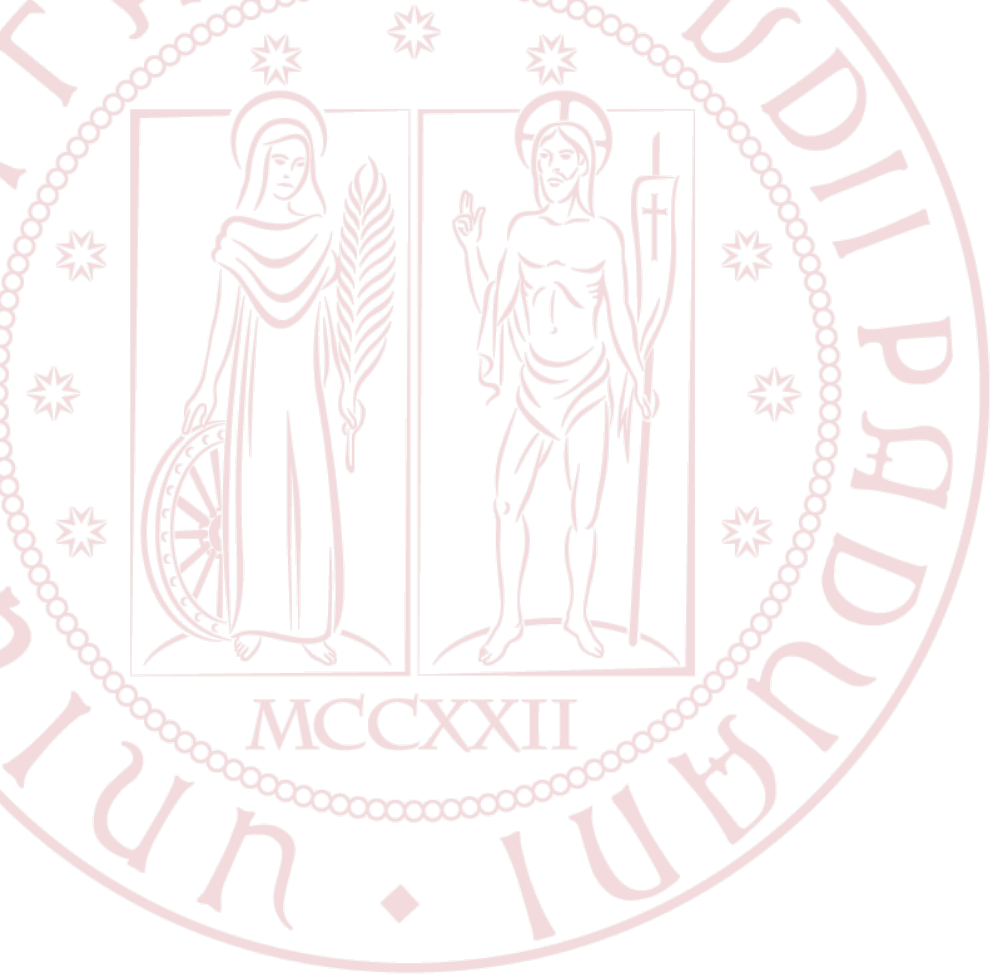


# DESIGN OF A ROBOTIC ARM FOR LABORATORY SIMULATIONS OF SPACECRAFT PROXIMITY NAVIGATION AND DOCKING

DOTTORANDO: **Andrea Antonello**

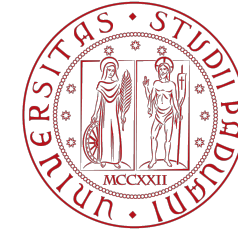
SUPERVISORE: **Alessandro Francesconi**





# OUTLINE OF THE PRESENTATION

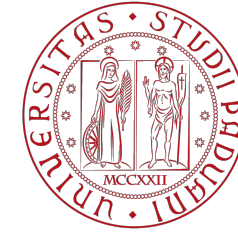
- MOTIVATION AND OBJECTIVES
- STATE OF THE ART
- WHAT'S NEW
- APPLICATIONS
- WBS
- CURRENT PROGRESS



# INTRODUCTION

- ◆ THE INCREASING NUMBER OF ORBITING OBJECTS HAS LAID THE NEED FOR SERVICING AND MAINTENANCE OPERATIONS (OSS)
- ◆ THE GOAL OF THIS RESEARCH IS THE DEVELOPMENT OF A MANIPULATOR ARM FOR GROUND TESTING OF OOS OPERATIONS
- ◆ OOS OPERATIONS CONSIST MAINLY OF:
  - ◆ RENDEZ-VOUS
  - ◆ DOCKING
  - ◆ BERTHING



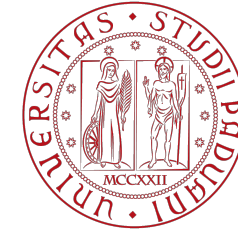


## STATE OF THE ART

- ♦ DLR's EPOS EXPERIMENT IS ONE OF THE LEADING EDGE OOS SIMULATION FACILITY
- ♦ TWO 6DOF INDUSTRIAL ROBOTS AND A 25 M LONG TESTING SITE
- ♦ THE HARDWARE-IN-THE-LOOP SYSTEM ALLOWS THE SIMULATION OF **DYNAMIC CONTACTS, GRAVITY AND EVEN SUNLIGHT ILLUMINATION**
- ♦ THE CONTROLLER IMPOSES A DYNAMICS THAT IS DIFFERENT FROM THE LABORATORY DYNAMICS, WHICH IS CONTINUOUSLY CORRECTED

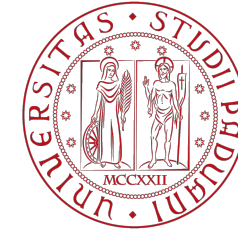




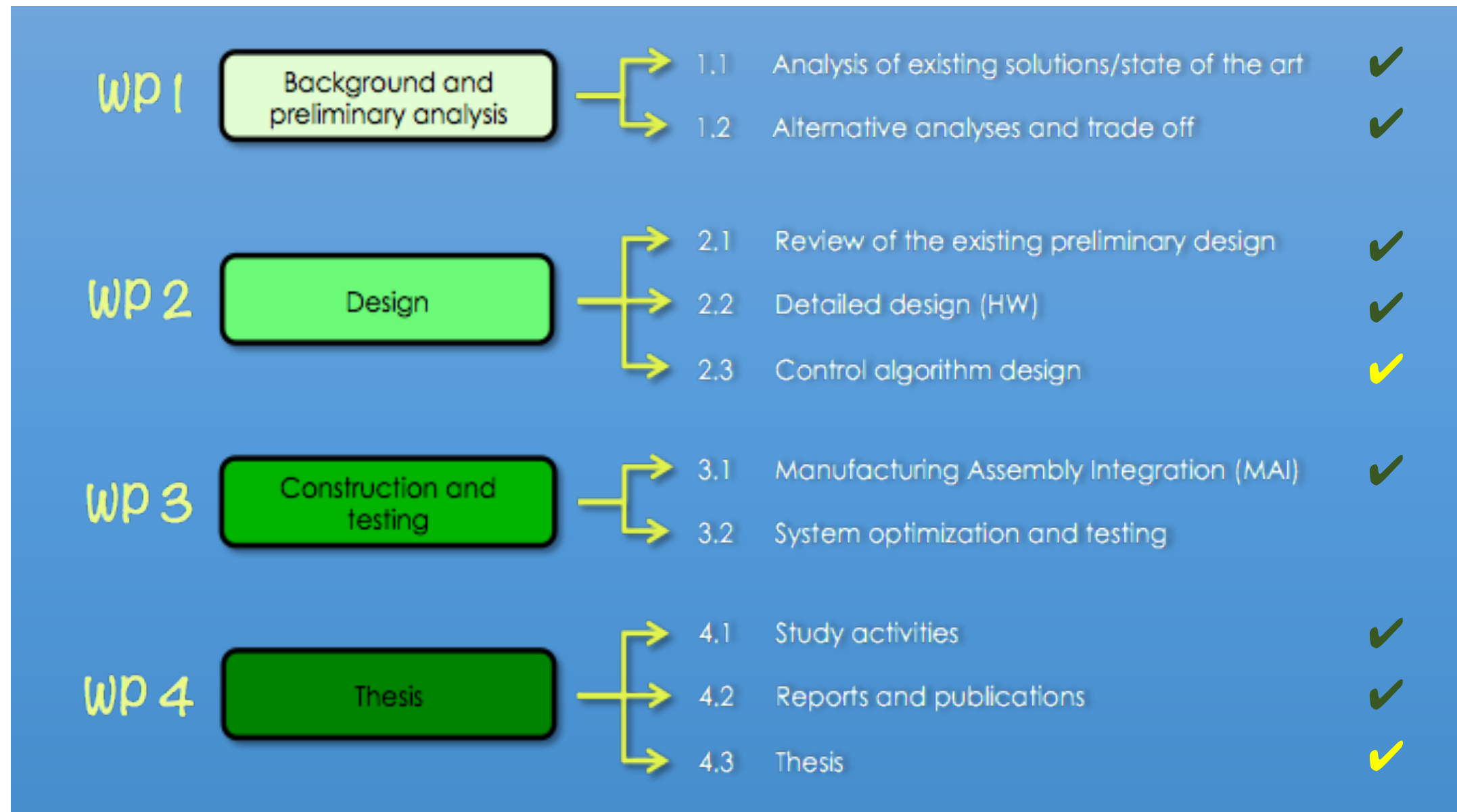


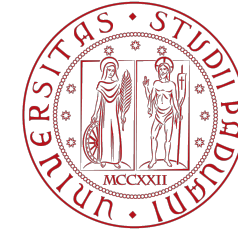
## WHAT'S NEW

- ◆ THIS PROJECT PROPOSES AN ALTERNATIVE TO THESE HUGE AND COSTLY FACILITIES
- ◆ THE SYSTEM WILL BE CONSTITUTED BY A SINGLE ROBOT WITH THE FOLLOWING PECULIARITIES:
  - ◆ THE CONTROLLER COMPUTES AND IMPOSES THE REAL-TIME ORBITAL TRAJECTORY, AND IMPOSES A DYNAMICS THAT IS DIFFERENT FROM THE LABORATORY DYNAMICS
  - ◆ ACTIVE CONTROL OF THE TRAJECTORY WITH FORCE TRANSDUCERS
  - ◆ POSSIBILITY TO SIMULATE MUCH BIGGER SCENARIOS THAN THE MERE WORKSPACE WITH THE AID OF DYNAMIC SCALING LAWS
  - ◆ MUCH SMALLER DIMENSIONS AND WEIGHT



# WORK BREAKDOWN STRUCTURE





## WHAT'S NEW

- ♦ THE WORK DONE UP TO NOW WILL BE PRESENTED AS BROKEN DOWN IN THE FOLLOWING SECTIONS:

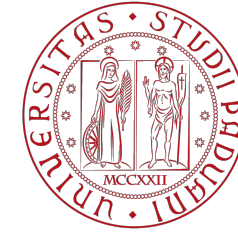
A) MECHANICAL DESIGN (PD)

