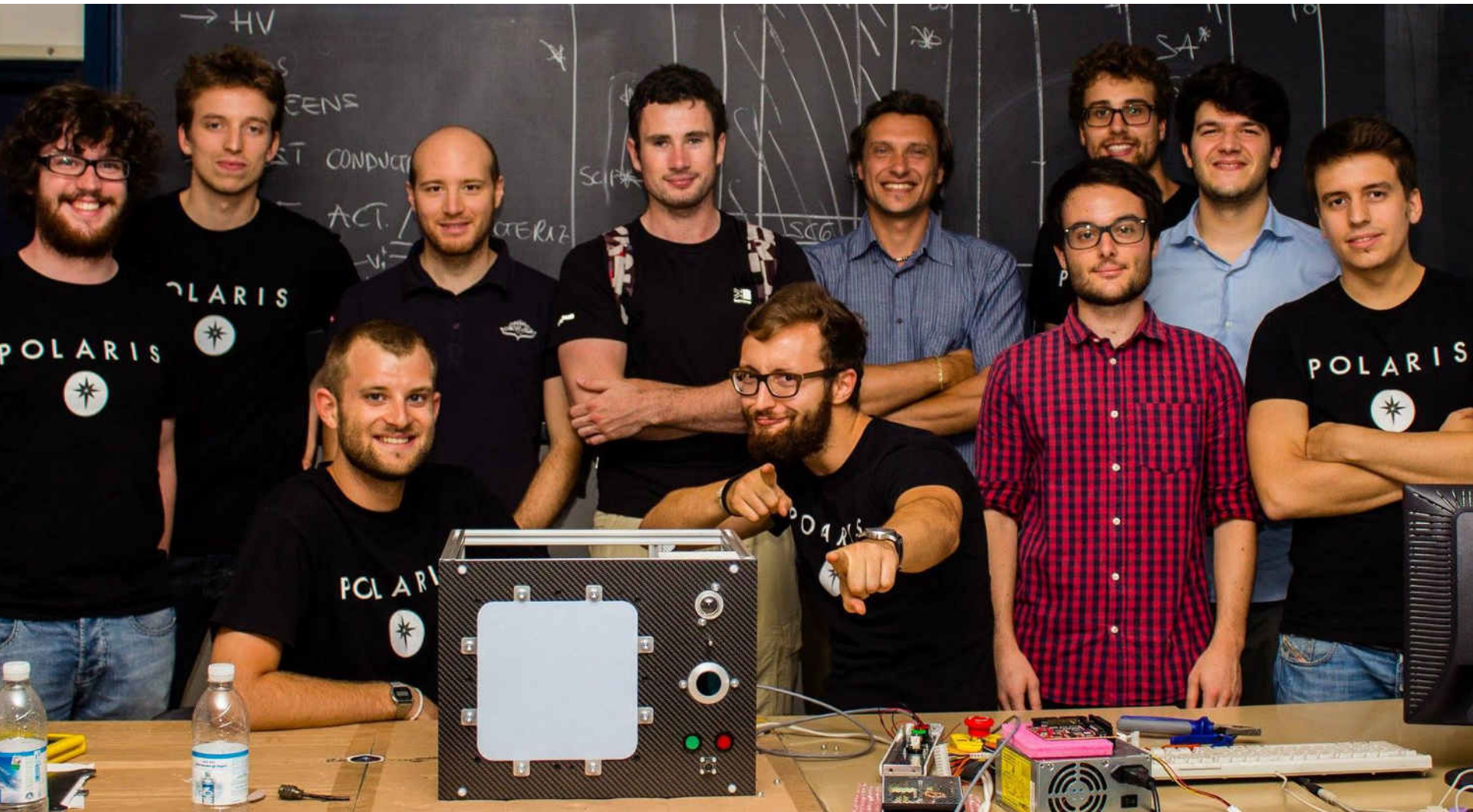


## DEFORMATION

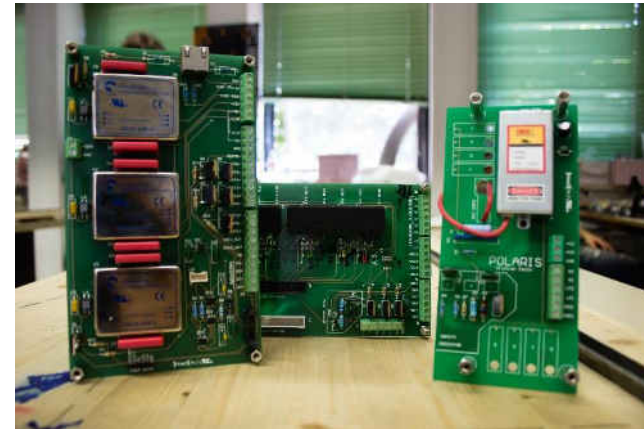


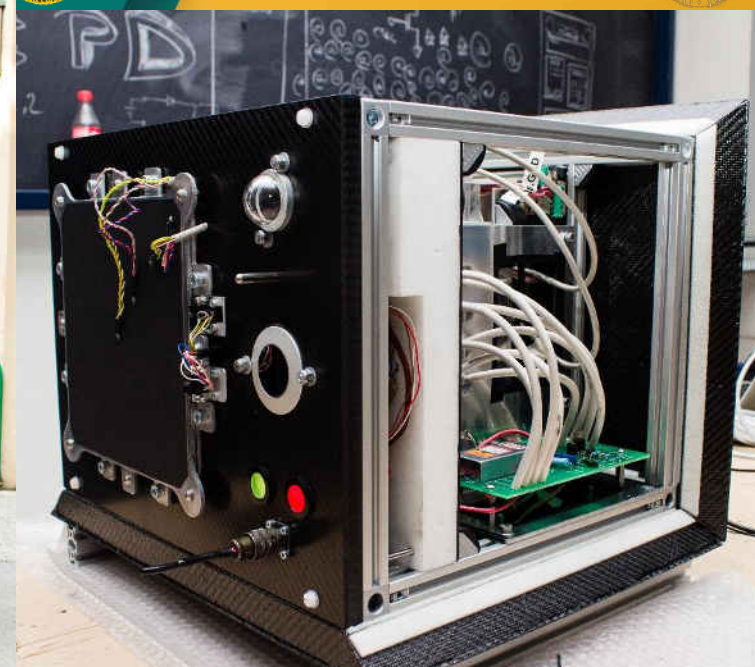
All the tested actuators show maximum **DEFORMATIONS OF 5 MM**, but limited performance for the mechanism requirements

