



UNIVERSITÀ
DEGLI STUDI
DI PADOVA

Analysis and modelling of spacecraft fragmentation in consequence of in-orbit collisions

Stefano Lopresti - 39th Cycle

Supervisor: Prof. Alessandro Francesconi

1° year acceptance - 19/10/2023



Introduction



Project description



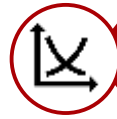
State of art



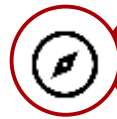
Work plan



Gantt Chart and WBS



Main expected results



Conclusions



Introduction

- Increasing number of objects resident in Earth orbits
 - Growing **risk of impacts** and collisions
 - Potential **cascade effect** and reduced capability to access **polluted orbits**
- Need to understand the **physical processes involved in break-ups**
 - Consequences for the space environment
 - Potential risks and mitigation techniques
- **Limited information** available on space fragmentation
 - Ground observations limited to **detectable objects (> 5 cm)**
 - Need to perform **numerical simulations** or **ground tests**





Project description

The project is divided into 3 main research lines

- Analysis of the fragmentation of **simple components**, aimed at characterizing the response of **new-generation materials** such as additive manufacturing components, composite materials, and polymers to hypervelocity impacts.
- Analysis of the fragmentation of **complex systems**, conducting **representative tests** of in-orbit fragmentation events resulting from explosions or accidental or deliberate kinetic collisions, gathering data from tests and literature, and improving the CSTS code.
- **Development and testing** of a high reliability, high resolution, and low-cost **sensor** for the detection of millimeter-sized and submillimeter-sized debris in orbit.



State of the art

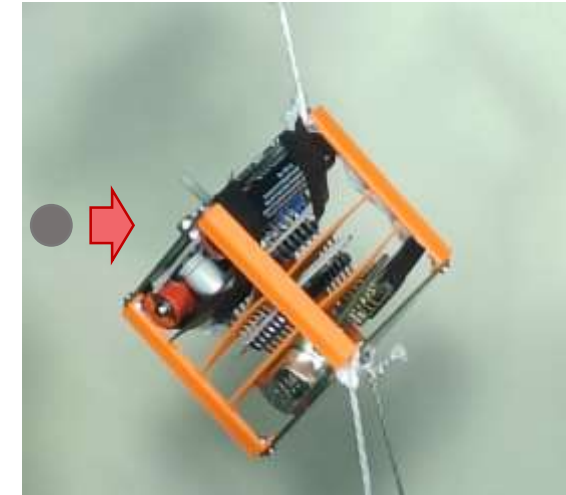
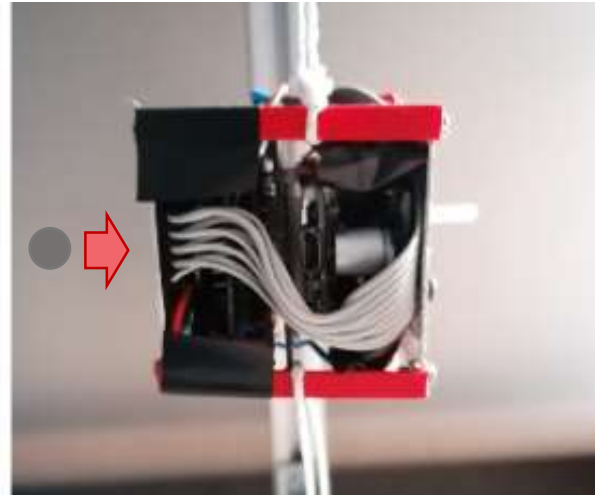
- **Two stages Light Gas Gun (LGG):** 100 mg projectiles up to 5.5 km/s
 - For aluminium and CFRP internal breakup models have been developed
 - Good correlation between calculated masses and actual fragment masses