B) SIMULATION SCENARIO DESIGN

C) SENSOR SUITE DESIGN

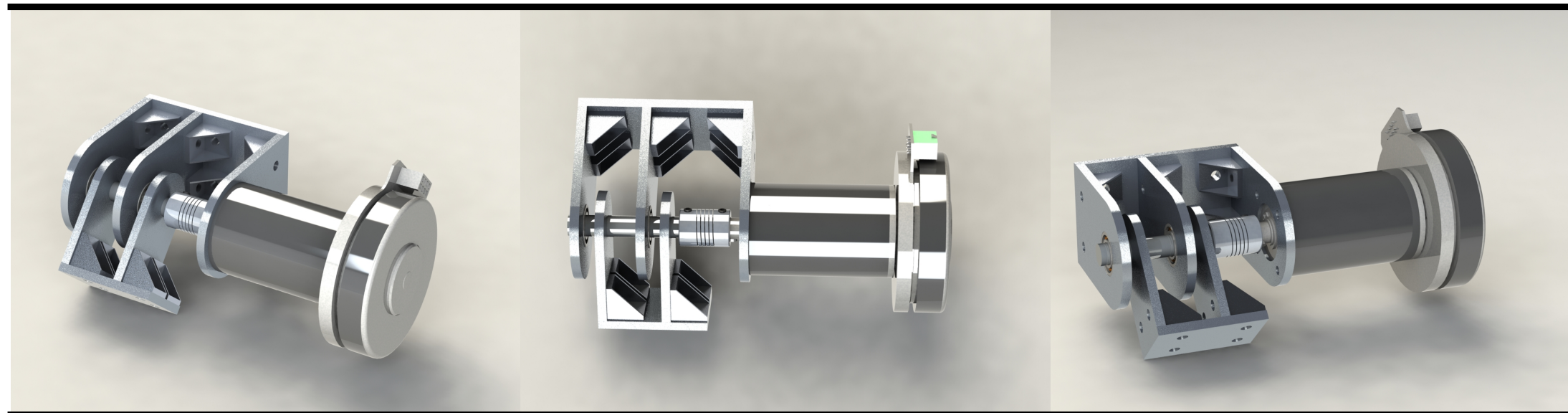
### INTERNATIONAL COLLABORATIONS

- BOSTON UNIVERSITY
- MIT
- GEORGIA INST. OF TECHNOLOGY

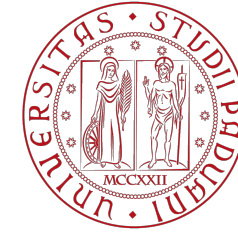


## MECHANICAL DESIGN

- ♦ FOLLOWING THE PRELIMINARY DESIGN AND REQUIREMENTS VERIFICATION (COMPLETED IN THE FIRST YEAR), STARTING FROM JANUARY 2015, I SUPERVISED A MASTER STUDENT FOR THE FINAL SELECTION OF THE MOTORS AND GEARINGS, WHICH LED IN SEPTEMBER 2015 TO THE WRITING OF A MASTER THESIS THAT CONTAINED THE COMMERCIAL PARTS TO BE PURCHASED FOR THE FACILITY.

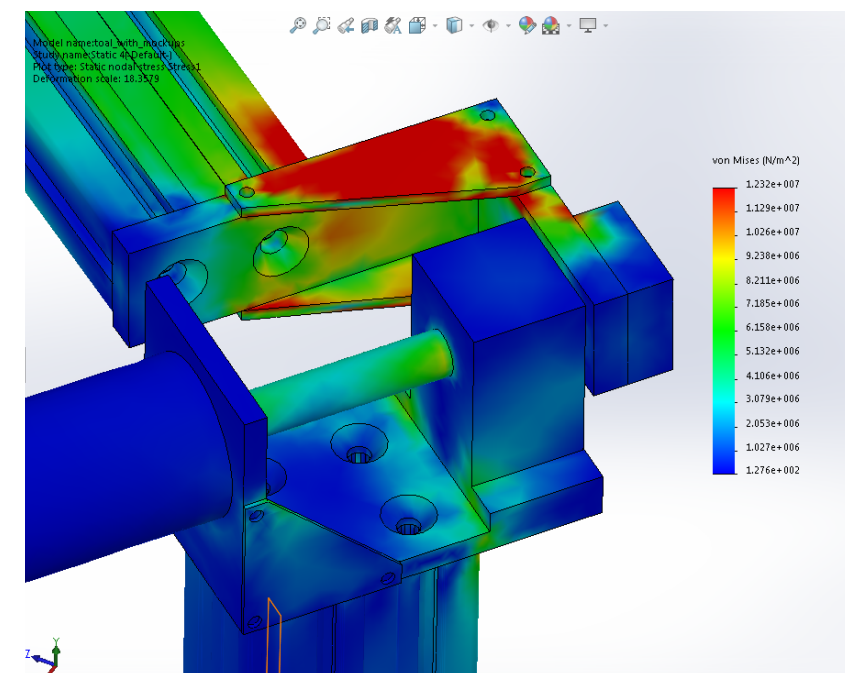
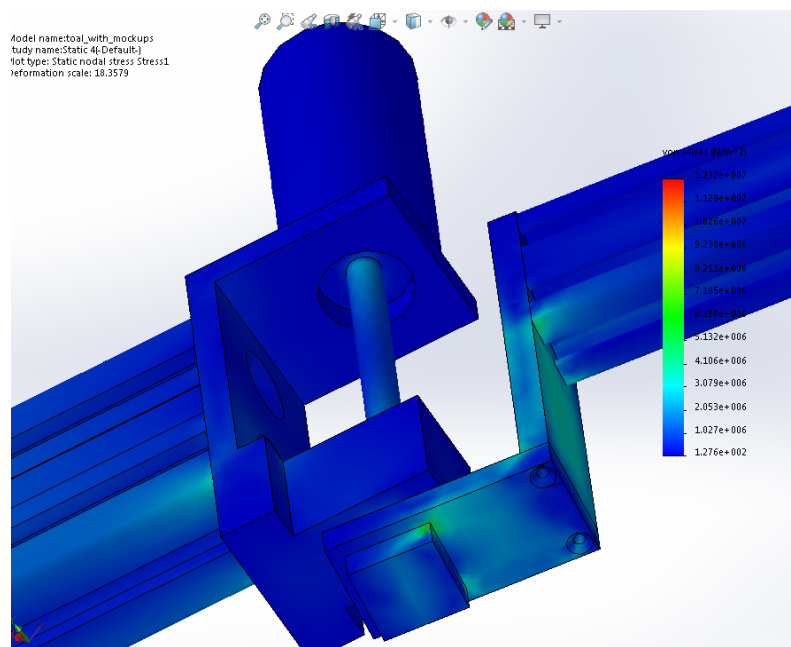


preliminary design

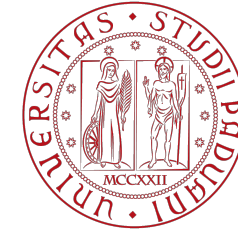


# MECHANICAL DESIGN

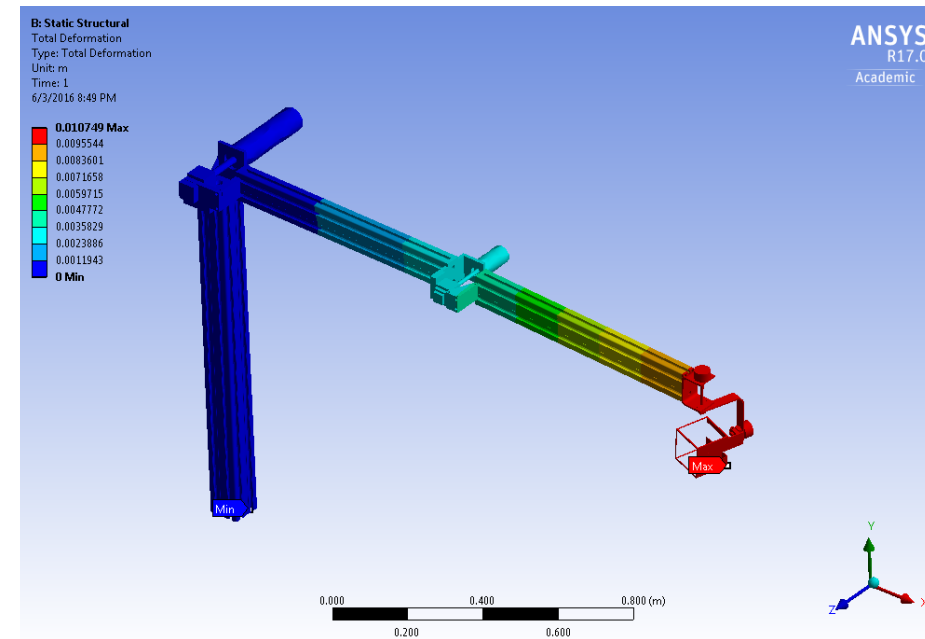
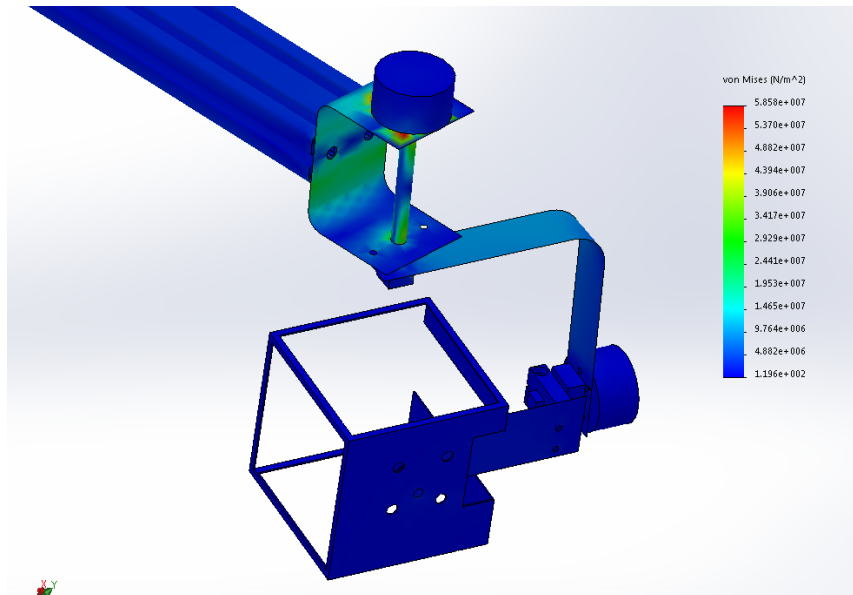
- ◆ ONCE A FIRST DRAFT OF THE JOINT BLOCKS WAS COMPLETED, I PROCEEDED WITH THE FEM SIMULATIONS TO VERIFY THE CHARACTERISTICS OF THE SYSTEM.
- ◆ I STARTED FROM A STATIC SIMULATION CAMPAIGN WHICH LED TO THE CORRECT SIZING OF THE COMPONENTS IN ORDER TO AVOID OVER-ENGINEERING.