## DURING RX/BX (IPR)



## DURING RX/BX (EAR)





## DURING RX/BX (LAUNCH CAMPAIGN)





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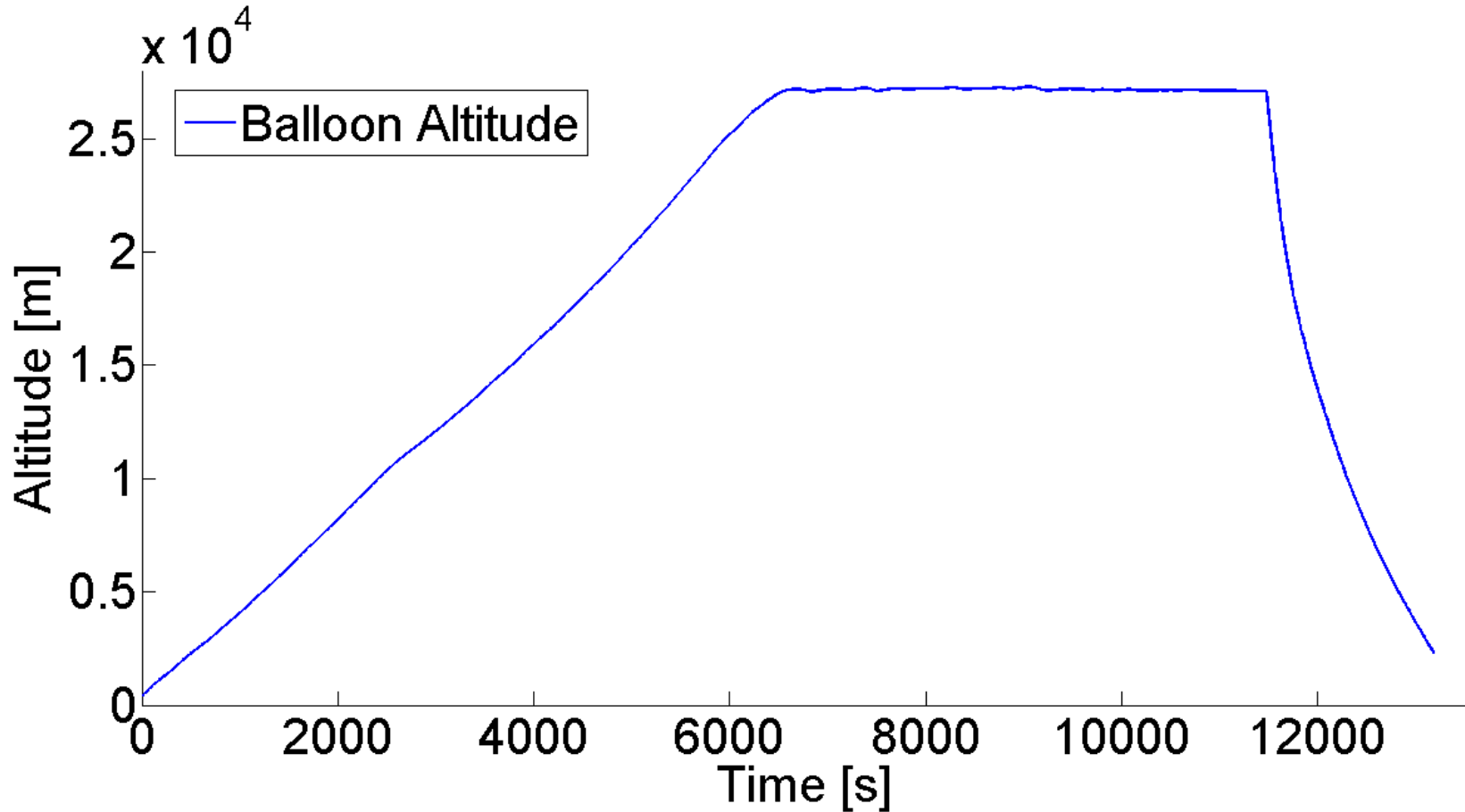
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# FLIGHT DATA