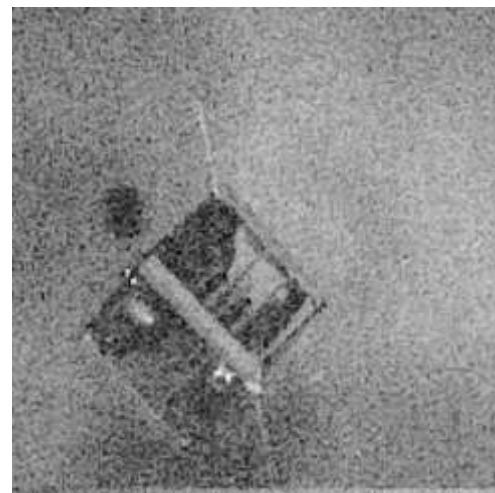
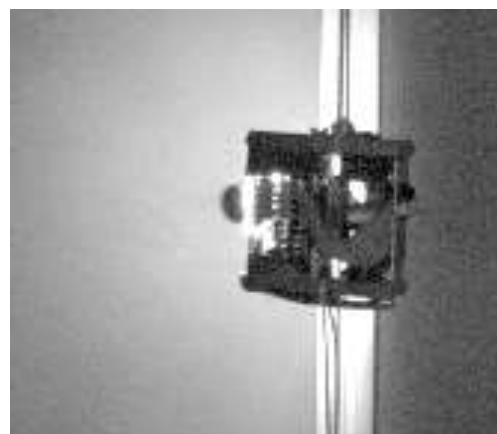
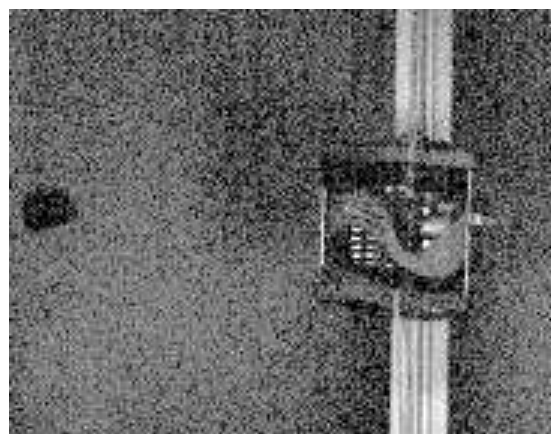
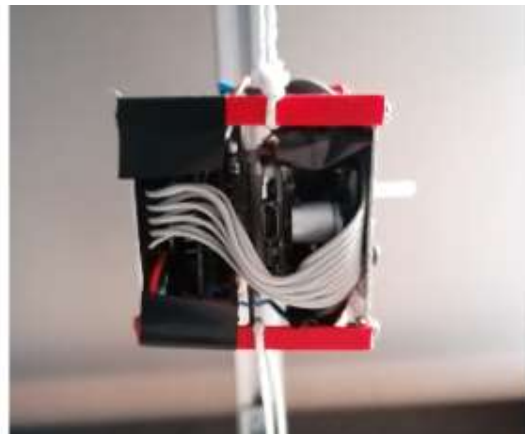
State of the art

- Impact tests recently performed on a picosatellite mockup.
- Projectile material and velocity kept constant to study the **correlation** between **generated debris population** and **impact geometry**.