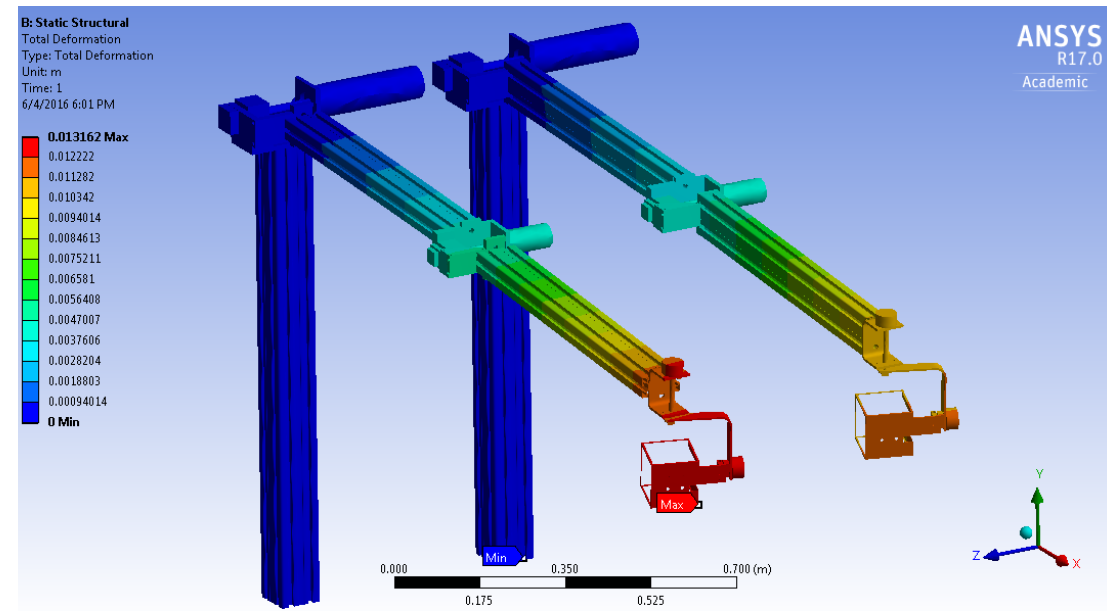
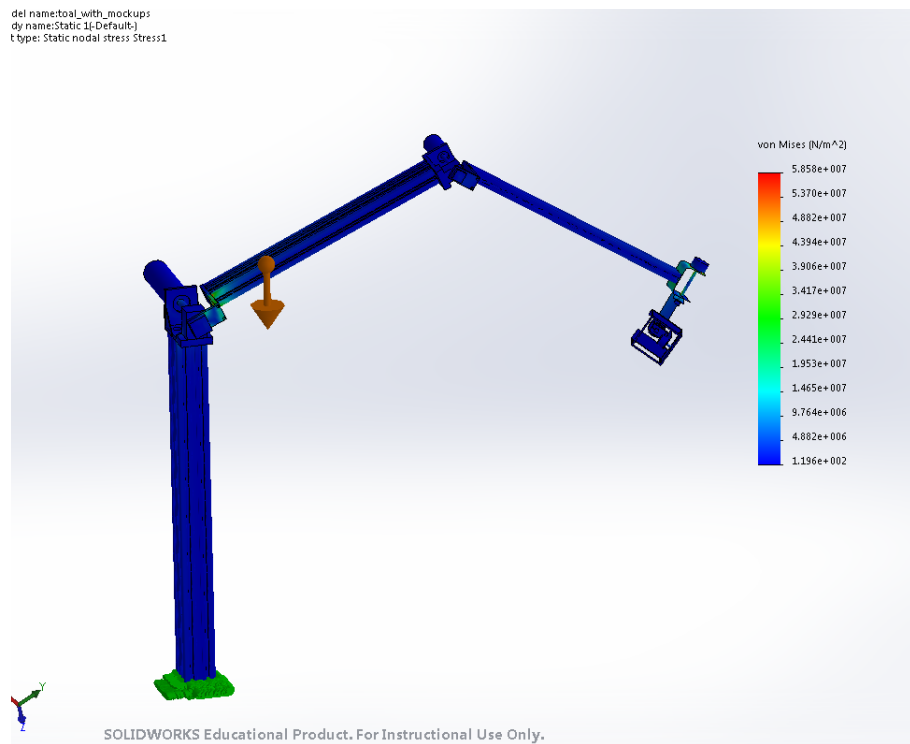




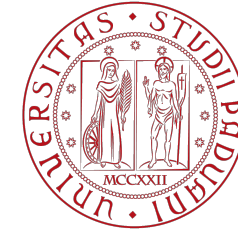
# MECHANICAL DESIGN



del name:toal\_with\_mockups  
dy name:Static 1(-Default-)  
t type: Static nodal stress Stress1

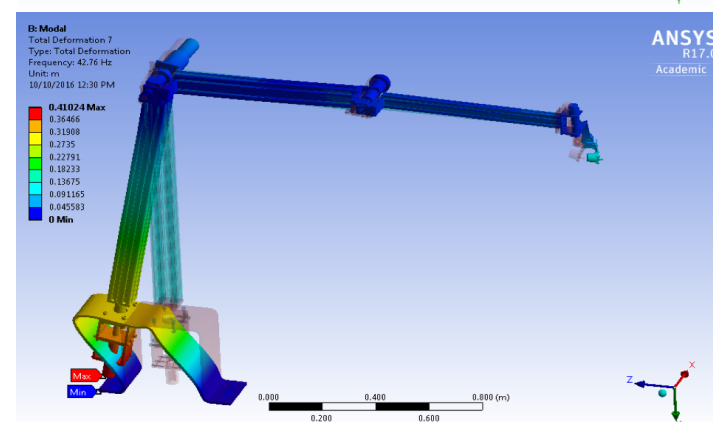
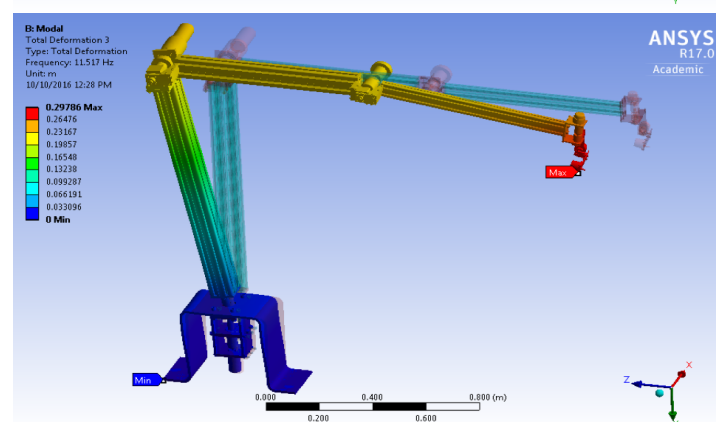
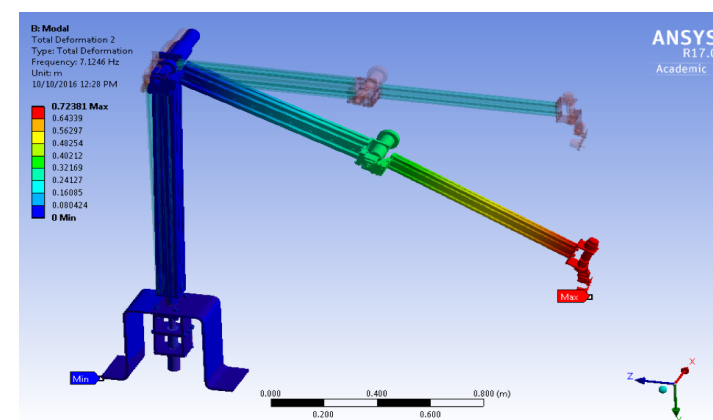
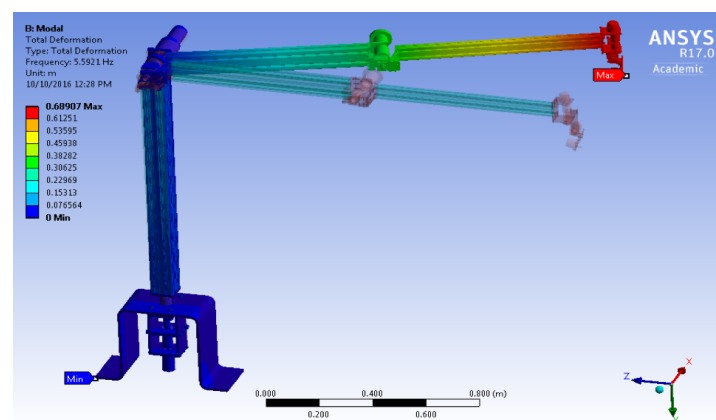


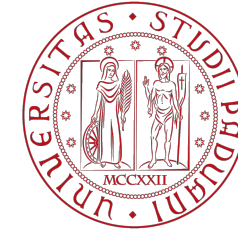




# MECHANICAL DESIGN

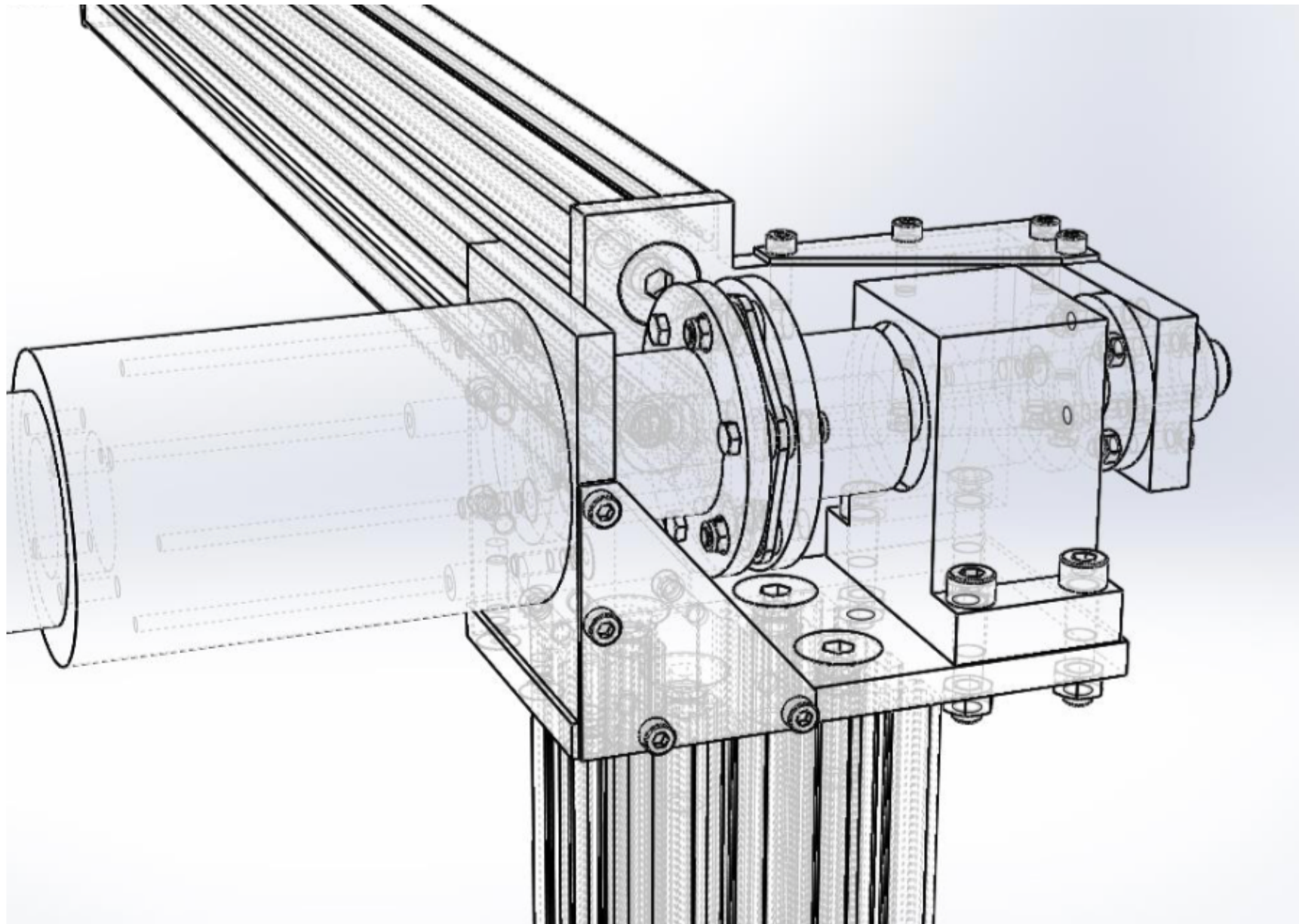
- ◆ FINALLY, A MODAL ANALYSIS WAS PERFORMED AND LED TO THE CALCULATION OF THE VIBRATIONAL FREQUENCIES OF THE SYSTEM.
- ◆ THIS LED TO A FURTHER REFINEMENT OF THE DESIGN, BY CHANGING THE THICKNESS OF SOME OF THE SUPPORTING PLATES AND BY ADDING STRENGTHENING TRIANGLES AT JOINT 2 AND 3.