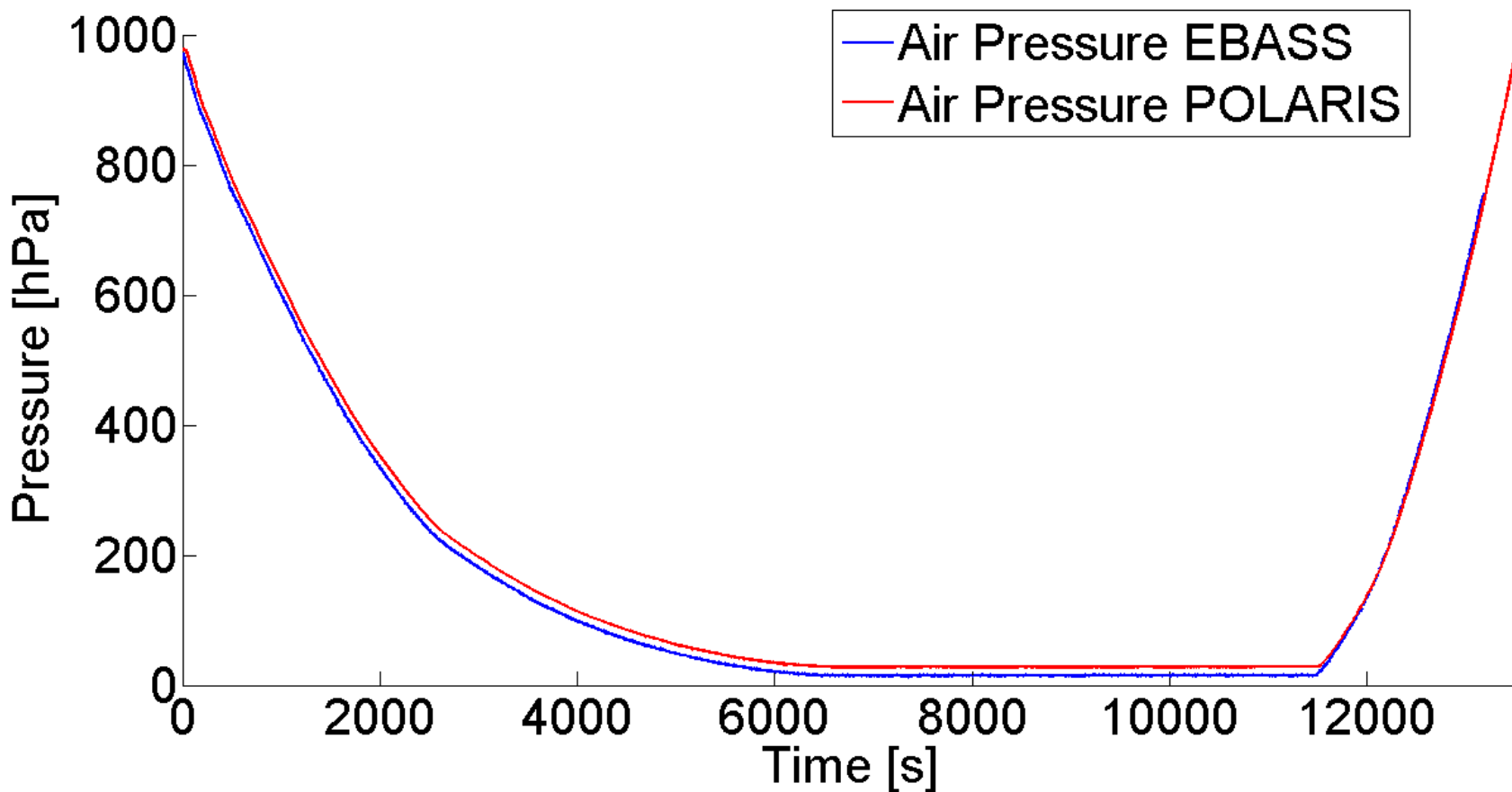
The setup flew into stratosphere on-board the balloon BEXUS18 from the ESRANGE Space Center (9 October 2014)