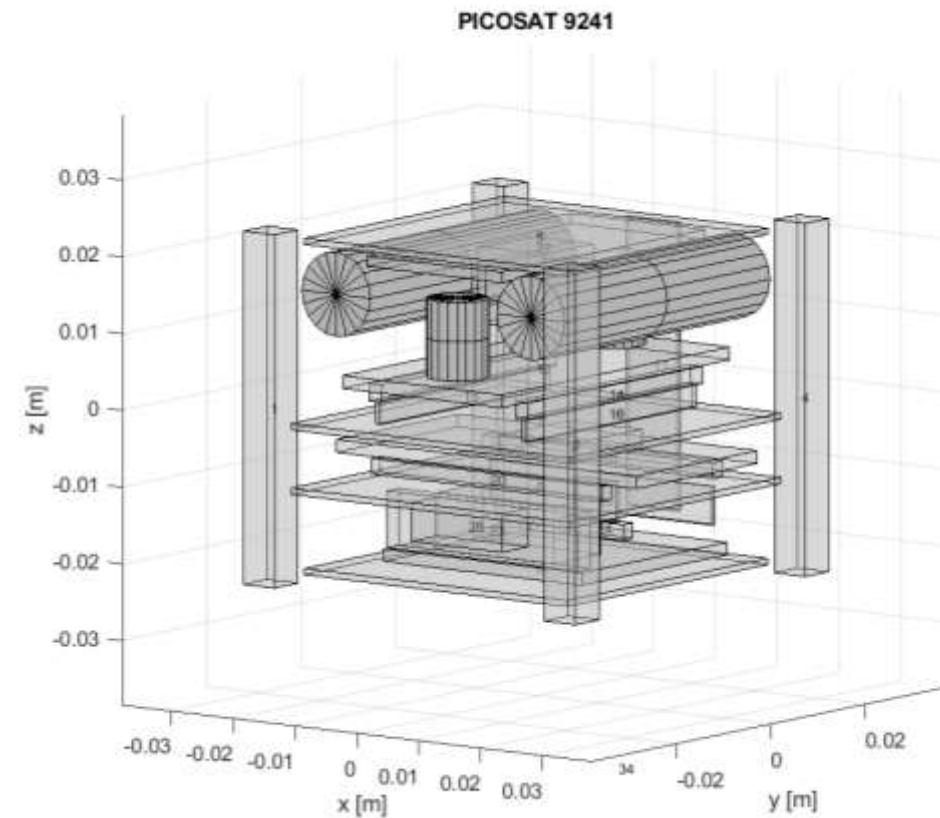
State of the art





State of the art

- Codes developed **semi-empirically** are used to perform simulations of fragmentation events in orbit
- Impacts between **very different or very large geometries** are particularly challenging
- Function that **correlates crater shape and size with impact geometry** is not currently implemented