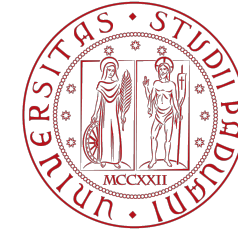




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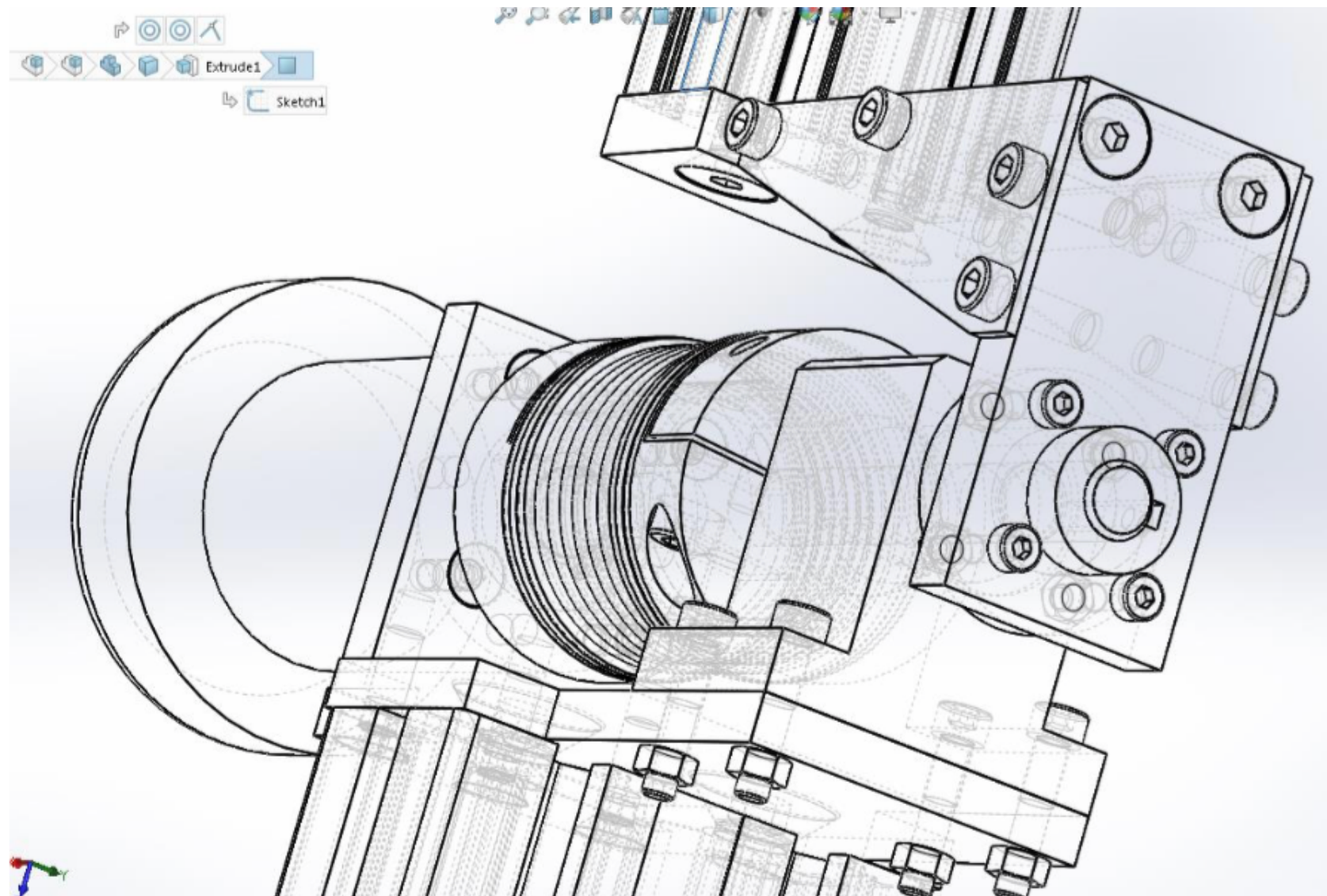
- ◆ ONCE THE SIMULATIONS PROVIDED SATISFACTORY RESULTS, THE ARM DESIGN WAS FINALIZED.



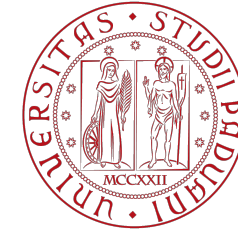


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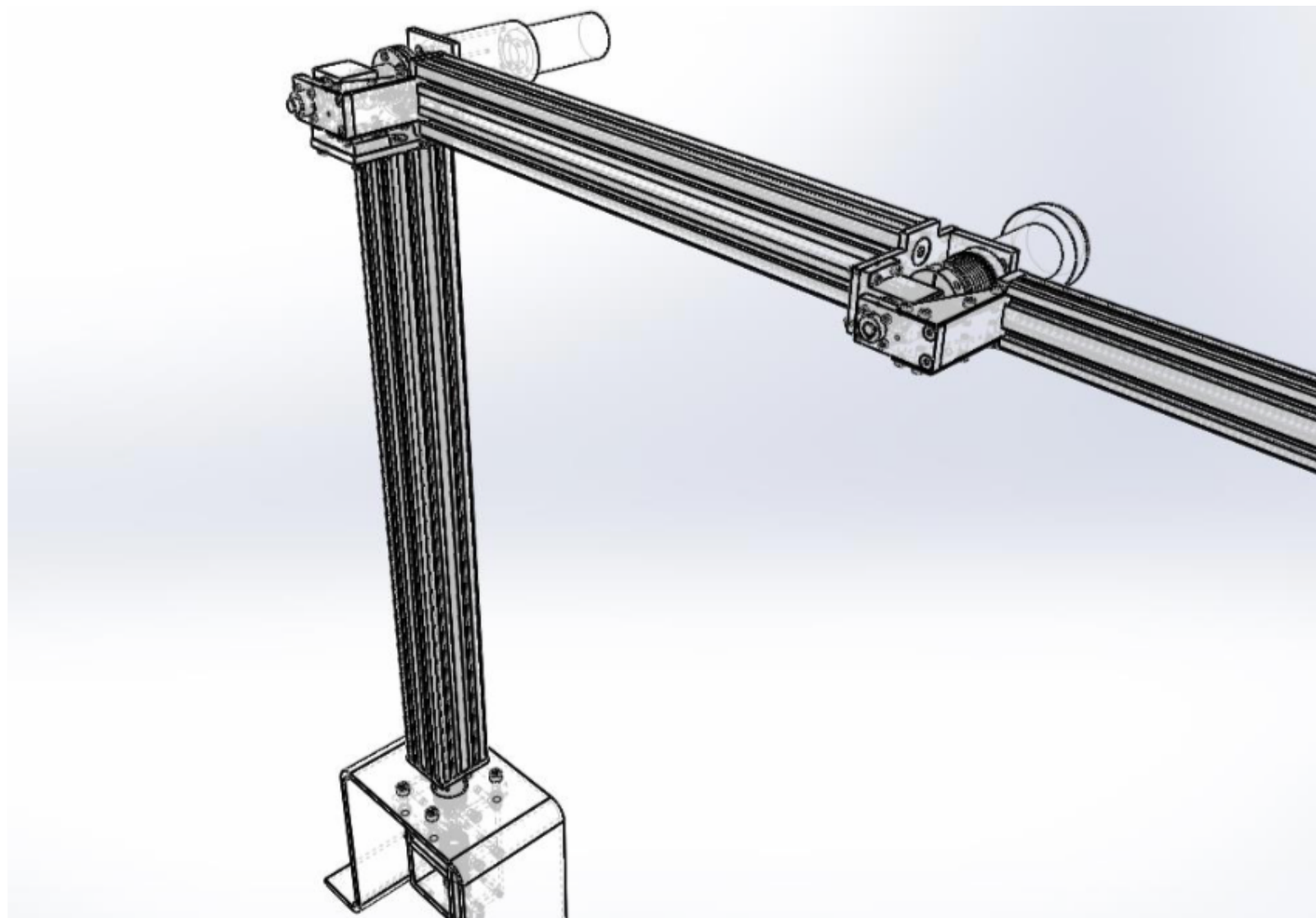