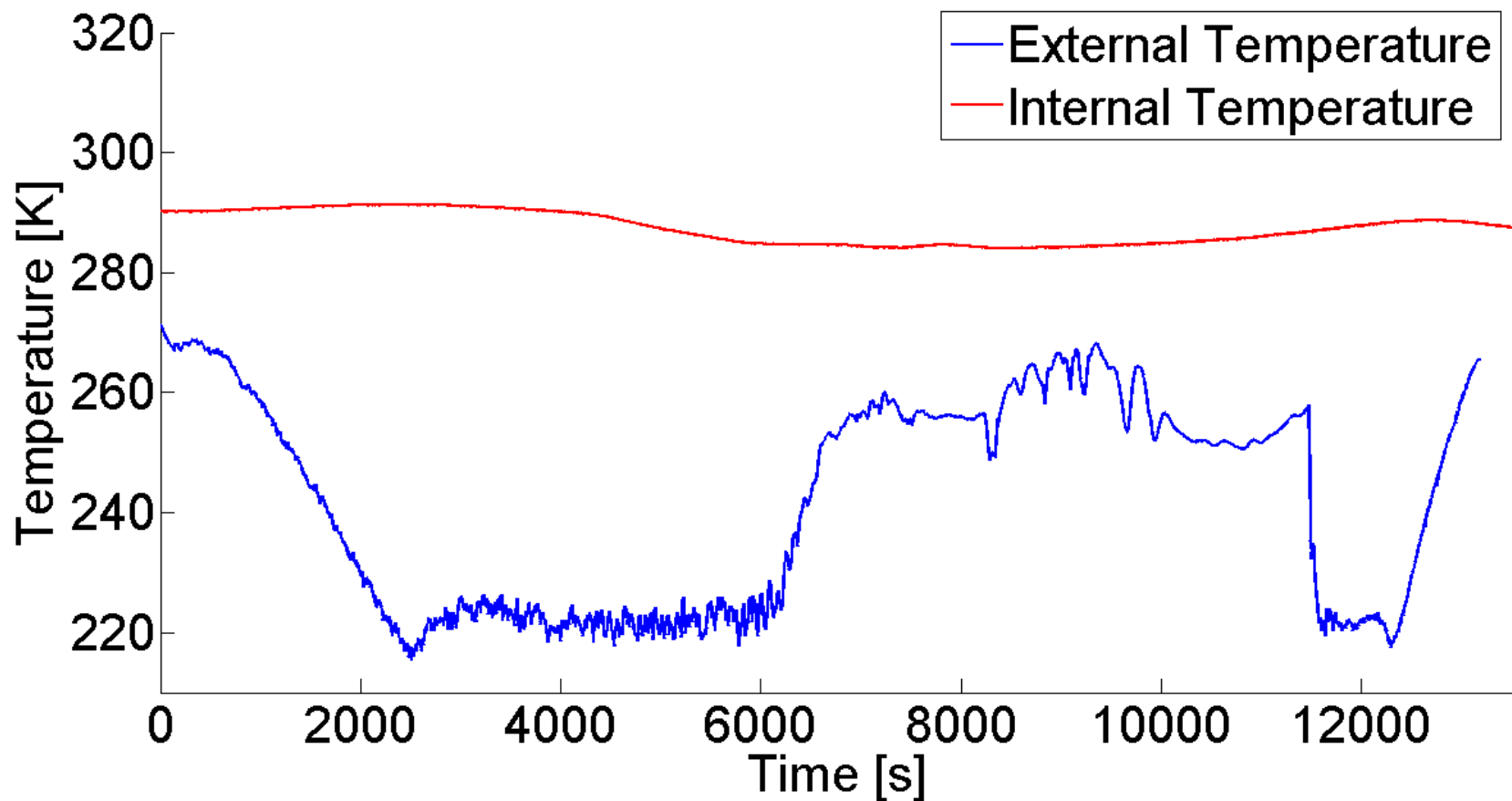
# FLIGHT DATA - ENVIRONMENT



