### Advanced plasma sources for space applications

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### Outline

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  - Physical Assessment
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- 6 Source Realization, and Testing
  - Conclusions

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Plasma exhibits complex Electromagnetic (EM) wave phenomena. It can be exploited in a broad range of advanced application:



 $\label{eq:Plasma} \begin{array}{l} \mbox{Plasma exhibits complex Electromagnetic (EM) wave phenomena.} \\ \mbox{It can be exploited in a broad range of advanced application:} \end{array}$ 

#### Space Propulsion:

#### Plasma Thrusters



Space Communication:

Gaseous Plasma Antennas



#### Plasma propulsion systems

Use electric power to ionize the propellant and impart kinetic energy to the plasma.

#### Critical issues:

- Limited lifetime
- Need for an external cathode
- Low power density.

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#### Helicon Plasma Thruster (HPT)



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#### Introduction

### Framework



#### Gaseous Plasma Antennas (GPAs)

Devices relying on an ionized gas to radiate EM waves.

#### Feautures:

- Electrically reconfigurable;
- Low RCS, and thermal noise;
- Minimize co-site interference and signal degradation;
- Virtually *transparent* above the plasma frequency and *invisible* once turned off.

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Motivation, and Objectives

# Although different in shape, fields of applications, and working conditions, GPAs and HPTs share:



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## Motivation, and Objectives

#### Objectives

- Physical investigation into plasma generation, charged particle transport in a magnetized plasma, and wave-plasma coupling mechanism
- Clarify the role of the antenna in the source of HPTs, and the behavior of GPAs taking into account realistic excitation circuit and plasma transport
- Coupling of the EM solution with the plasma transport
- Design, and development of innovative plasma sources to be exploited as a GPA.

### Global Model

Plasma transport within a plasma source modeled by a 0-D fluid model.

#### Input

- Source Geometry: R, L;
- Neutral pressure p<sub>n</sub>;
- Deposited power P;

### Output

- Average plasma density  $n_e = n_i$ ;
- Average electron temperature  $T_e$ .

#### From input to output

System at equilibrium:

- Particles produced chemically = Particles lost in walls
- EM deposited power = Chemical losses + wall losses

### ADAMANT

Wave-plasma coupling modeled by an EM solver.

- Full-wave approach
- Coupled surface and volume integral equations
- Arbitrarily-shaped circuit
- Inhomogeneous and anisotropic plasma

### Plasma Model

- cold, and collisional
- multispecies
- non-uniform
- if magnetized,  $B_0 \parallel z$  axis

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 $\overline{\varepsilon}_{rk} = \begin{bmatrix} S_k & jD_k & 0\\ -jD_k & S_k & 0\\ 0 & 0 & P_k \end{bmatrix}$ 

## ADAMANT

#### Input

- Plasma mesh;
- PEC mesh;
- Source type, number of feeding points, *f*;
- Gas Type;
- n<sub>e</sub>, n<sub>i</sub>, T<sub>e</sub>, T<sub>i</sub>, B<sub>0</sub>, p<sub>n</sub>.

### Output

- Current distributions;
- Z-matrix, S-parameters;
- Scattered fields;
- Input, absorbed, and radiated power.

#### From input to output

- Surface Integral Equation
- Volume Integral Equation
- Excitation on the feeding port (voltage-gap approximation)

### Global Model and ADAMANT coupling



### Antenna Input Impedance, and Current Distribution



Cylindrical argon plasma column with:  $n_0 = 1 \cdot 10^{19} \text{ m}^{-3}$ ,  $T_e = 3 \text{ eV}$ ,  $p_n = 0.02 \text{ mbar}$ , L = 75 mm, and  $\Phi = 2.5 \text{ mm}$ 

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### The GPA Radiation Pattern







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### Numerical, and Experimental Approach



### Numerical Analysis - Plasma Source

Matlab genetic algorithm + Global Model

Decision-space variable	Optimization parameters
Plasma Radius Plasma Length	Plasma density of $10^{19} \text{ m}^{-3}$
Neutral Pressure Input Power	Minimize input power

Plasma Radius [mm]	5 - 10
Plasma Length [mm]	50 - 75
Neutral Pressure [mbar]	0.06 - 0.5
Input Power [W]	20 - 100

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### Numerical Analysis - Plasma Source

#### Matlab genetic algorithm + Global Model

	Case 1	Case 2	Case 3	Case 4
Plasma Radius [mm]	8.7	6.6	7.5	6.42
Plasma Length [mm]	54.6	56.2	78.9	72.5
Neutral Pressure [ mbar]	0.75	0.39	0.09	0.4
Input Power [W]	84.1	23.8	94.2	28.9