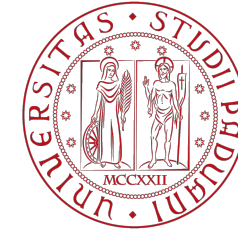




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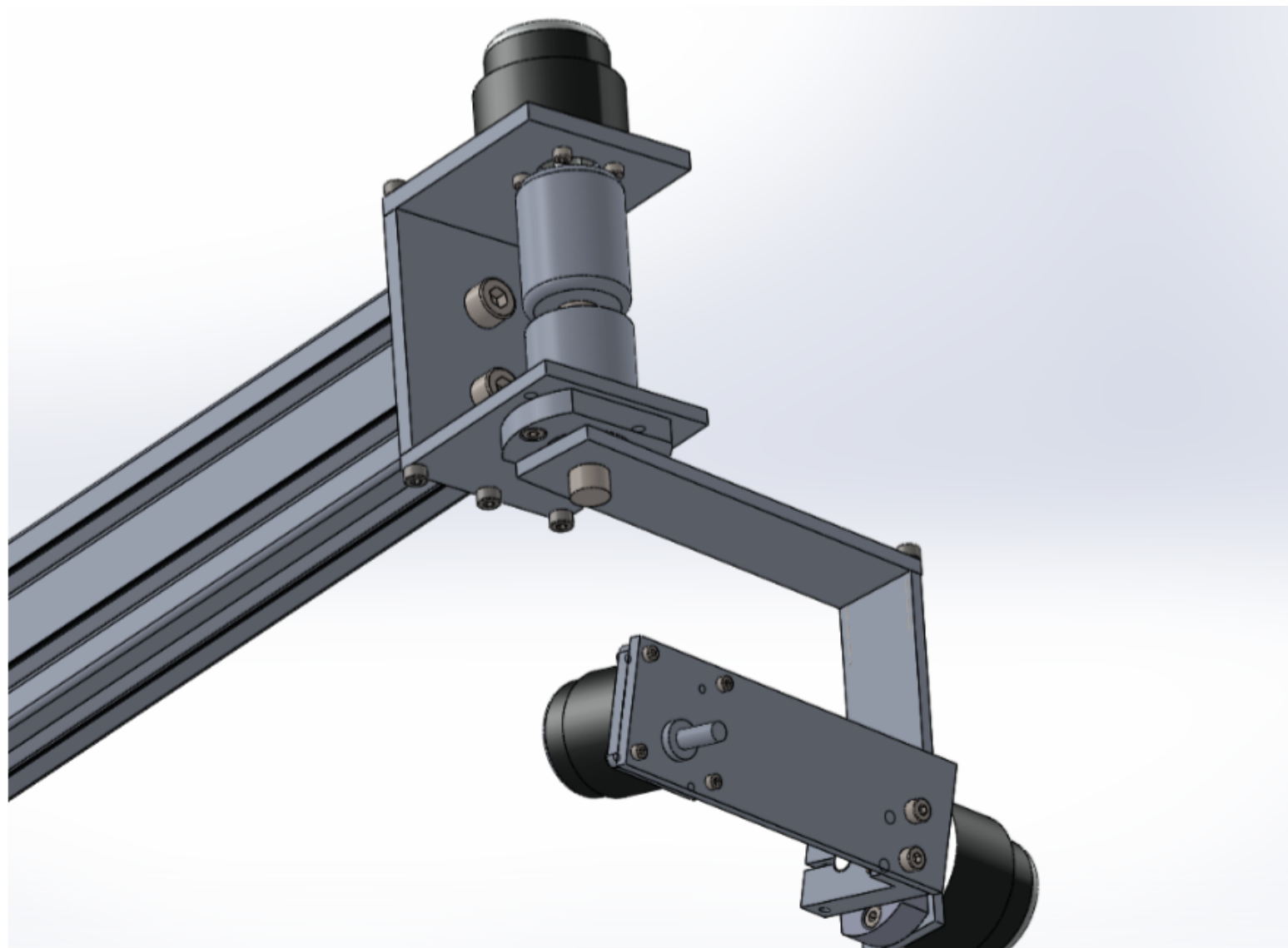
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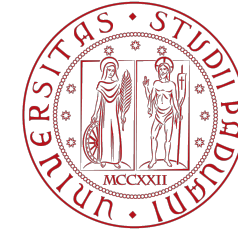




## MECHANICAL DESIGN

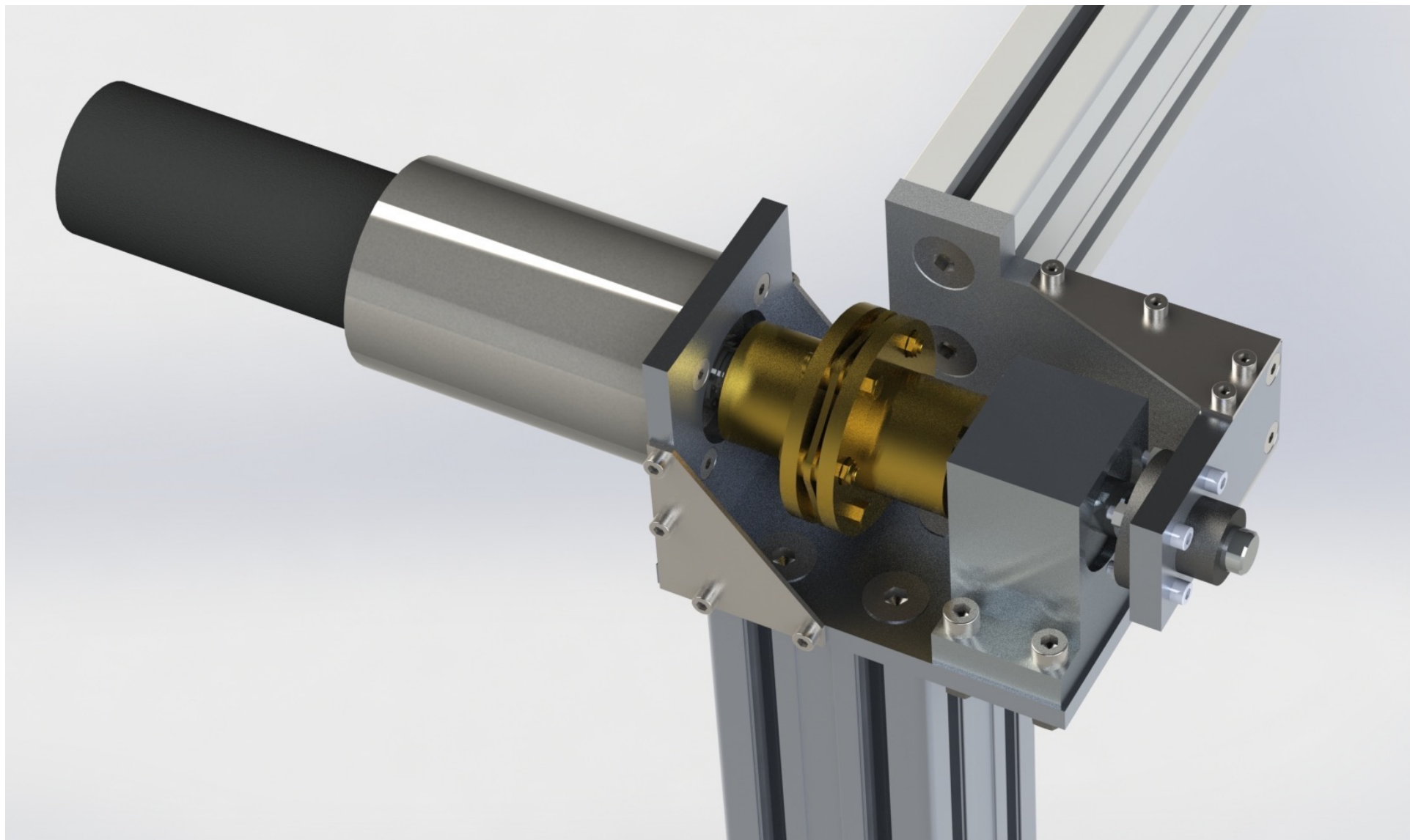
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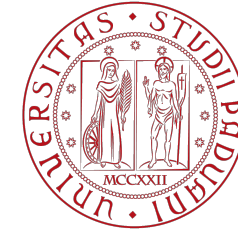


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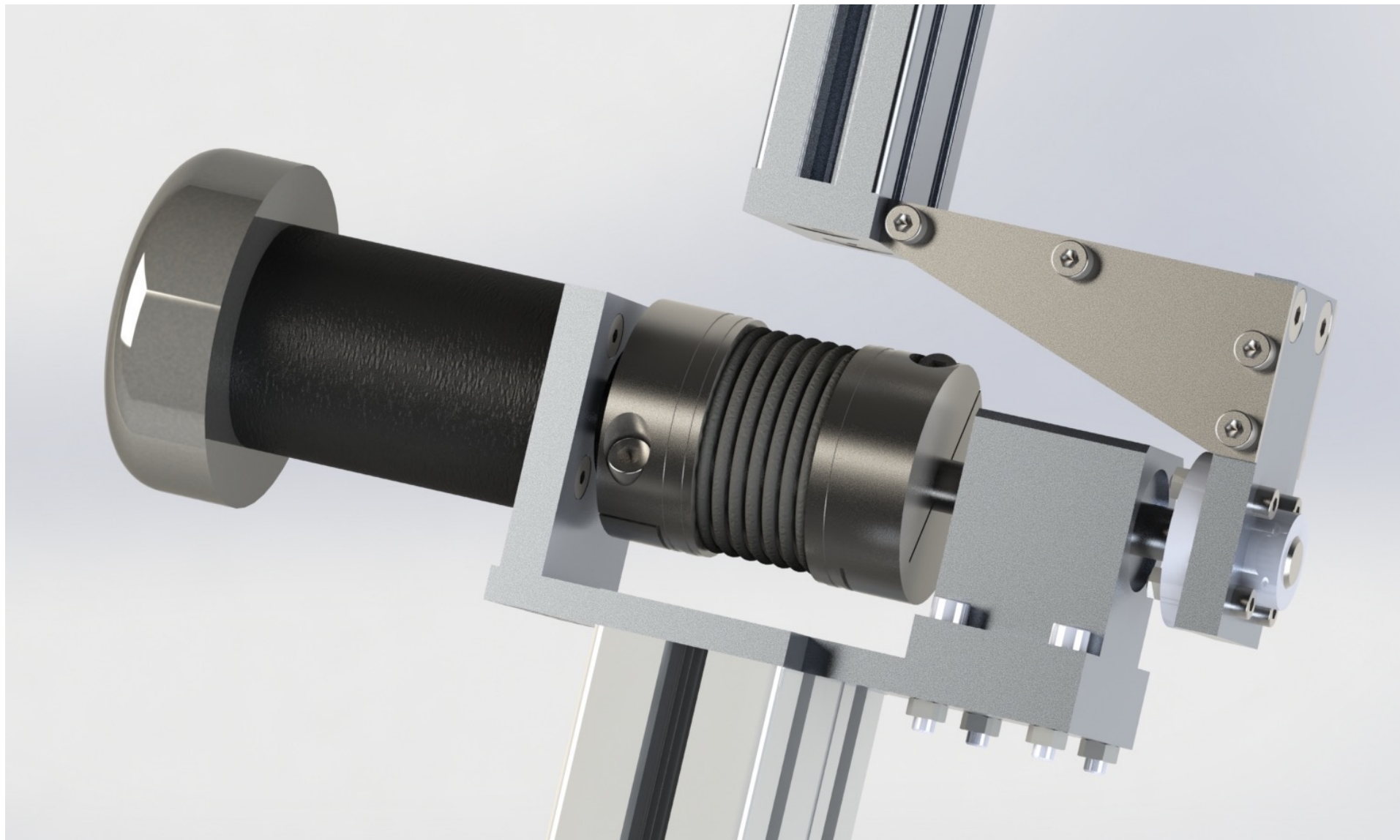