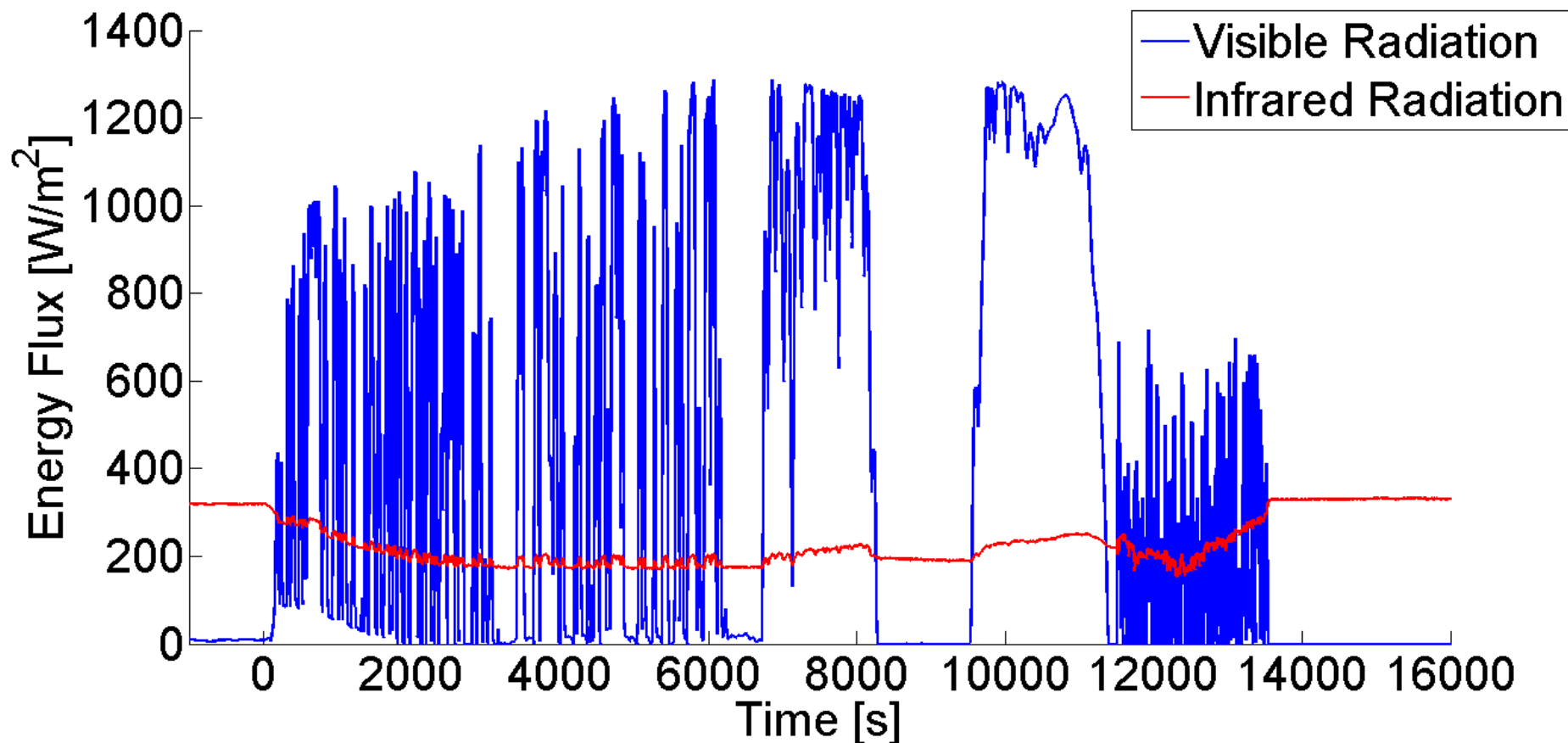
# FLIGHT DATA - ENVIRONMENT



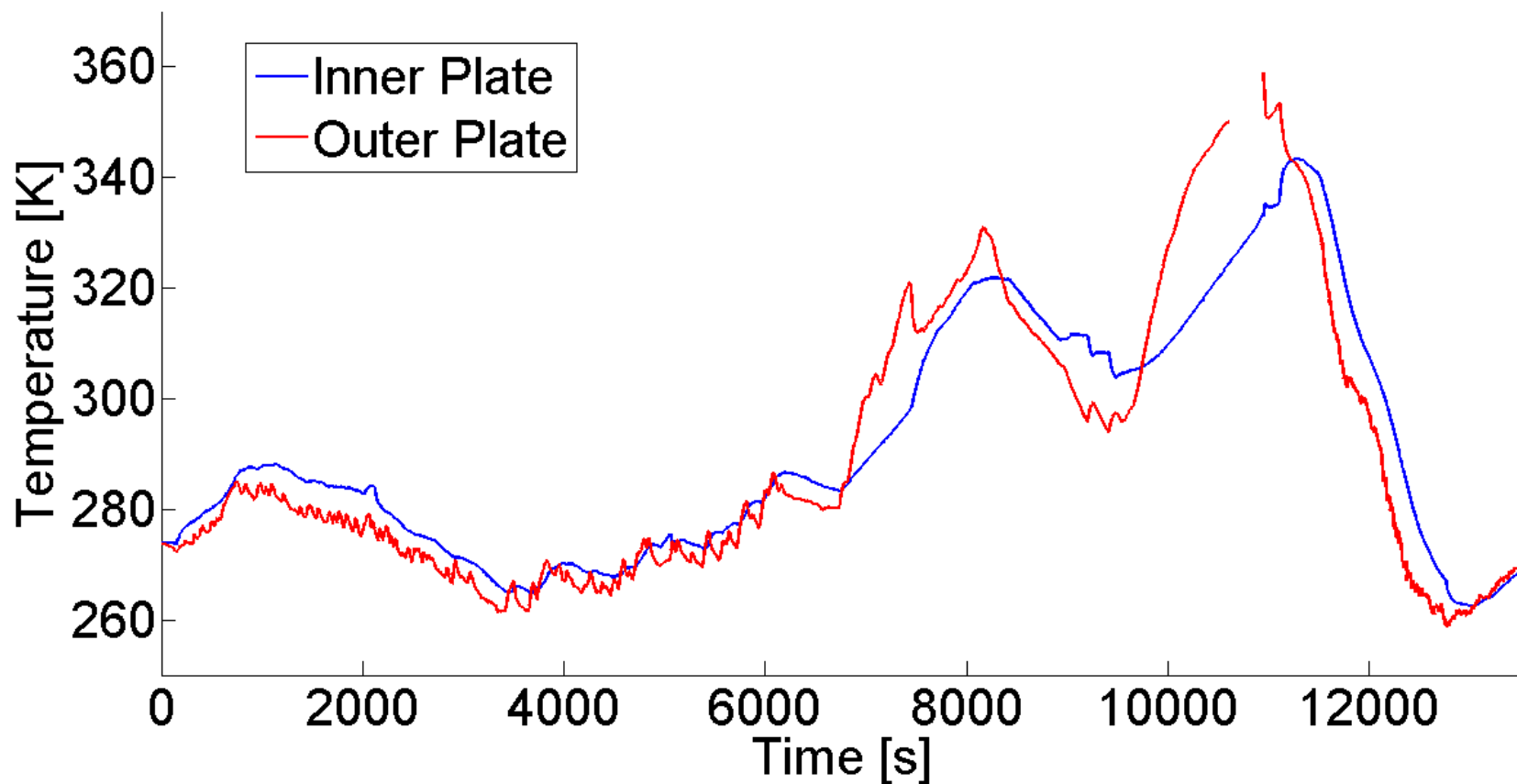