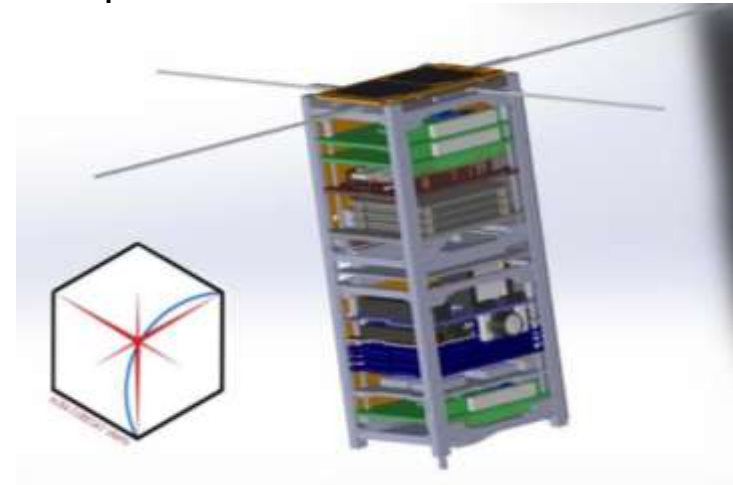




State of the art



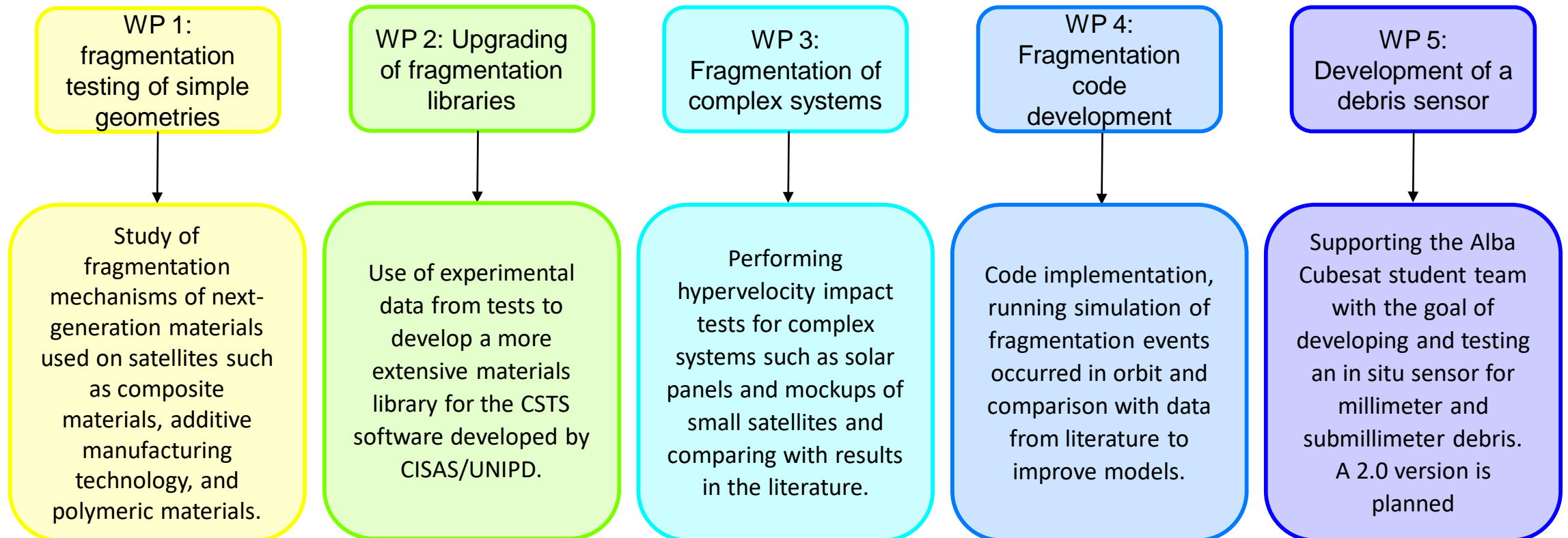
- Numerous **in-situ impact detectors** have been used or are under development with different baseline technologies
- In Padova, the **student team Alba Cubesat** has realized the **preliminary design** of an extremely lightweight and compact debris sensor