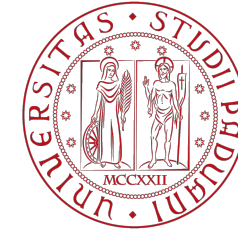




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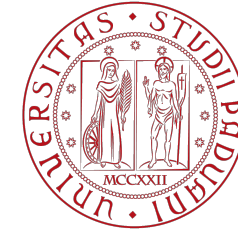


# MAIT - MANUFACTURING & ASSEMBLY

- ♦ AS OF OCTOBER 2016, MOST OF THE PARTS TO BE ASSEMBLED HAVE BEEN PROCURED. ASSEMBLY HAS STARTED AND IS EXPECTED TO BE FINISHED IN THE FIRST WEEKS OF NOVEMBER

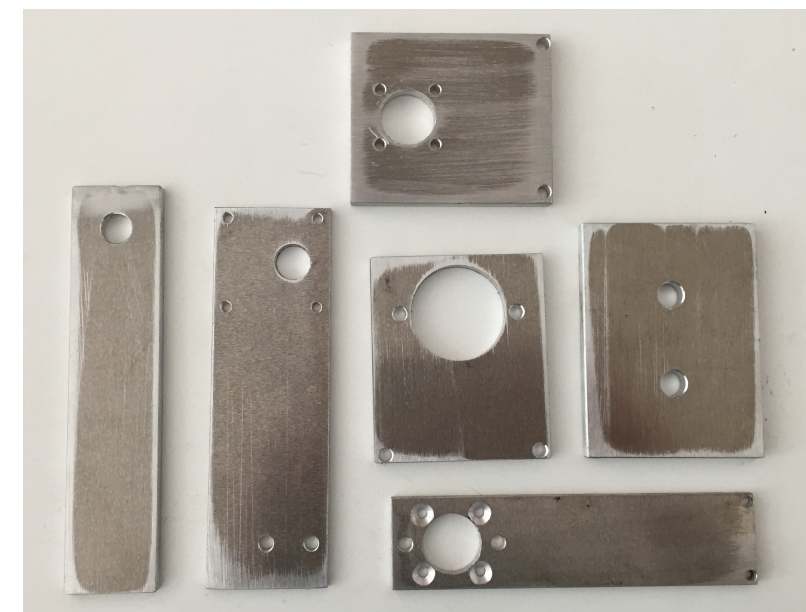
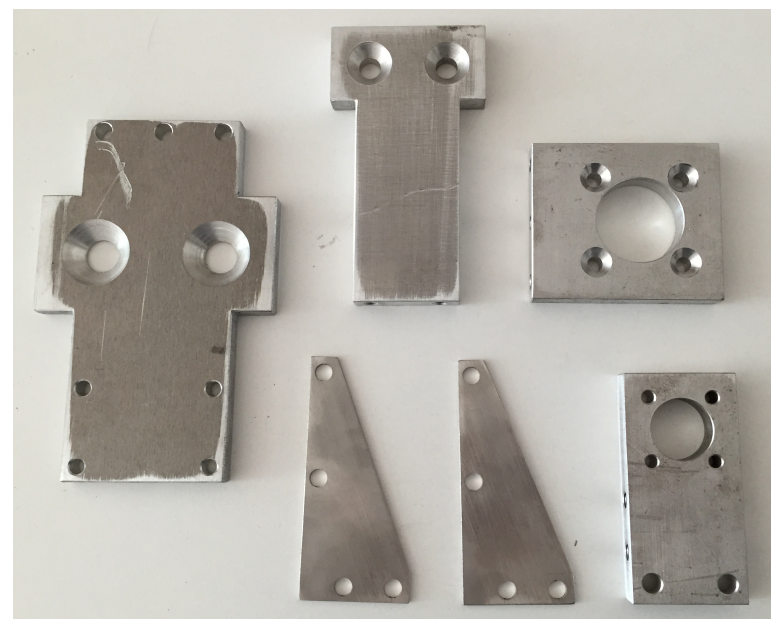
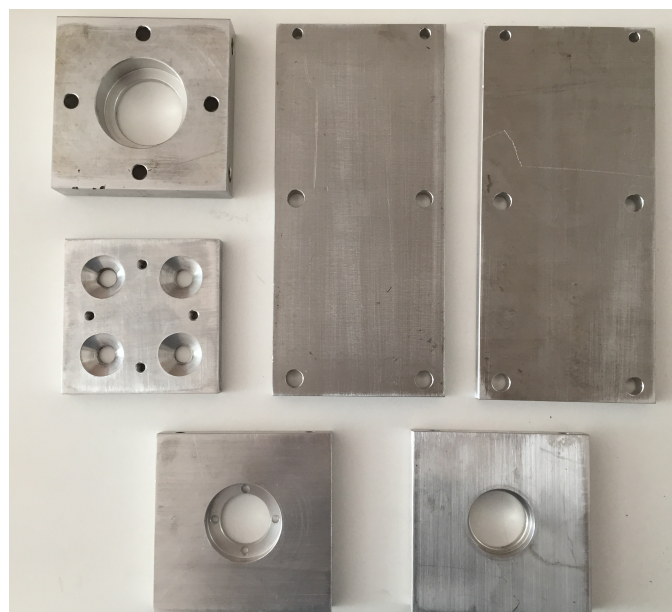


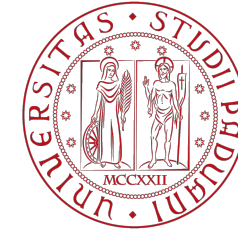




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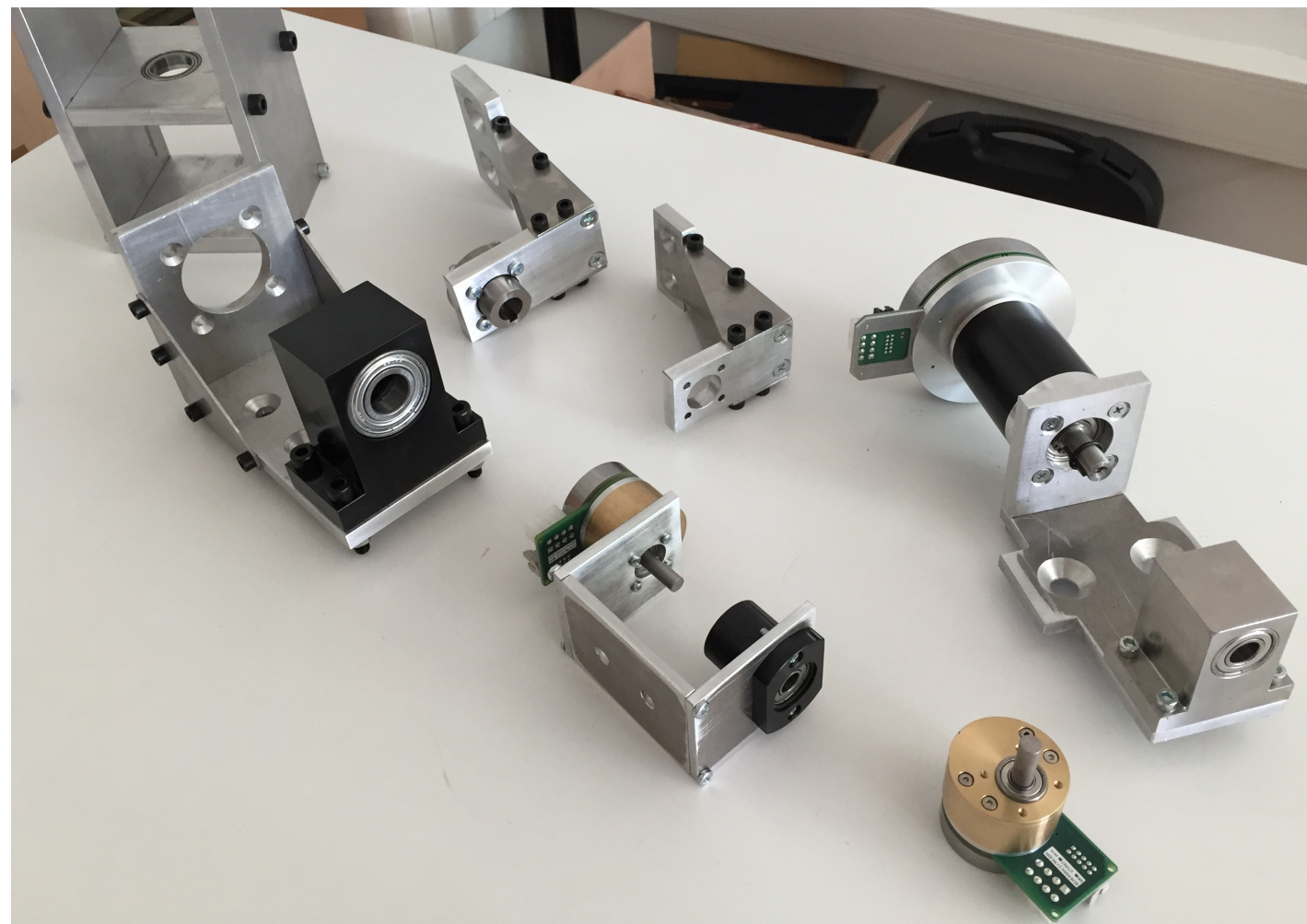
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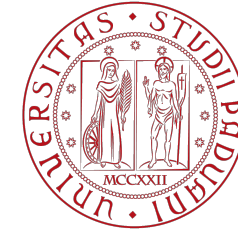


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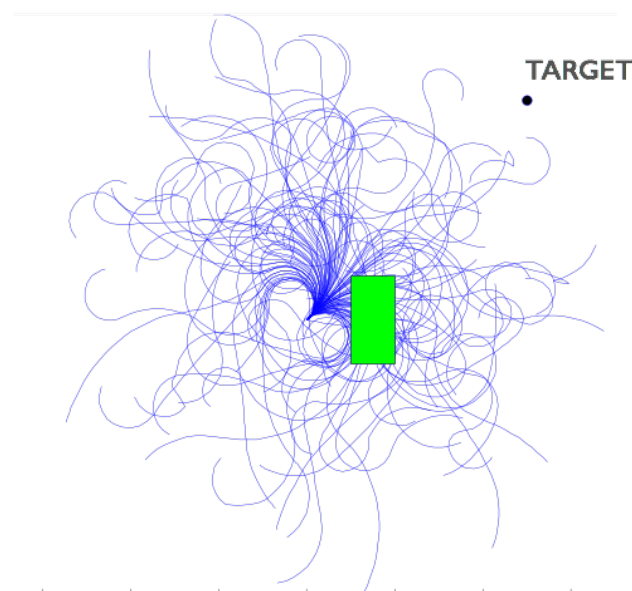