# FLIGHT DATA - ENVIRONMENT



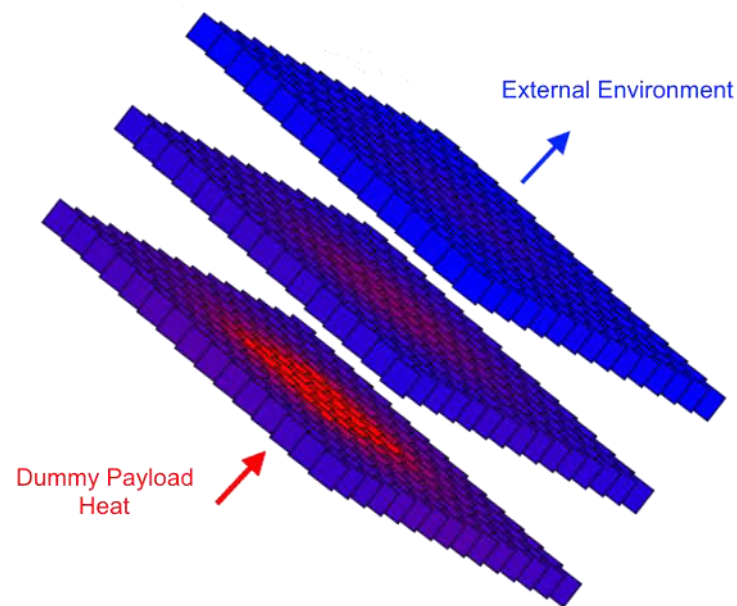
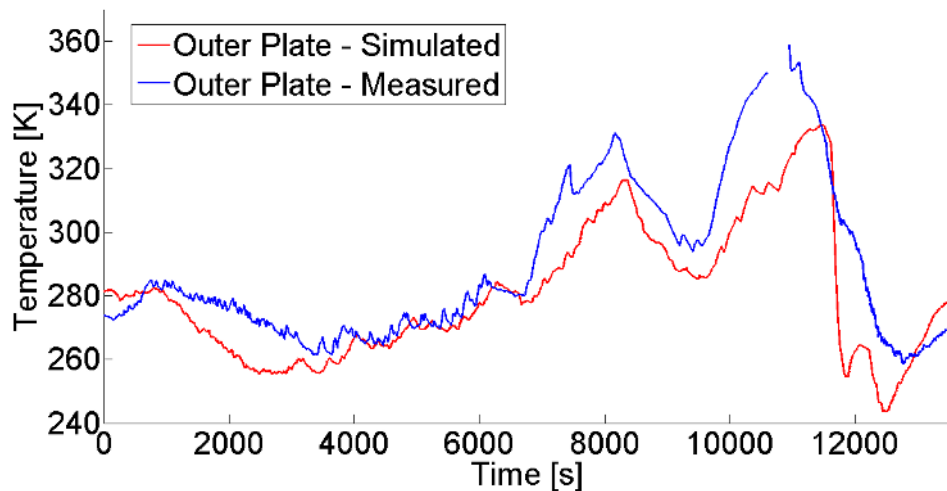