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### Numerical Analysis - Antenna Performances

Coupler Metal-coupler length Metal-coupler Φ	Sleeve, Half-Nagoya 30 – 42 mm 14 – 30 mm
Antenna Configurations Plasma $\Phi$	Monopolar, Bipolar 3 – 10 mm
Column length Column distance	50 — 130 mm 0 — 12 mm
Neutral gas	Ar, He, Ne
Neutral pressure	0.5 - 10  mbar
Working frequency	0.8 – 1.8 GHz
Voltage	1 V

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### Numerical Analysis - Antenna Performances



### Generation Method

#### Plasma generation: 2 techniques

#### RF - External Electrodes



#### HF - Internal Electrodes



 $p_n = 1 - 2mbar$   $\phi = 5 - 6mm$ Higher, and more uniform  $n_e$ Dirty atmosphere





#### Pyrex vessel with ad hoc interface





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#### Pyrex vessel with ad hoc interface



#### Sealing process

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#### Pyrex vessel with ad hoc interface



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#### Vessel Preparation

Commercial electrodes sealed with a tube of the desired dimensions



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#### Vessel Preparation

Commercial electrodes sealed with a tube of the desired dimensions



### Aging process





### Diagnostic



## Source Characterization

**RF** Discharges



Argon, 
$$p_n = 1$$
 mbar,  
 $\Phi = 10$  mm,  $L = 130$  mm.

Argon,  $p_n = 10$  mbar,  $\Phi = 3$  mm, L = 130 mm.

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### Source Characterization RF Discharges





Argon,  $p_n = 1$  mbar,  $\Phi = 10$  mm, L = 130 mm.

Argon,  $p_n = 10$  mbar,  $\Phi = 3$  mm, L = 130 mm.

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## Source Characterization

**HF** Discharges

We explored different gas pressures, and mixture

Gas	p <sub>n</sub> [mbar]	n <sub>0</sub> [m <sup>-3</sup> ]
Ar	1	$3.70\cdot 10^{18}\pm 1.84\cdot 10^{17}$
Ar - Ne	2	$3.84\cdot 10^{18}\pm 8.57\cdot 10^{16}$
Ar	2	$4.40\cdot 10^{18}\pm 5.09\cdot 10^{17}$
Ar - Hg	2	$3.18\cdot 10^{18}\pm 5.88\cdot 10^{16}$
Ar - Hg	1	$2.59\cdot 10^{18}\pm 1.50\cdot 10^{17}$

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## Source Characterization

#### **HF** Discharges



### Antenna Characterization - Reflection Coefficient

Argon,  $p_n = 10$  mbar,  $\Phi = 3$  mm, L = 130 mm.



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### Antenna Characterization - Reflection Coefficient

Argon,  $p_n = 10$  mbar,  $\Phi = 3$  mm, L = 130 mm.



### Antenna Characterization - Gain Pattern

Antenna testing with a well-known Log-Hallo Antenna as transmitter.



Friis Transmission Equation:

$$G_r = P_r - P_t - G_t - 10 \log_{10} \left(\frac{\lambda}{4\pi R}\right)^2$$



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#### Source Testing

## Antenna Characterization

 $P_r$  on the E-plane

 $G_{max}$  on the E-plane

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### Conclusions

- Development of a tool that couples the EM solution with the plasma transport, useful to study both GPAs, and Plasma Thrusters.
- Physical assessment on wave propagation in a plasma column.
- Physical assessment on the radiation properties of a plasma dipole.
- Design of 2 plasma sources to be exploited in a GPA.
- Preliminar assessment on the antenna performance of a GPA.

### Publications

- Trezzolani, F., Magarotto, M., Manente, M., Moretto, D., Bosi, F.J., Gallina, G., De Carlo, P., Melazzi, D., Pavarin, D., Pessana, M., Development of a counterbalanced pendulum thrust stand for electric propulsion, (2017) 4th IEEE International Workshop on Metrology for AeroSpace, MetroAeroSpace 2017 - Proceedings, art. no. 7999554, pp. 152-157.
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- Melazzi, D., De Carlo, P., Manente, M., Lancellotti, V., Pavarin, D., Numerical results on the performance of gaseous plasma antennas, (2015) Proceedings of the 2015 International Conference on Electromagnetics in Advanced Applications, ICEAA 2015, art. no. 7297180, pp. 569-572.
- Trezzolani, F., Bosi, F., Melazzi, D., De Carlo, P., Selmo, A., Manente, M., Ferraris, S., Pessana, M., Pavarin, D., Development of a kW-level plasma thruster in project SAPERE-STRONG, (2015) Proceedings of the International Astronautical Congress, IAC, 10, pp. 8106-8113.

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