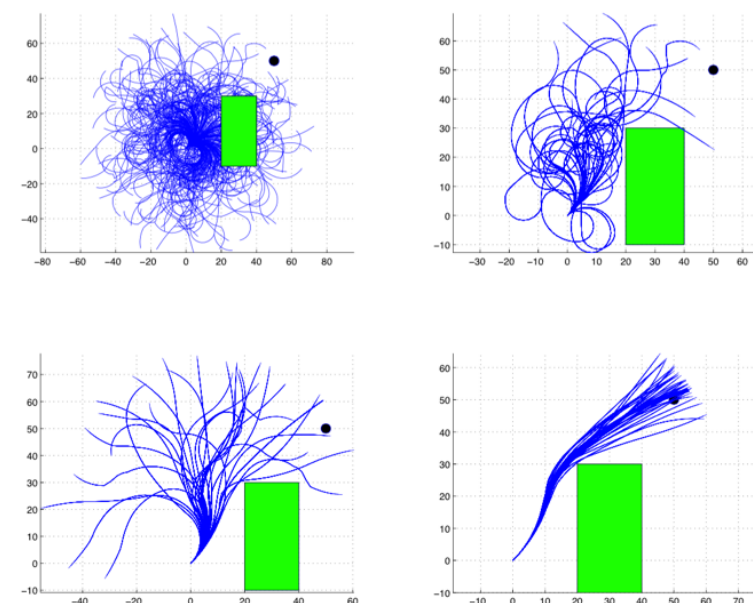
# CROSS ENTROPY TRAJECTORY PLANNING

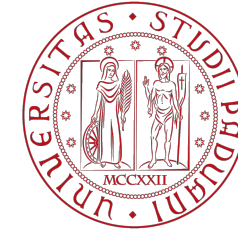
- ◆ LOOKING AT SIMULATION SCENARIOS FOR THE ROBOTIC FACILITY, I WORKED ON CROSS-ENTROPY BASE SLAM TECHNIQUES UNDER THE SUPERVISION OF PROF. P. TSIOTRAS FROM GEORGIA TECH UNIVERSITY (ATLANTA, USA)
- ◆ CROSS ENTROPY STRATEGIES ARE A STOCHASTIC WAY TO PERFORM SLAM MINIMIZING, THROUGH STATISTICAL TOOLS, THE COMPUTATION AND ACTUATION COSTS.

MONTECARLO



CROSS ENTROPY (4 iterations)





# CROSS ENTROPY TRAJECTORY PLANNING

- ♦ NOVELTY: EXISTING WORK IN PROXIMITY OPERATIONS SOLVE THE PROBLEM OF CONTROL AND ESTIMATION **INDEPENDENTLY**
- ♦ WE INTEGRATE STOCHASTIC OPTIMIZATION WITH AGENT LOCALIZATION TO THE PROBLEM OF A CHASER CIRCUMNAVIGATING A TARGET. THE GOAL IS TO LOCALIZE SOME LANDMARKS LOCATED ON THE TARGET SATELLITE.
- ♦ GOAL IS TO MINIMIZE A COST FUNCTION THAT ENCLOSSES BOTH THE FINAL UNCERTAINTY OF THE ESTIMATE AND THE ACTUATION COST:

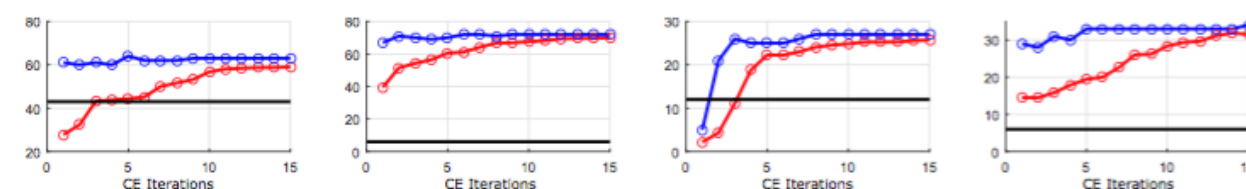
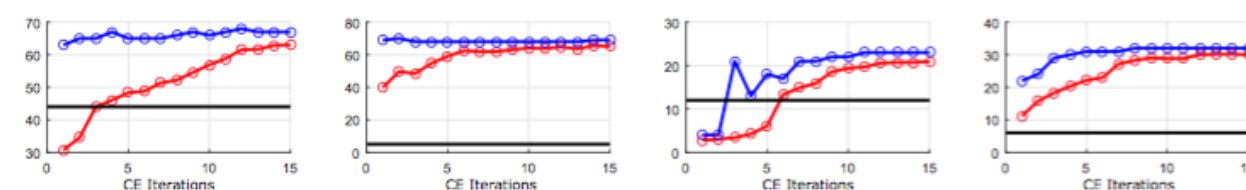
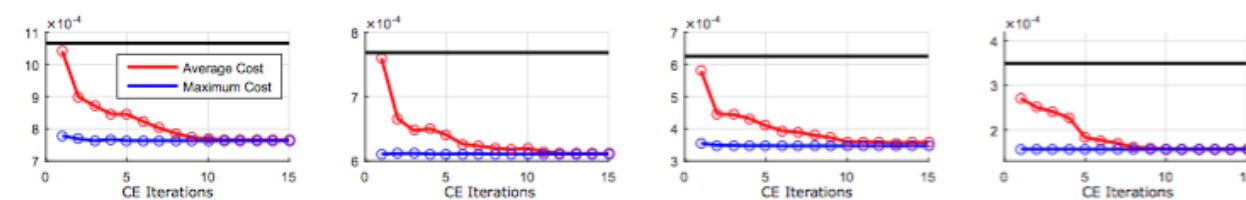
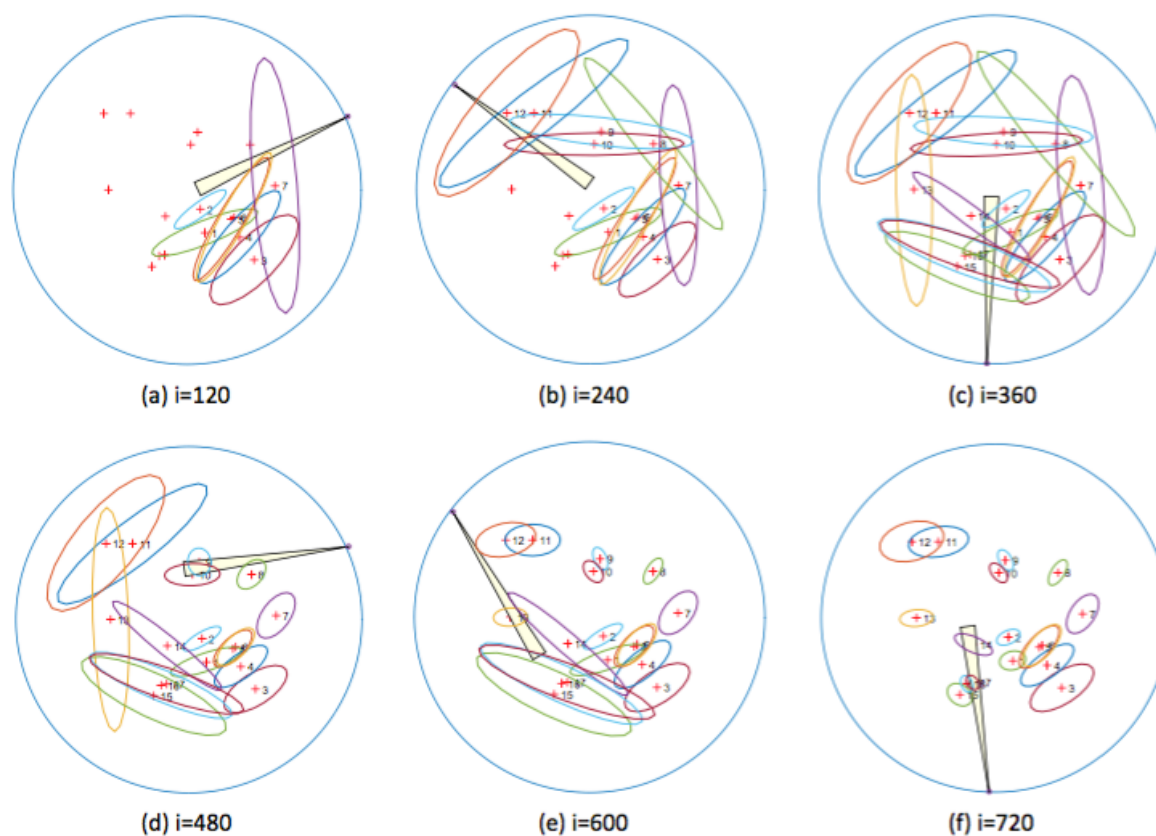
$$\mathcal{L}(\mathbf{x}, \mathbf{u}) = \|\mathbf{e}^2(t_N)\| + \int_0^{t_N} \left( Q(\mathbf{x}) + \frac{1}{2} \mathbf{u}^T R \mathbf{u} \right) dt$$

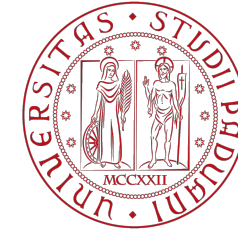




# CROSS ENTROPY TRAJECTORY PLANNING

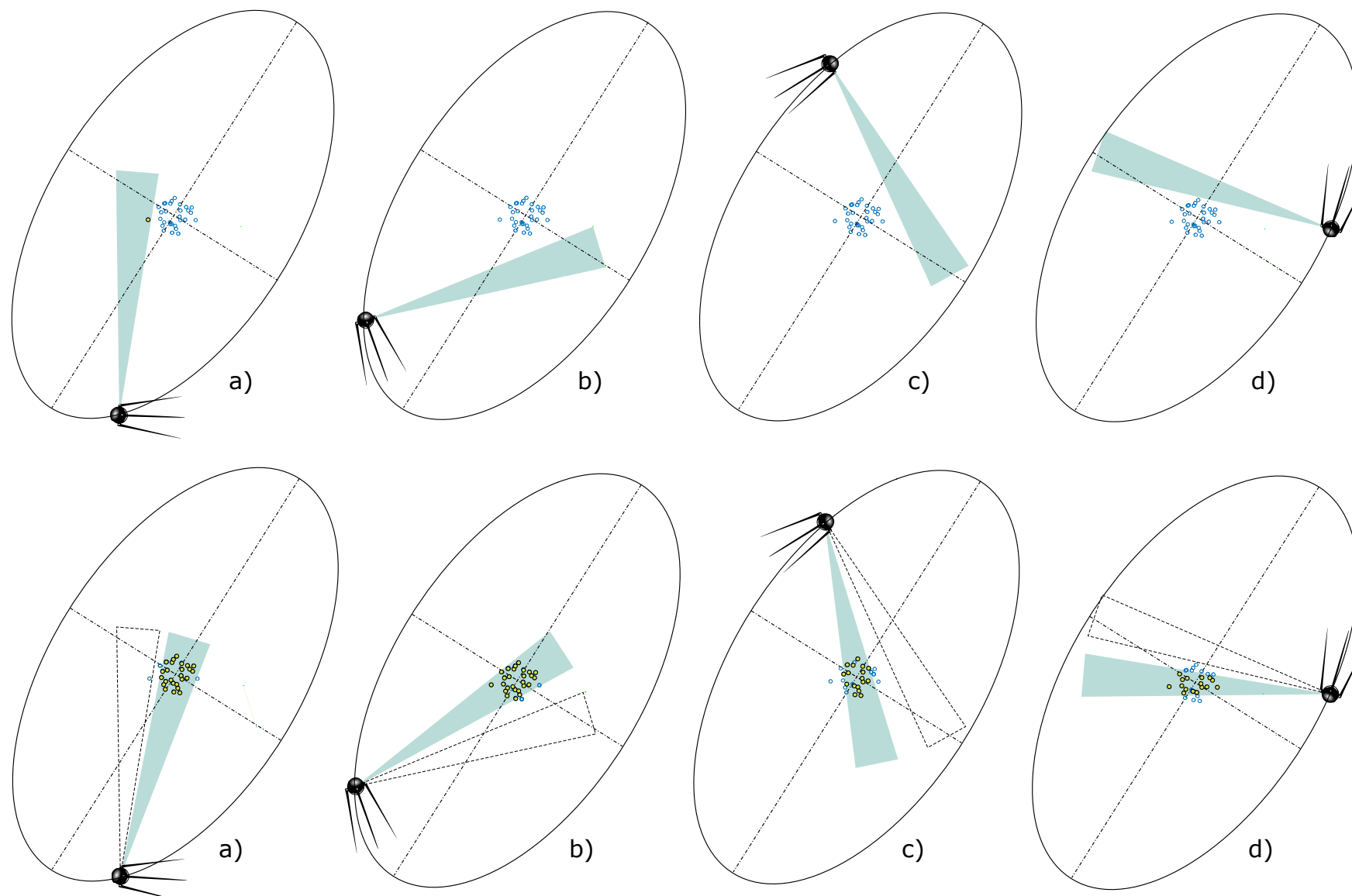
- ◆ TWO PUBLICATIONS WERE PRODUCED BY APPLYING THIS STRATEGY TO THE SCENARIO OF CONTROLLING A CAMERA MOUNTED ON A SATELLITE CIRCUMNAVIGATING A TARGET, PERFORMING SLAM
- ◆ SEVERAL CONTROL POLICIES WERE PERFORMED AND A FULLY 3D SIMULATION WITH A CLOHESSY-WILTSHIRE FRAME WAS USED