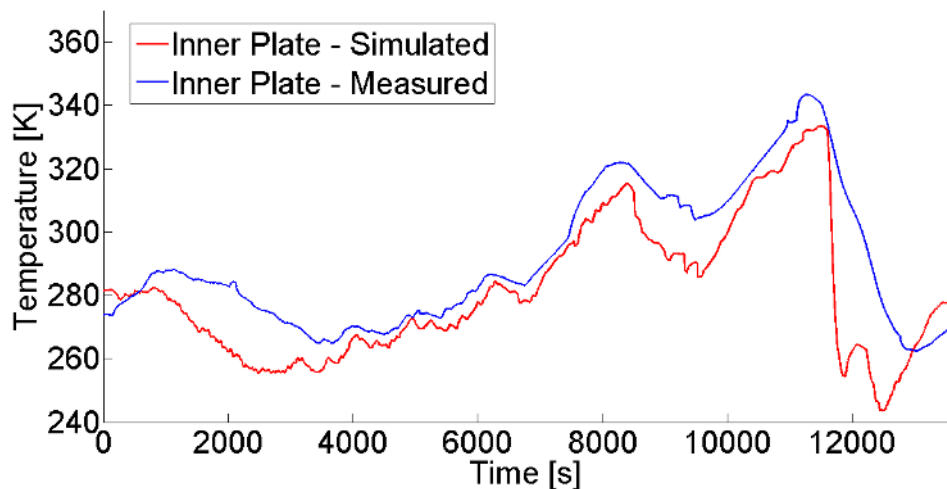
# FLIGHT DATA - ENVIRONMENT