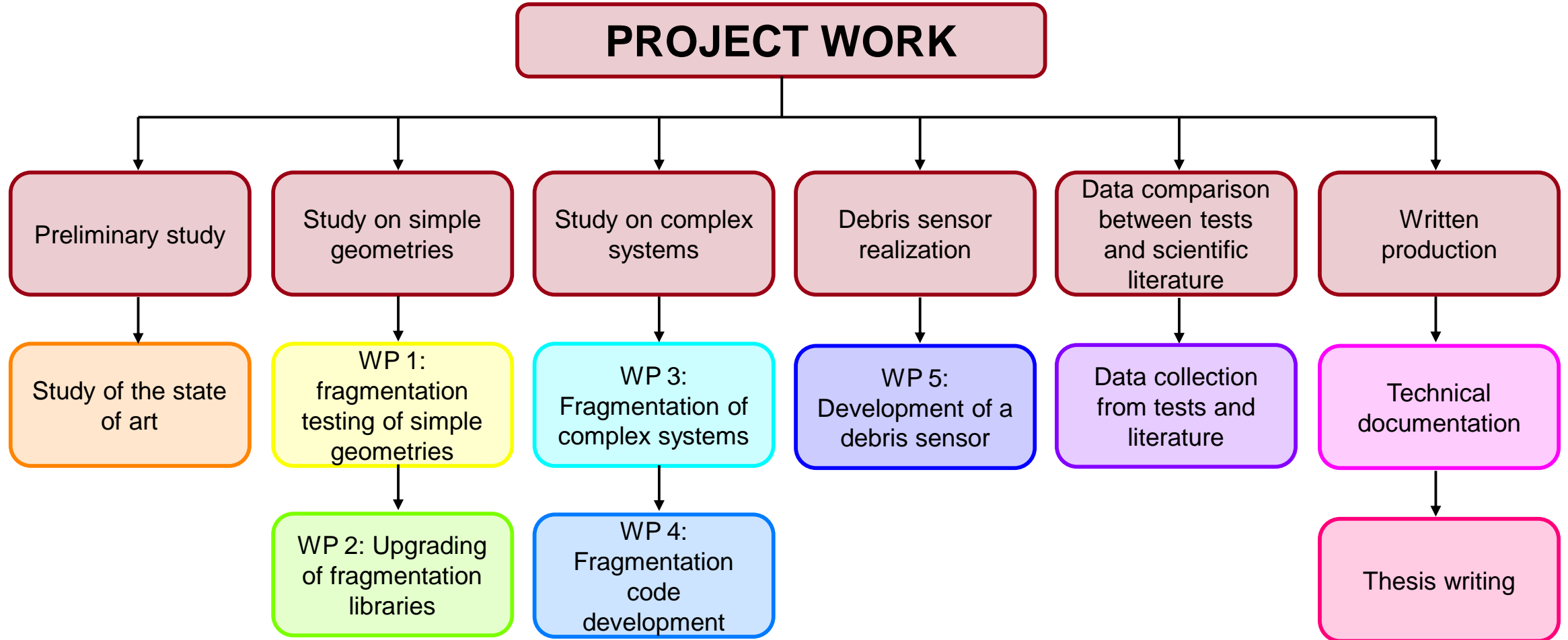
Work plan

- This project is divided into 5 mutually independent work packages



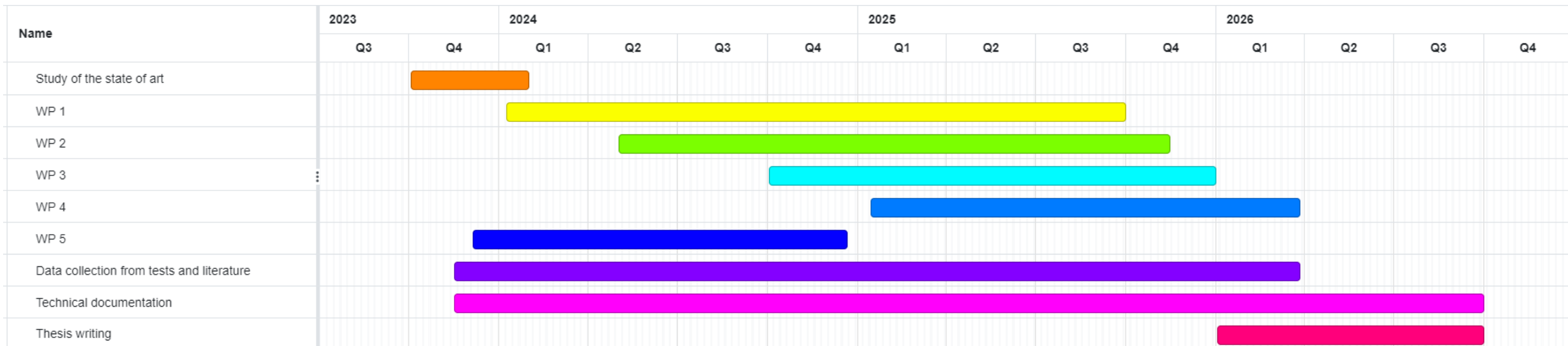


Gantt Chart e WBS





Gantt Chart e WBS





Main expected results

- Better understanding of the phenomenon of fragmentation under hypervelocity impact conditions.
- Upgrading CSTS software by creating simplified fragmentation databases for innovative materials and improving the capability to simulate complex fragmentations.
- Acquiring test data and compare them with those in the literature.
- Develop a debris sensor to support data collection to reduce the uncertainty of submillimeter debris distribution models.
- Improving the ability to collaborate within interdisciplinary projects with other researchers. For this reason, it is planned to stay 6/18 months abroad. There are already contacts with research centers abroad (in particular with the Malta College of Arts, Science & Technology, MCAST) and with company (Thiot Ingénierie).

Thanks for your attention



UNIVERSITÀ
DEGLI STUDI
DI PADOVA