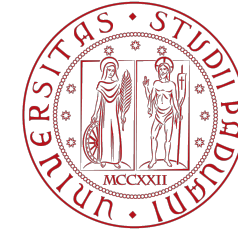
# CROSS ENTROPY TRAJECTORY PLANNING

- ◆ THIS NEW TECHNIQUE OUTPERFORMED BOTH THE NON CONTROLLED CASE AND A PD CONTROLLER



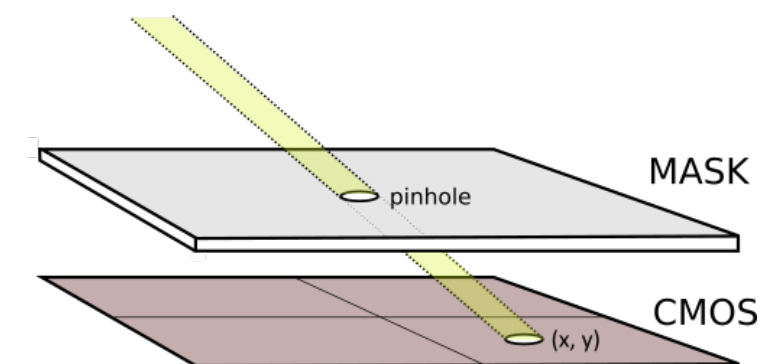
NON CONTROLLED

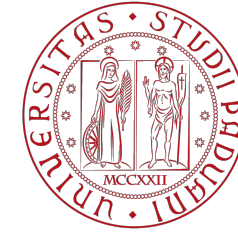
CONTROLLED



# SUN SENSOR

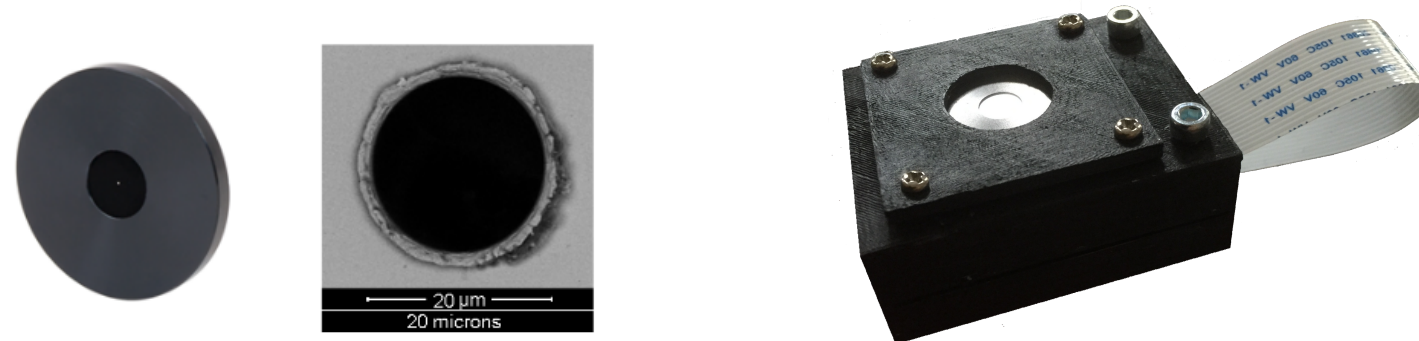
- ♦ AS A SPINOFF FROM THE STUDENT PROJECT ONORBIT, I DEVELOPED A SUN SENSOR THAT WILL SERVE AS AN ATTITUDE SENSOR MOUNTED IN THE SATELLITE MOCKUP AT THE TIP OF THE ROBOT END EFFECTOR.
- ♦ AS OF TODAY, THERE EXIST SEVERAL PRIVATE COMPANIES THAT OFFER SENSORS WHICH HAVE A MAXIMUM ACCURACY OF  $0.3-0.5^\circ$  AND PRICES STARTING AT \$3000/UNIT
- ♦ WE USED OFF THE SHELF, ACTIVE PIXEL SENSOR (CMOS) COMPONENTS WITH HIGH RESOLUTION AND LIMITED FOOTPRINT.
- ♦ THE SIMPLIFIED WORKING PRINCIPLE IS BASED ON THE PINHOLE CAMERA





# SUN SENSOR

- ♦ A PROTOTYPE OF THE SENSOR HAS BEEN CREATED USING OFF THE SHELF COMPONENTS:
  - ★ A CMOS CAMERA, A 20UM PINHOLE MASK & A 3D PRINTED ENCLOSURE WITH DIMENSIONS 28X30X15 MM



- ♦ TOTAL MASS IS 15 GRAMS AND THE SIZE IS 28X30X15 MM.

FIELD OF VIEW

66.2 x 51.1°

ACCURACY

±0.02°

RESOLUTION

±0.023°

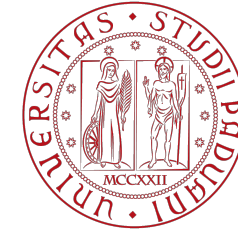
PRECISION

±0.024°

COST

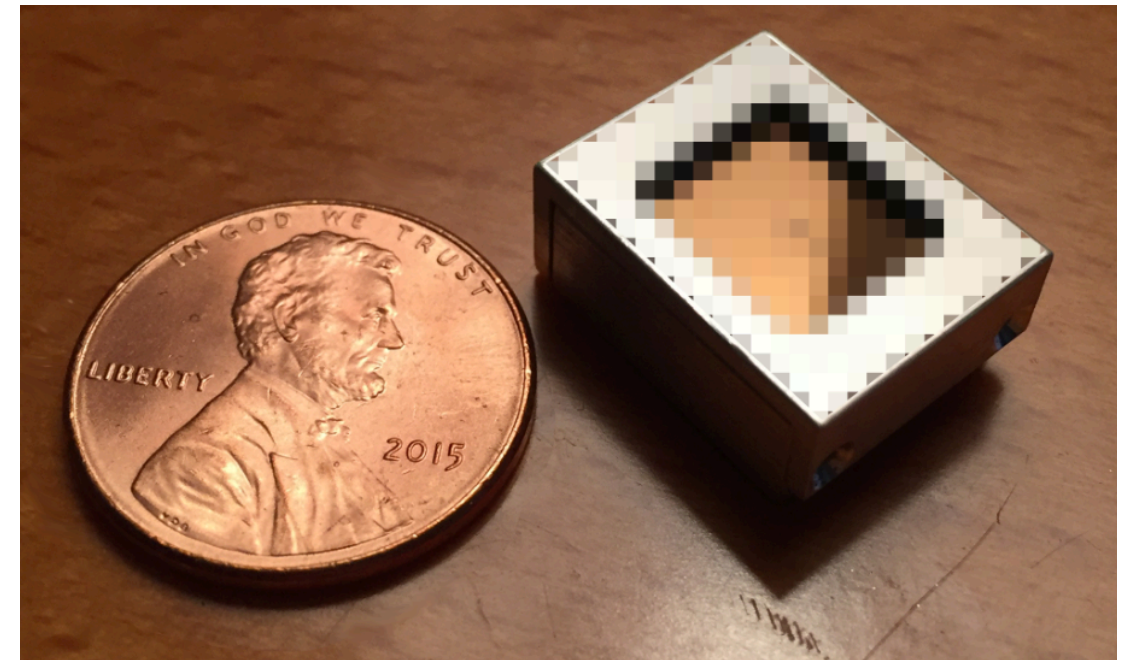
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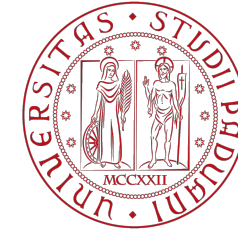




# SUN SENSOR

- ♦ STARTING FROM THE CURRENT PROTOTYPE, WE REFINED THE SENSOR ARCHITECTURE AND MINIATURIZED THE FINAL PRODUCT: THE NEW SENSOR WEIGHS ONLY 2.5 GRAMS AND HAS A DIMENSIONS SUM OF 35 MM, MAKING IT THE **SMALLEST** SUN SENSOR ON EARTH IN 2016 WITH THESE PERFORMANCES

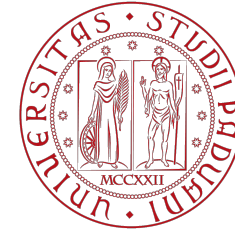




# CONCLUSIONS

- ♦ COMPLETED THE KINEMATIC AND DYNAMIC MODELING OF THE FACILITY, WITH CONTACT SIMULATION THROUGH THE VIRTUAL FORCE METHOD (IEEE BENEVENTO PAPER, 2014)
- ♦ COMPLETED THE MECHANICAL DETAILED DESIGN AND FEM ANALYSIS (STATIC, MODAL AND CONTACT) OF THE MANIPULATOR
- ♦ PROCUREMENT COMPLETED AND ASSEMBLY IN FINAL PHASES
- ♦ APPLICATION OF CROSS ENTROPY METHOD TO THE PROBLEM OF OSS IN RENDEZVOUS SCENARIOS, TO BE SIMULATED BY THE FACILITY
- ♦ DEVELOPED AND MANUFACTURED A MINIATURIZED SUN SENSOR TO BE USED AS A PART OF THE SENSOR SUITE AT THE TIP OF THE END EFFECTOR





QUESTIONS?