# FLIGHT DATA - RADIATOR

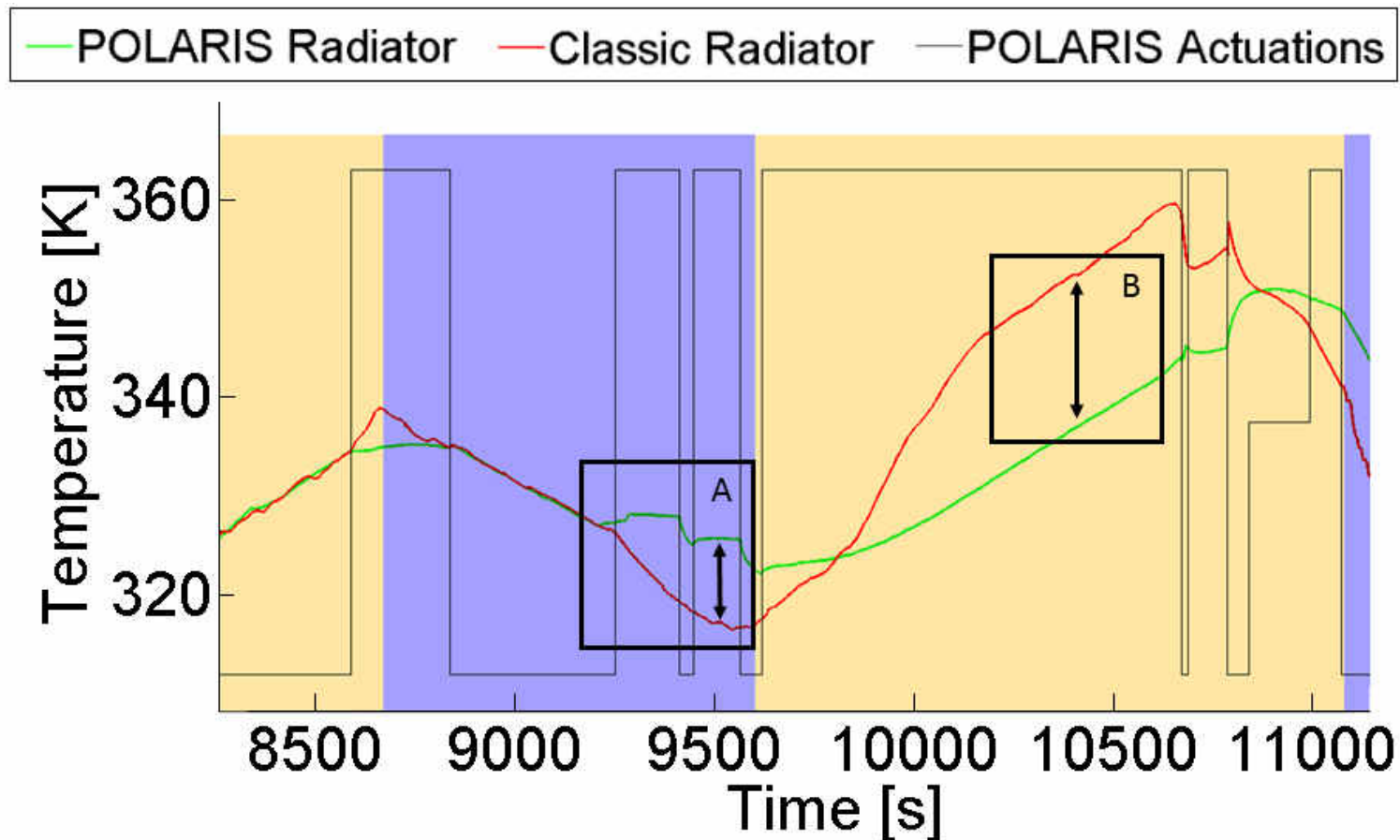


# FLIGHT OUTCOMES





# FLIGHT OUTCOMES



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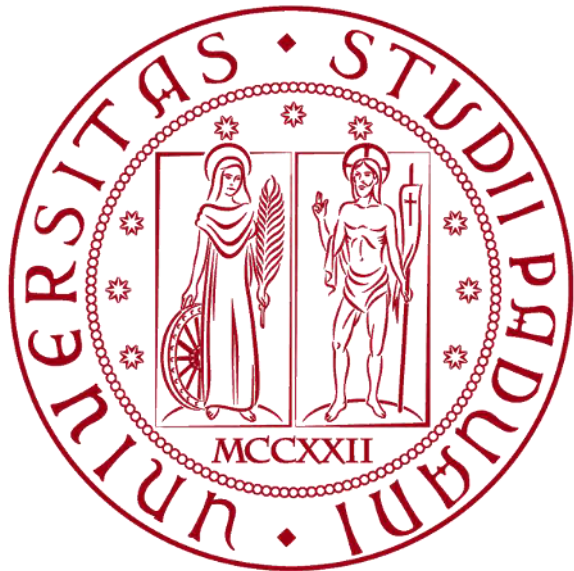
# LESSONS LEARNED & CONCLUSIONS

- Despite of an unexpected flight condition, it was possible to prove functionality and potentialities of this concept of radiator
- Experiment bus & software were perfectly functional during the whole flight
- Dielectric Elastomers Actuators manufacturing process was developed, even though the DEs performance were not enough to fulfill the experiment requirements
- New model was developed and post flight work was focused on vacuum test to characterize the radiator performance in a more stable environment

# LESSONS LEARNED & CONCLUSIONS

- How to deal with the shipment and procurement of the experiment components: the experiment manufacturing (and thus all the following phases) should be carefully planned
- How to deal with the development of new technologies without extensive literature references (i.e. Dielectric Elastomers): the technology development should not be absolutely underestimated
- KEEP IT SIMPLE
- Define your objectives, state the requirements you need in order to fulfill them and then base your design choice on them. Once you froze your design (CDR) don't change desing!

# ACKNOWLEDGEMENTS



## Special thanks to:

- Prof. Alessandro Francesconi
- Dr. Francesco Branz
- Dr. Lorenzo Olivieri
- Dr. Francesco Sansone





THANK YOU

Questions?

# CONTACTS

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Team Site: <http://polaris.armadillo.space/>

Facebook Page: [https://www.facebook.com/polarisexperiment?hc\\_location=ufi](https://www.facebook.com/polarisexperiment?hc_location=ufi)